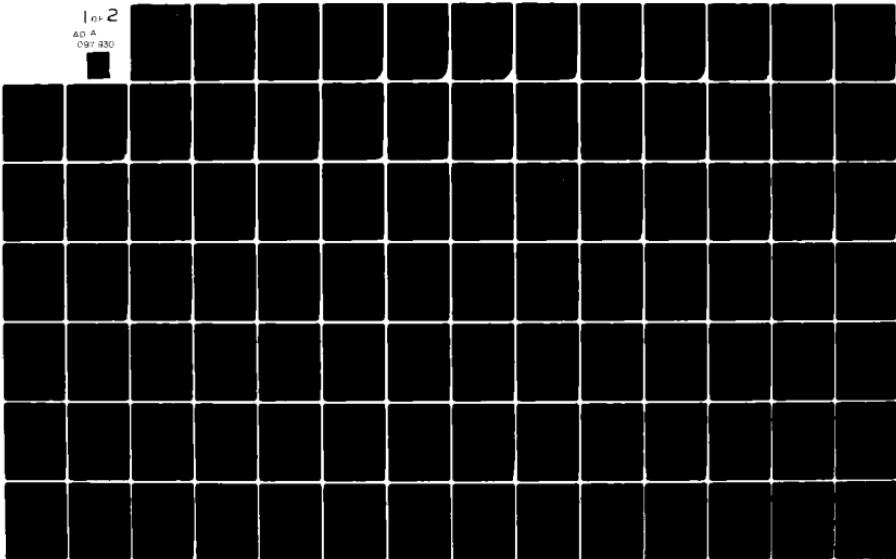


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DAVID W. TAYLOR NAVAL SHIP
RESEARCH AND DEVELOPMENT CENTER

Bethesda, Maryland 20084



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MEASUREMENTS OF THE EFFECT OF TRIM ON
THE NOMINAL WAKE OF THE NAVAL
AUXILIARY OILER AO-177

by

Michael B. Wilson

Gary A. Hampton

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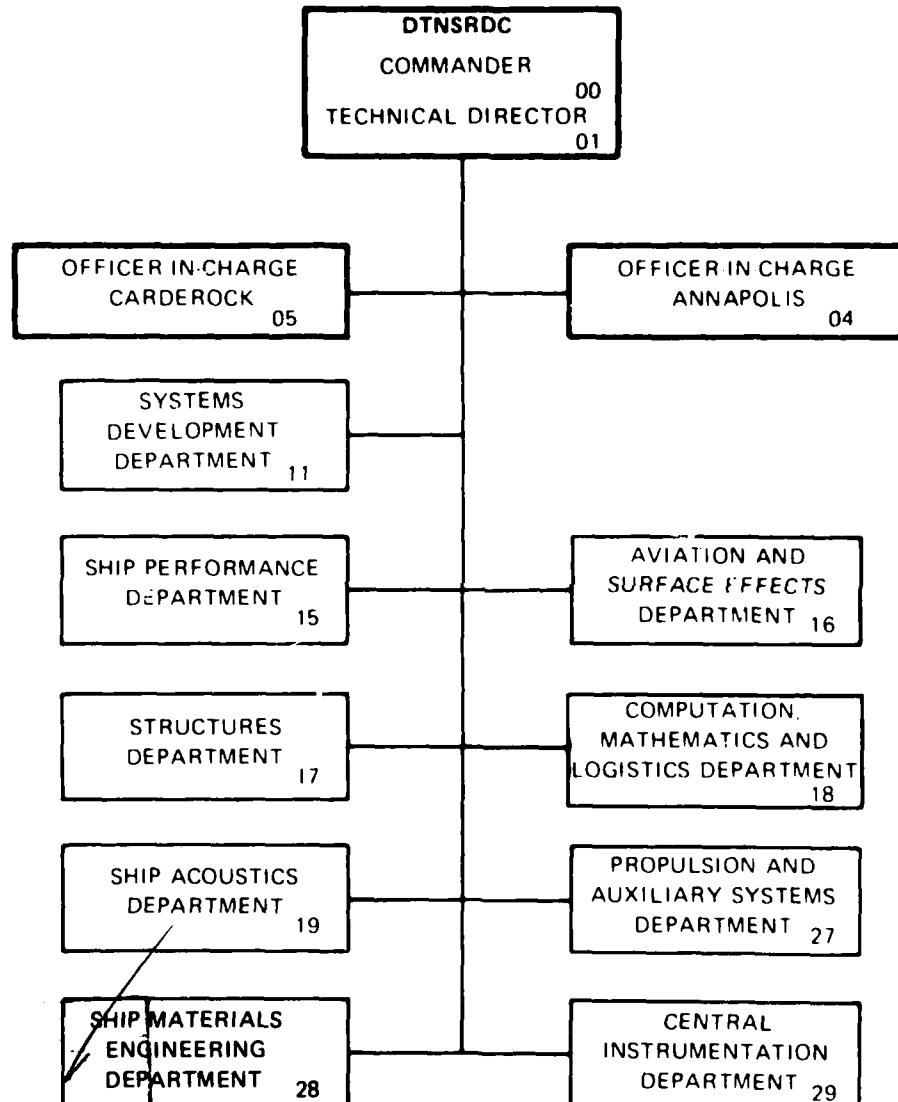
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NOTATION

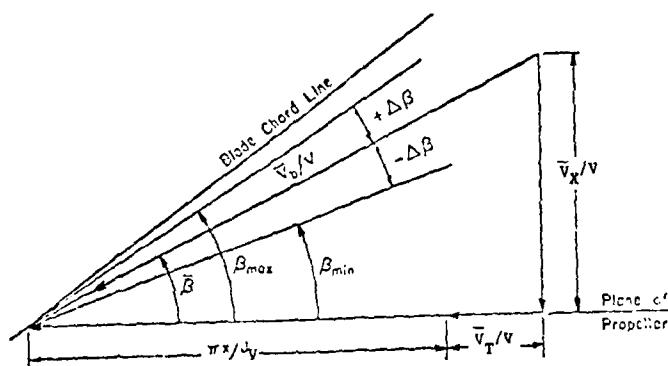
<u>Conventional Symbol</u>	<u>Symbol Used on Computer Plots</u>	<u>Definition</u>	<u>Units*</u>
A_N	COS COEFF	Cosine coefficient of the N^{th} harmonic	-
B_N	SIN COEFF	Sine coefficient of the N^{th} harmonic	-
C_B		Block coefficient	-
C_P		Prismatic coefficient	-
C_X		Maximum section area coefficient	-
D		Propeller diameter	L
G_W		Wake gradient parameter	-
J_V	JV	Apparent advance coefficient $J_V = V/nD$	-
N	N	Harmonic number	-
n		Propeller rate of revolution(rps) Rev/T	
r/R or x	RAD	Radial distance (r) from propeller axis expressed as ratio to propeller radius	-
r_{hub}		Hub radius	L
R		Propeller radius	L
Ψ		Ship displacement volume	L^3
V		Ship or model velocity (free stream)	L/T
$v_R(x, \theta)$	VR	Radial component of fluid velocity at a given point	L/T
$\bar{v}_R(x)$		Circumferential average radial velocity component	L/T

* M = mass; L = length; T = time.

<u>Conventional Symbol</u>	<u>Symbol Used on Computer Plots</u>	<u>Definition</u>	<u>Units</u>
$v_R(x, \theta)/V$	VR/V	Radial velocity ratio	-
$\bar{v}_R(x)/V$	VRBAR	Average radial velocity ratio	-
$v_T(x, \theta)$	VT	Tangential component of the fluid velocity at a given point	L/T
$\bar{v}_T(x)$		Circumferential average tangential velocity component	L/T
$v_T(x, \cdot)/V$	VT/V	Tangential velocity ratio	-
$\bar{v}_T(x)/V$	VTBAR	Average tangential velocity ratio	-
$(v_T(x)/V)_N$	AMPLITUDE	Amplitude (B_N for single-screw symmetric; C_N otherwise) of N th harmonic of tangential velocity component ratio	-
$v_X(x, \theta)$	VX	Longitudinal component of fluid velocity at a given point (parallel to propeller axis)	L/T
$\bar{v}_X(x)$		Circumferential average velocity component	L/D
$v_X(x, \cdot)/V$	VX/V	Axial velocity ratio	-
$\bar{v}_X(x)/V$	VXBAR	Average axial velocity ratio	-
$(v_X(x)/V)_N$	AMPLITUDE	Amplitude (A_N for single-screw symmetric; C_N otherwise) of N th harmonic of longitudinal velocity component ratio	-
$v_{X_c}(x, \theta)$		$=v_X(x, \theta) - v_T(x, \theta) \tan(x, \cdot)$ longitudinal velocity component corrected for tangential velocity	L/T
Z		Number of propeller blades	-

<u>Conventional Symbol</u>	<u>Symbol Used on Computer Plots</u>	<u>Definition</u>	<u>Units</u>
ϕ_N	PHASE ANGLE	Phase angle of the N^{th} harmonic	deg
$1-w_X(r/R)$	1-WX	Volumetric mean velocity ratio corrected for tangential velocity	-
		$1-w_X = \frac{2 \int_{r_{\text{hub}}/R}^{r/R} (\bar{v}_{X_c}(x)/V) x dx}{(r/R)^2 - (r_{\text{hub}}/R)^2}$	
		where $\bar{v}_{X_c}(x)/V =$	
		$\int_0^{2\pi} \left(\frac{v_{X_c}(x, \theta)}{2\pi V} \right) d\theta$	
$1-w_{VX}(x)$	1-WVX	Volumetric mean velocity ratio without tangential velocity correction	-
		$1-w_{VX} = \frac{2 \int_{r_{\text{hub}}/R}^{r/R} (\bar{v}(x)/V) x dx}{(r/R)^2 - (r_{\text{hub}}/R)^2}$	
$\beta(x, \theta)$		Advance angle for a given point in flow	deg
$\bar{\beta}(x)$	BBAR	Circumferential average advance angle	deg
$+\Delta\beta$	BPOS	Variation of maximum advance angle from the mean value (see sketch)	deg
$-\Delta\beta$	BNEG	Variation of the minimum advance angle from the mean value (see sketch)	deg

<u>Conventional Symbol</u>	<u>Symbol Used on Computer Plots</u>	<u>Definition</u>	<u>Units</u>
θ	ANGLE	Position angle of point in wake field, measured positive counter-clockwise from 12 o'clock, looking forward	deg



VELOCITY DIAGRAM OF BETA ANGLES

ABSTRACT

Wake survey measurements of the nominal wake behind the AO-177 hull have been obtained from model experiments and are presented for cases involving trim variations and a slight change in displacement. Extensive tables and plots of the basic wake data display the three velocity component distributions, summaries of useful averaged wake quantities, and distributions of wake harmonics. Radial distributions of a wake steepness parameter are compared for the various trim conditions, and it appears that any difficulties produced by the ship wake are not introduced by the expected changes in trim.

ADMINISTRATIVE INFORMATION

This work was funded under Naval Sea Systems Command (NAVSEA) Work Request Number OH738, Ship Performance Department Work Unit Number 1532-100.

INTRODUCTION

The AO-177 Class Naval Auxiliary Oiler is a single-screw, finely-shaped oil tanker equipped with a seven-bladed, 45-degree skewed propeller operating behind a conventional clearwater stern arrangement. Some of its principal particulars are given below.

Particulars of the AO-177

	U.S. Customary	S.I.
Waterline Length, L_{WL}	560.5 ft	170.8 m
Beam	88 ft	26.8 m
Mean Draft (Design)	32.5 ft	9.9 m
Mean Draft (Trial)	31.5 ft	9.6 m
Displacement (Design)	27,380 Tons	27,820 tonnes
Displacement (Trial)	26,390 Tons	26,810 tonnes
Propeller Diameter, D	21 ft	6.4 m
Vertical Tip Clearance, a_z	6.12 ft	1.865 m
Horizontal Tip Clearance, a_x	11.4 ft	3.48 m
Full Power Speed, V	21.5 knots	

Particulars of the AO-177 (Continued)

Form Coefficients Based on L_{WL} :

$$C_B = 0.605$$

$$C_P = 0.62$$

$$C_X = 0.975$$

$$\frac{V}{L_{WL}} = 5.24 \times 10^{-3}$$

On Builders Trials conducted in July 1980 with the first of the Class AO-177, there were experienced serious levels of airborne noise, traces of propeller blade cavitation erosion with bent trailing edge, and some unpleasant localized structural vibration levels. It is likely that all these problems can be traced in one way or another to the characteristics of fluctuating cavities that appear near the propeller blade tips as each blade passes through the large wake shadow behind the hull centered about the 12 o'clock position of the propeller disc. The details of this wake are important to understanding the sources of any propeller-excited difficulties.

As best as can be determined, the initial trial was run in a bow-down trim condition that differed somewhat from the design condition. It could not be estimated with assurance just what the effects of trim were on the wake distribution of this ship, so the experiments described in this report were performed to establish what changes were introduced by operation at the estimated trim of the trial, and also to establish any trends produced by a more extreme bow-down condition.

The estimated trial displacement of 26,390 tons (26,810 tonnes) is 96.4 percent of the original full load design value. Relative changes of static trim, away from the design condition, can be defined in terms of an angle based on the length between perpendiculars and the difference in trim.

	Trim Condition	Change in Trim Angle
Design	1.5 ft (0.457 m) x Stern	—
Trial (Estimated)	1.0 ft (0.305 m) x Bow	0.26 deg
Extreme Trim Case	3.5 ft (1.067 m) x Bow	0.92 deg

PROCEDURE

Wake survey experiments were conducted in the deep water basin of Carriage 2 using David W. Taylor Naval Ship Research and Development Center (DTNSRDC) Model 5326 which represents the AO-177 hull with a linear scale ratio of $\lambda = 25.682$. The model was run with the bilge keels attached, but with the rudder removed to accommodate the wake survey probe.

The three velocity components of the wake velocity were obtained with DTNSRDC pitot tube rake Number 8 connected to a bank of differential pressure transducers. This rake consists of five, 5-hole spherically-headed pitot tubes mounted on a foil shaped housing. The tips of the pitot tubes were located at a nominal propeller plane taken as 4.62 ft (1.408 m) aft of Station 19.5 full-scale, corresponding to the location of the plane of the original wake survey reported by Remmers and Hendrican.^{1*} As indicated in Figure 1, the experimental wake plane is 1.12 ft (0.3414 m) aft full-scale of the propeller reference plane, and happens to pass through the leading edge of the propeller swept outline at a radius of about 0.75R. Some details of the propeller geometry for DTNSRDC Propeller Number 4677 are given by Hendrican and Remmers² and Valentine and Chase.³

The radial locations of the pitot tubes are of course fixed on the model apparatus, and are expressed nondimensionally as radius ratios r/R , given for full-scale propeller diameters of 21 ft (6.4 m) and 23 ft (7.01 m) in the following table.

	Radius Ratios, r/R	
	D=21 ft (6.4 m)	D=23 ft (7.01 m)
Tube 1	0.359	0.328
Tube 2	0.556	0.508
Tube 3	0.774	0.707
Tube 4	1.017	0.929
Tube 5	1.178	1.076

The radius ratios for the 23 ft (7.01 m) diameter are useful for certain comparisons that overlap with the original AO wake survey results of Reference 1.

*References are listed on Page 9.

However, the final design propeller installed on the ship has a diameter of 21 ft (6.4 m), and all the new wake data have been analyzed on that basis. The radial positions of the pitot tube locations and relative position of the 21 ft (6.4 m) propeller disc with respect to the afterbody hull sections are shown in Figure 2.

Certain other definitions and conventions are also presented in Figure 2. When viewed looking forward, the angle θ is measured positive counterclockwise, with zero at 12 o'clock. The tangential and radial velocity components, V_T and V_R respectively, are taken as shown in Figure 2 to agree with the convention described by Hadler and Cheng.⁴

The shape of the pitot tube rake is such that the trim of the model could be altered by forces generated on the rake in various angular positions. In order to maintain the natural trim the same throughout each survey, the model was first towed free to trim with the rake in the zero degree position for each experimental condition at a speed corresponding to the ship speed of 20 knots. The model was then locked in trim for that particular experiment.

Five wake survey experiments reported on here correspond to a ship speed of 20 knots and the following ship conditions of displacement and trim.

Date	Experiment No.	Displacement (Tons)	Displacement (tonnes)	Trim (ft)	Trim (m)
May 1974	21	27,380	27,820	1.5	0.457 x Stern
Aug 1980	1	27,380	27,820	1.5	0.457 x Stern
Aug 1980	2	26,390	26,810	1.5	0.457 x Stern
Aug 1980	3	26,390	26,810	1.0	0.305 x Bow
Aug 1980	4	26,390	26,810	3.5	1.067 x Bow

The new data presented in this report are for the latter four cases (Experiments 1 through 4).

The model basin water temperature was constant throughout the test period at 74 deg Fahrenheit (23 deg Celsius).

RESULTS AND DISCUSSION

Results of the individual wake experiments are presented in both graphical and tabular form in Appendices A through D for the results of Experiments 1 through 4, respectively. In each appendix, the plotted data for the three velocity component ratios are given in the first five figures. Two additional figures in each appendix show the radial distributions of the mean velocity component ratios and the

advance angles. Five of the tables of each appendix present summary information on the several averaged velocity component ratios, advance angles, and the results of harmonic analysis for the longitudinal and tangential velocity components. The final table in each appendix presents the measured data for the three velocity component ratios at the experimental radius ratios. All the interpolated results in the appendices have been based on propeller diameter $D=21$ ft (6.4 m).

To confirm and also to establish a basis on which to correlate the wake survey data, Experiment 1 (August 1980) of the present series was performed with the same condition as Experiment 21 (May 1974) reported in Reference 1. A comparison between some results of these two experiments at a condition representing a full-scale displacement of 27,380 Tons (27,820 tonnes) and 1.5 ft (0.457 m) trim by the stern is given in Table 1 on the basis of propeller diameter $D=23$ ft (7.01 m). The repeatability of magnitudes of the various averaged wake quantities in the outer half of the propeller disc ranges from 1 to 3 percent for \bar{V}_X/V and 0.2 to 2 percent for $1-w_X$. At the innermost measurement radius, the repeatability degrades to plus or minus 6 to 8 percent for those same quantities.

The effects of trim may be displayed in several ways. Table 2 summarizes the averaged velocity ratios and other derived wake characteristics for the results of Experiments 2, 3, and 4 pertaining to a full-scale displacement of 26,390 Tons (26,810 tonnes). Plotted data for the detailed trim comparison of the three velocity component ratios versus the position angle θ are given in Figures 3 through 7. Figure 8 is a composite graph of the radial distributions of the several averaged velocity component ratios. Figure 9 shows comparisons of the radial distributions of the advance angles.

In general, it appears that the influence of trim-by-the-bow is to deepen the wake shadow very slightly, noticeable only at the inner radii. That is, the V_X/V ratios near the 12 o'clock position in the wake field are decreased systematically as the trim is changed from 1.5 ft (0.457 m) by the stern to 3.5 ft (1.067 m) by the bow. At the middle and outer radii ($r/R = 0.774$ and outward) the velocity patterns for V_X/V at the different trims become indistinguishable. These trends are reflected in the slight decreases in the circumferential average velocity ratio \bar{V}_X/V and the volumetric mean velocity ratio $1-w_X$ with increasing trim-by-the-bow. The effect on the magnitudes of both \bar{V}_X/V and $1-w_X$ ranges from about 7 to 8 percent decrease at the innermost radius to about 2 percent decrease at the outermost radii.

In Figure 9, the effect of trim on mean angle of advance $\bar{\beta}$ is negligible even at the inner radius ratios. The variations of minimum and maximum advance angles from the mean, $-\Delta\beta$ and $+\Delta\beta$, indicate that increasing trim-by-the-bow tends to exaggerate slightly the deviations of advance angle on the negative side. Qualitatively, this indicates that increasing bow down trim produces slight increases in the mean foil section angles of attack, so that suction side cavitation would be slightly exaggerated as well. Quantitatively, the effect is very small, and is confined to the inner radii of the propeller disc.

It is interesting to try to discern any effect of trim on the wake harmonics. Figures 10 and 11 contain a series of composite plots for the harmonics $N=1$ through 8, showing the radial distributions of harmonic amplitudes for the longitudinal velocity component V_x/V . Figures 12 and 13 contain the same series of comparisons for the tangential velocity component V_T/V . This presentation is useful for displaying the harmonic amplitude as well as the location of changes of phase angle (where the amplitude curve crosses zero). From these graphs, the measurable influence of trim on the harmonic amplitudes is only slight, and confined to the lower harmonic orders.

For the case of the longitudinal velocity component harmonics (Figures 10 and 11), there is a slight increase in the magnitude of the harmonic amplitude with increasing bow-down trim, at the inner radii ($r/R < 0.6$) for the orders $N=1$ and 2. This is reversed for $N=3$, and appears mixed for the higher orders. This is consistent with the slight deepening of the V_x/V velocity defect at the inner radii that was noted previously.

For the case of the tangential velocity component harmonics (Figures 12 and 13), the effect of increasing bow-down trim is to reduce the harmonic amplitude at the inner radii for $N=1$. The higher harmonic orders show no particular effect of trim.

In general terms, the wake of the AO-177 has a definite spike-like character, with a deep velocity defect and rather steep velocity gradients centered about the 12 o'clock position. These are prominent features evident in Figure 14 which is an iso-wake contour plot corresponding to the trial condition of 26,390 Tons (26,810 tonnes) displacement and 1 ft (0.305 m) trim by the bow. One measure of the relative steepness of such a wake has been suggested by Odabasi and Fitzsimmons,⁵ who have defined a parameter describing the local wake gradient per unit axial velocity at a given radius.

$$G_W = \frac{1}{r/R} \left| \frac{dw/d\theta}{1-w} \right|$$

where the local circumferential wake slope is $dw/d\theta$, θ is in radians, and w is the local wake.

It has been proposed in Reference 5, as part of a complete wake criterion for avoiding problems with cavitation-induced pressure fluctuations, that the wake gradient parameter should satisfy

$$G_W < 1.0$$

inside the angular interval θ_B ($\frac{1}{2}\theta_B$ to either side of the 12 o'clock position), and for radii ranging from $0.7R$ to $1.15R$. The angular interval θ_B in degrees depends on the number of propeller blades

$$\theta_B = \frac{360}{Z} + 10$$

so for the seven-bladed A0 propeller, $\theta_B = 61.4$ degrees. This criterion is supposed to be based in part on wake data from model experiments of ships with known propeller-excited vibration problems. The complete wake assessment scheme of Reference 5 has become known as the British Ship Research Association (BSRA) wake quality criterion, and although it has been widely quoted, there is as yet relatively little published evidence on how well it works. Only the wake gradient part of it is considered here, and is used only to indicate the relative effects of trim.

Figure 15 is a composite plot of the radial distribution of the wake gradient parameter G_W for the A0-177, showing the effects of trim changes compared with the original design case of displacement and trim. The G_W values are shown along the ray $\theta = 30^\circ$ which is near the edge of the recommended angular interval $\frac{1}{2}\theta_B$, and passes through the characteristic wake slope feature of the prominent hull wake shadow. On the basis of the gradient parameter, it appears that the wake of this ship is quite steep. The G_W exceeds 1.0 everywhere in the interval $0.7R \leq r \leq 1.15 R$. It is fair to state that this constitutes a single indicator of a possibly troublesome wake, but without firmer connection between wake steepness and fluctuating propeller-induced pressures and resulting hull vibration or airborne

noise problems, a general conclusion is premature. What is useful to the present discussion is to note that trim seems to exert a relatively minor effect on this steepness tendency in the crucial outer radius region. It appears that bow-down trim variation in the range considered should not introduce any substantially worse wake characteristics than exist already with the wake produced in the original trim condition. It should be noted that at the time of the original AO-177 wake experiments, wake quality assessment schemes were not available for making preliminary judgments on the possibility of problems with the interaction of steep wakes and propeller-induced fluctuating pressures.

CONCLUSIONS

1. Extensive wakedata are made available for future investigations of the AO-177 propeller-wake interaction problems.
2. There are noticeable, consistent, but small changes in the velocity patterns and harmonic amplitudes of the wake of the AO due to trim-by-the-bow. The effects are discernible mainly at the inner radii (typically $r/R < 0.6$) and are confined to the region centered about the 12 o'clock position in the wake. The changes due to the maximum trim excursion of 0.52 deg are only slightly larger than the typical repeatability bounds for a wake survey. The effect of a trim change of 0.26 deg appears to be within the error bounds inherent in a wake survey experiment.
3. Radial distributions of the wake gradient parameter $G_w > 1.0$ for this wake shows that difficulties from unsteady cavitation pressure pulses may be expected. The effect on G_w of the trim change of 0.26 deg is small, and would not be expected to produce new or different problems with the wake.

ACKNOWLEDGMENTS

The authors are grateful to Ms. Rae Hurwitz and Mr. Stewart Jessup for performing the analyses of the wake survey data. The assistance of Mr. Mark Vranicar in preparing the graphs is also appreciated.

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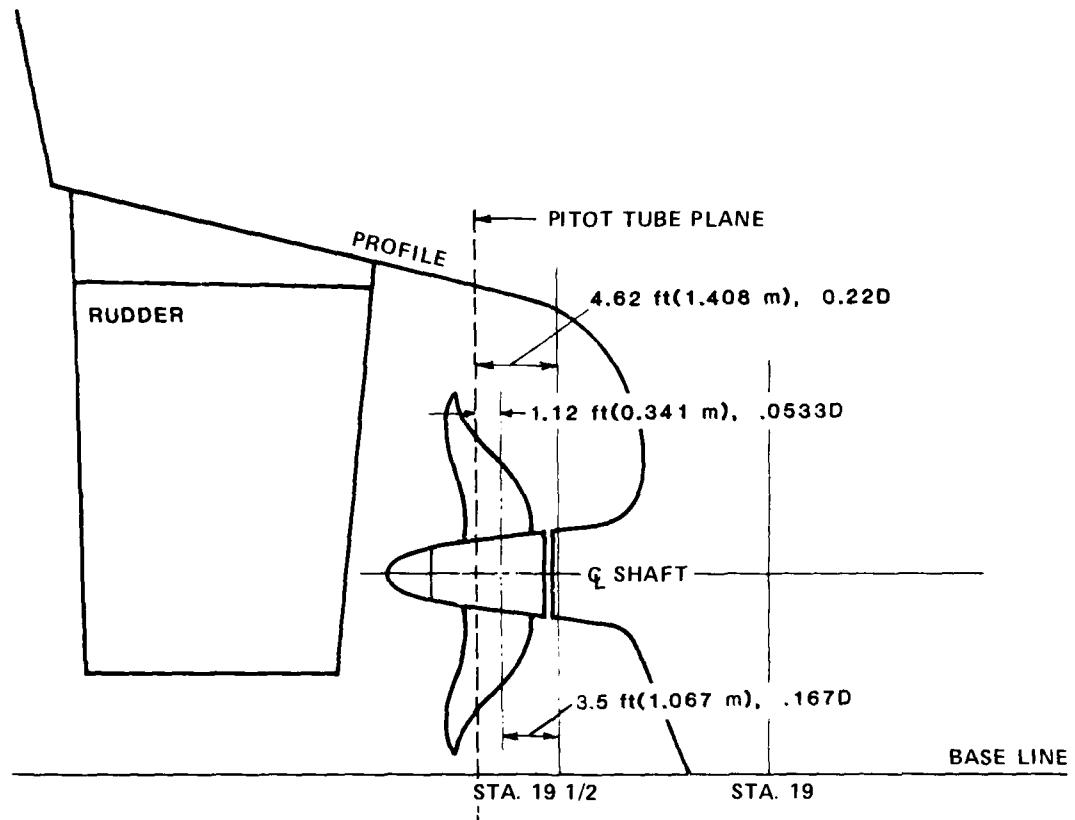


Figure 1 - AO 177 Wake Rake Pitot Tube Location in Relation to the 21-Foot (6.4 m) Design Propeller

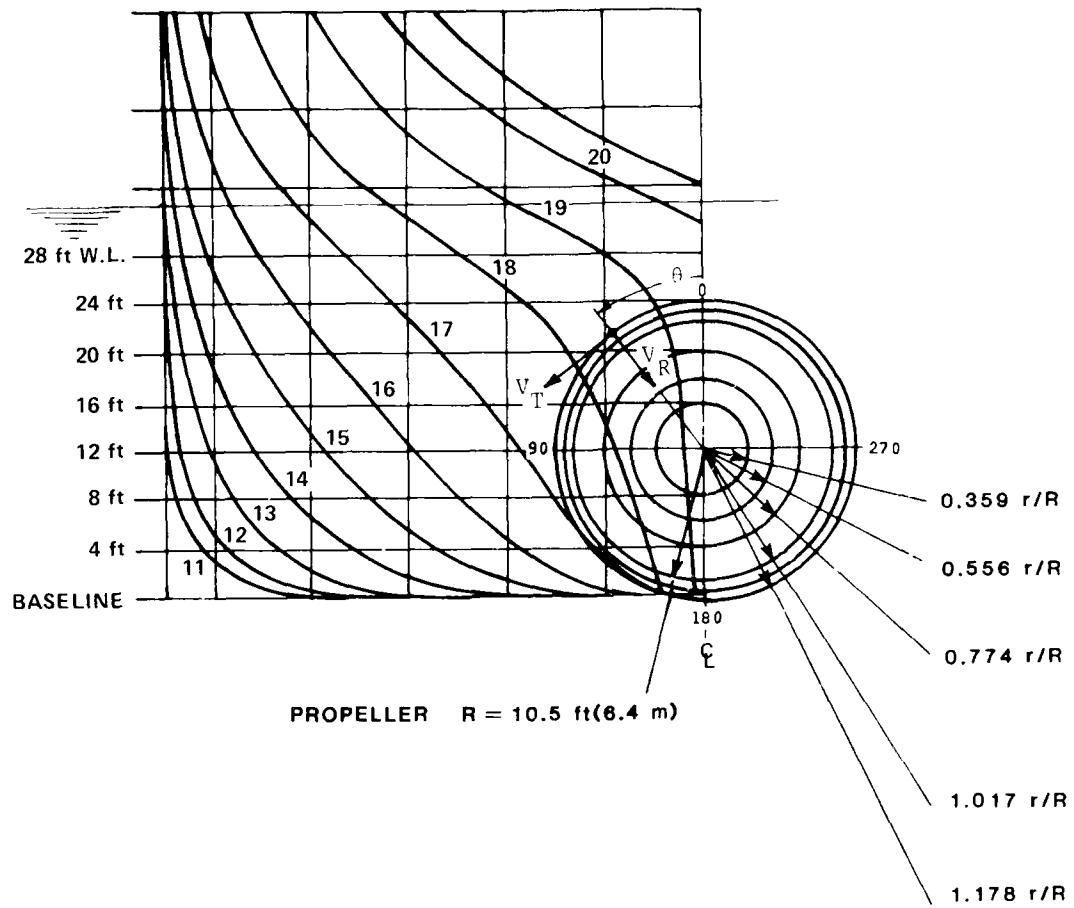


Figure 2 - AO 177 STERN BODY PLAN WITH TEST RADII

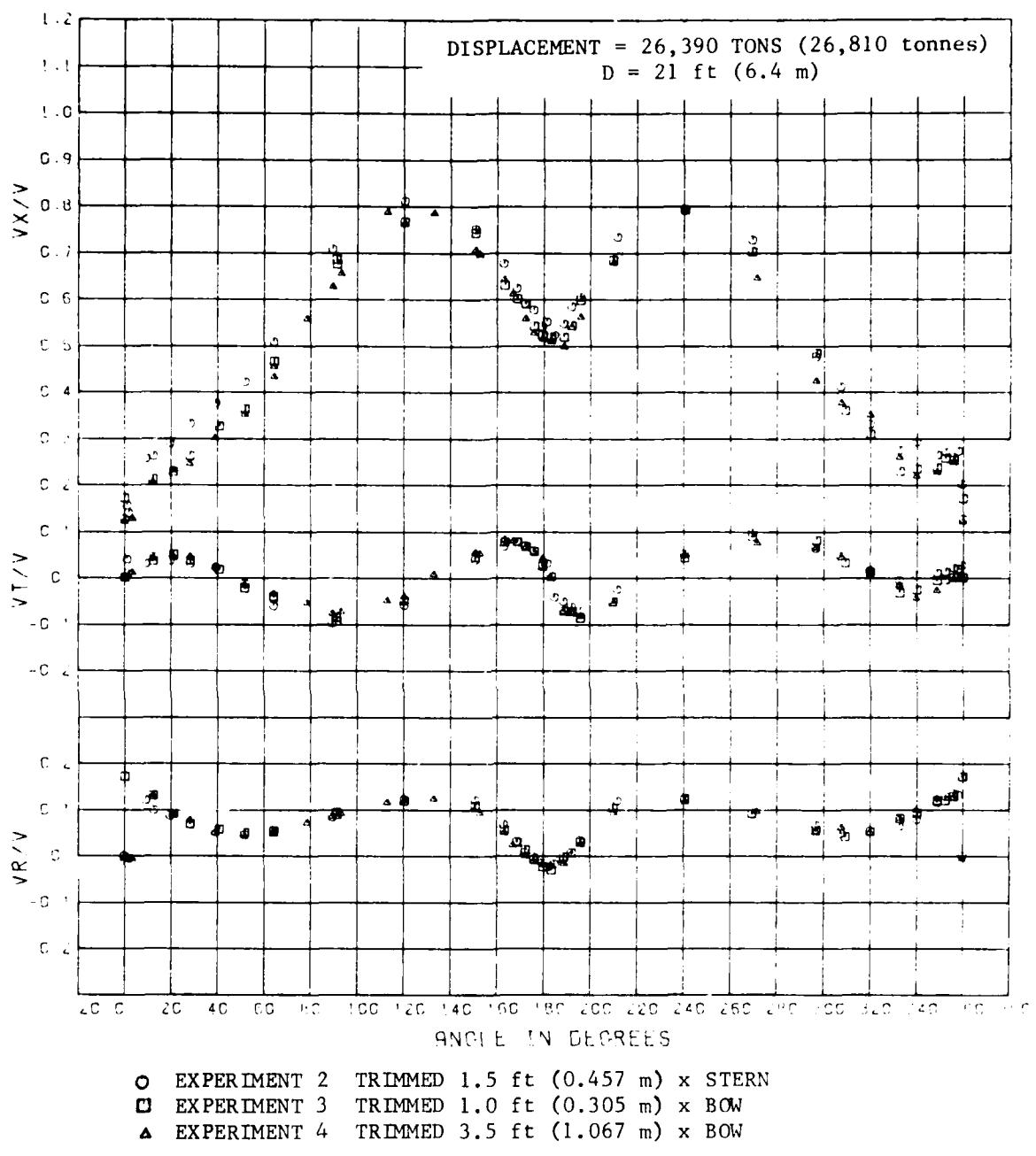


Figure 3 - CIRCUMFERENTIAL DISTRIBUTION OF THE LONGITUDINAL, TANGENTIAL AND RADIAL VELOCITY COMPONENT RATIOS - RADIUS RATIO = 0.359
 COMPOSITE OF EXPERIMENTS 2, 3, AND 4

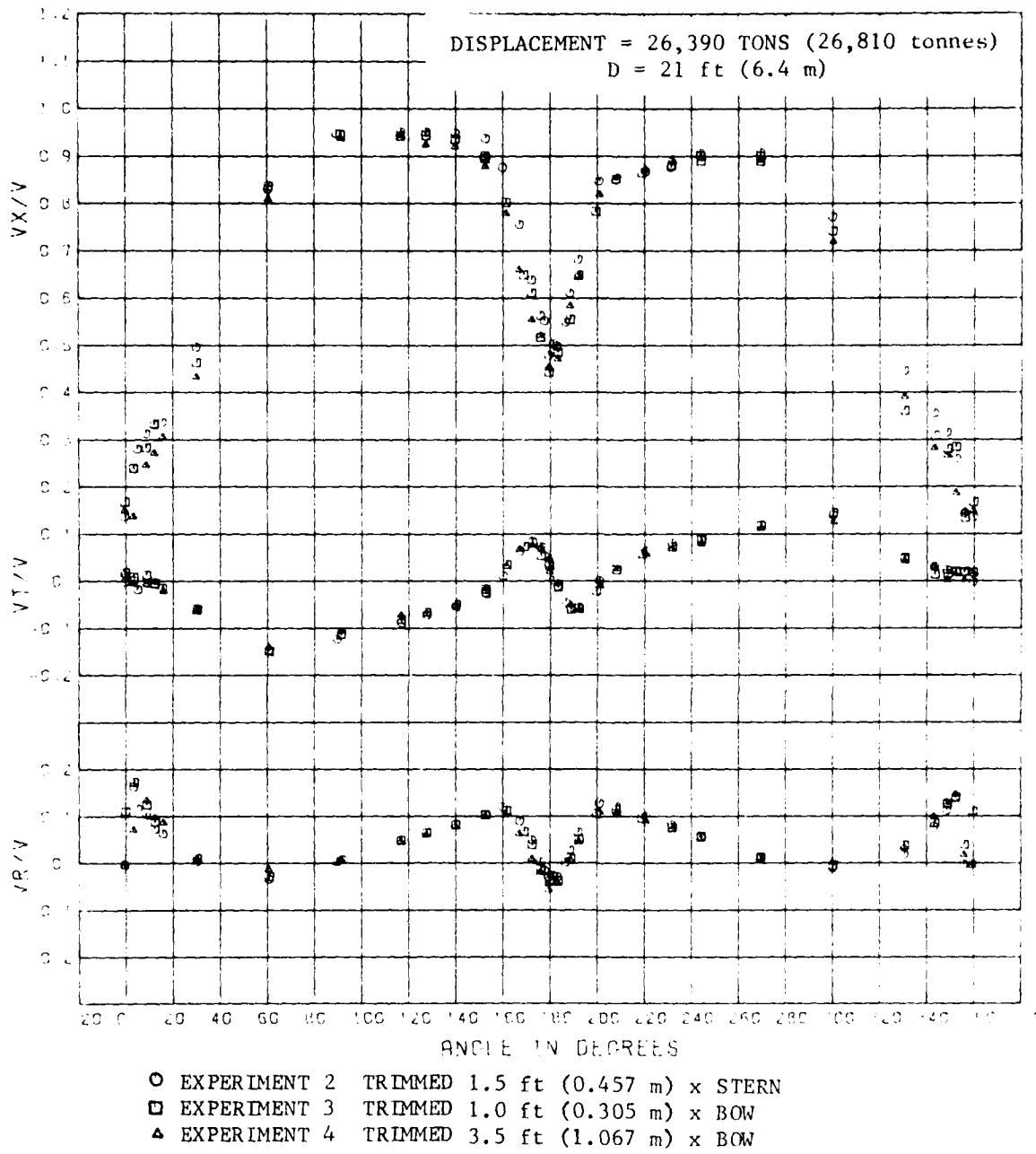


Figure 4 - CIRCUMFERENTIAL DISRTIBUTION OF THE LONGITUDINAL, TANGENTIAL AND RADIAL VELOCITY COMPONENT RATIOS - RADIUS RATIO = 0.556
 COMPOSITE OF EXPERIMENTS 2, 3, AND 4

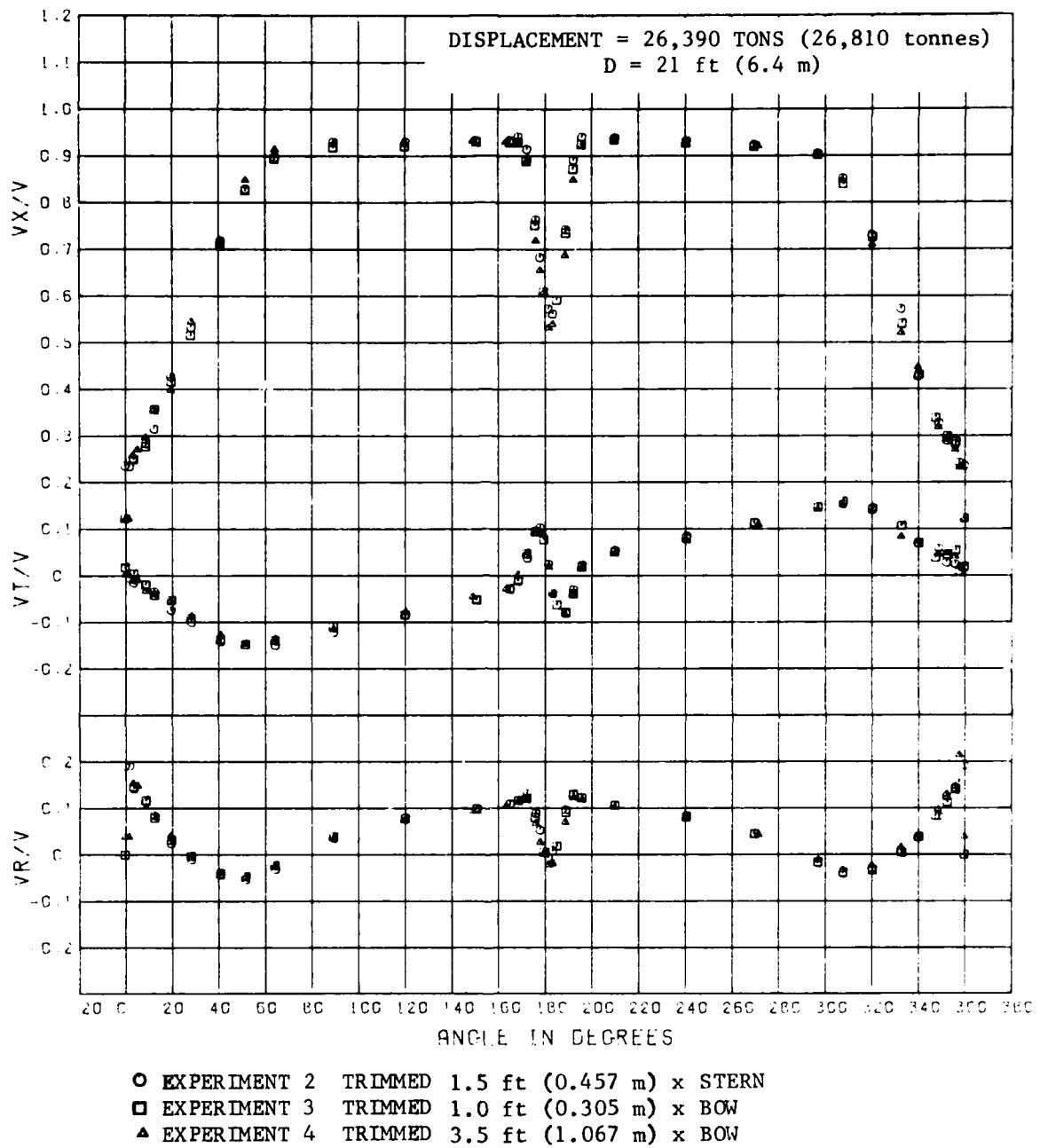


Figure 5 - CIRCUMFERENTIAL DISTRIBUTION OF THE LONGITUDINAL, TANGENTIAL AND RADIAL VELOCITY COMPONENT RATIOS - RADIUS RATIO = 0.774
 COMPOSITE OF EXPERIMENTS 2, 3, AND 4

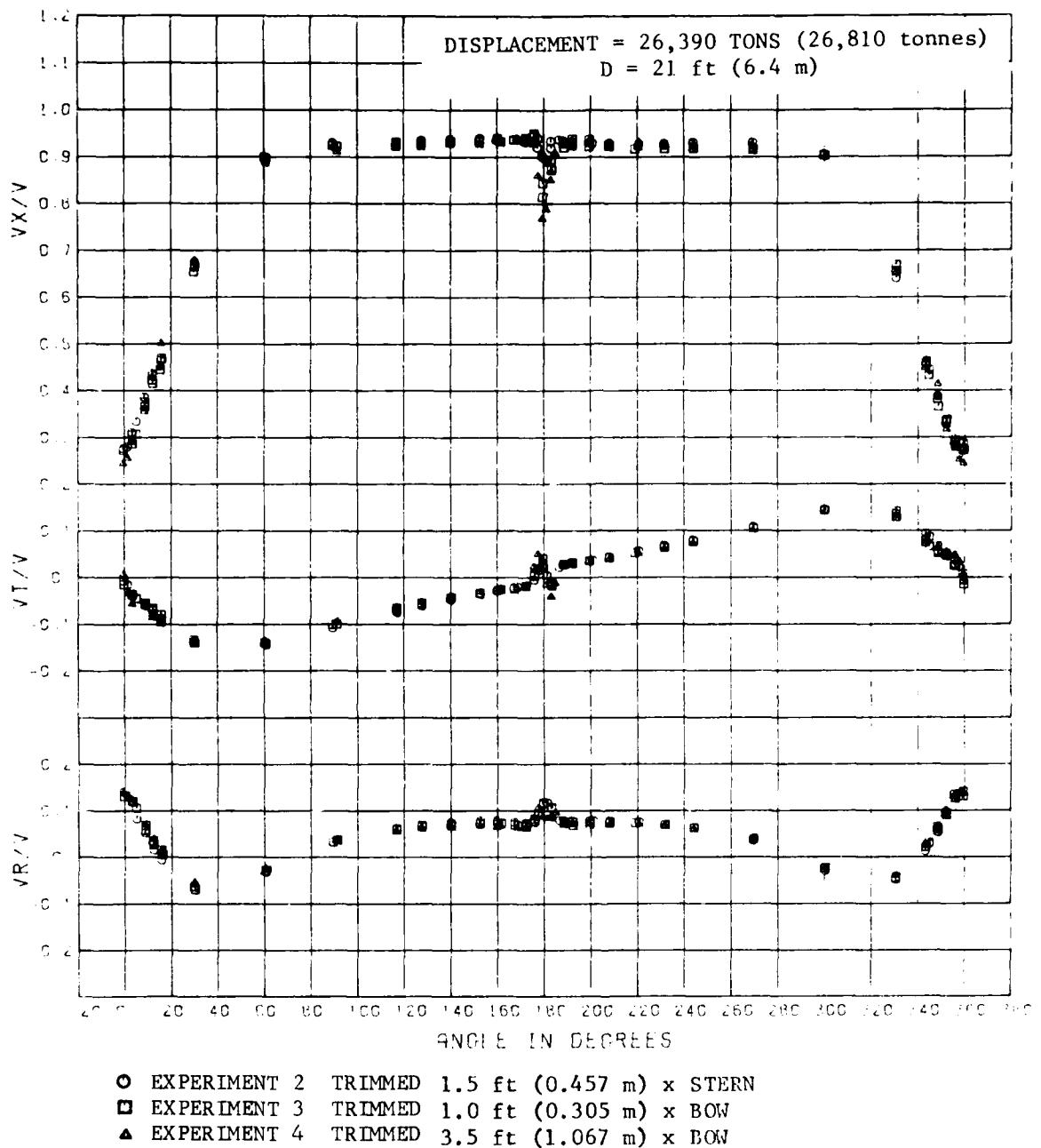


Figure 6 - CIRCUMFERENTIAL DISTRIBUTION OF THE LONGITUDINAL, TANGENTIAL AND RADIAL VELOCITY COMPONENT RATIOS - RADIUS RATIO = 1.017
 COMPOSITE OF EXPERIMENTS 2, 3, AND 4

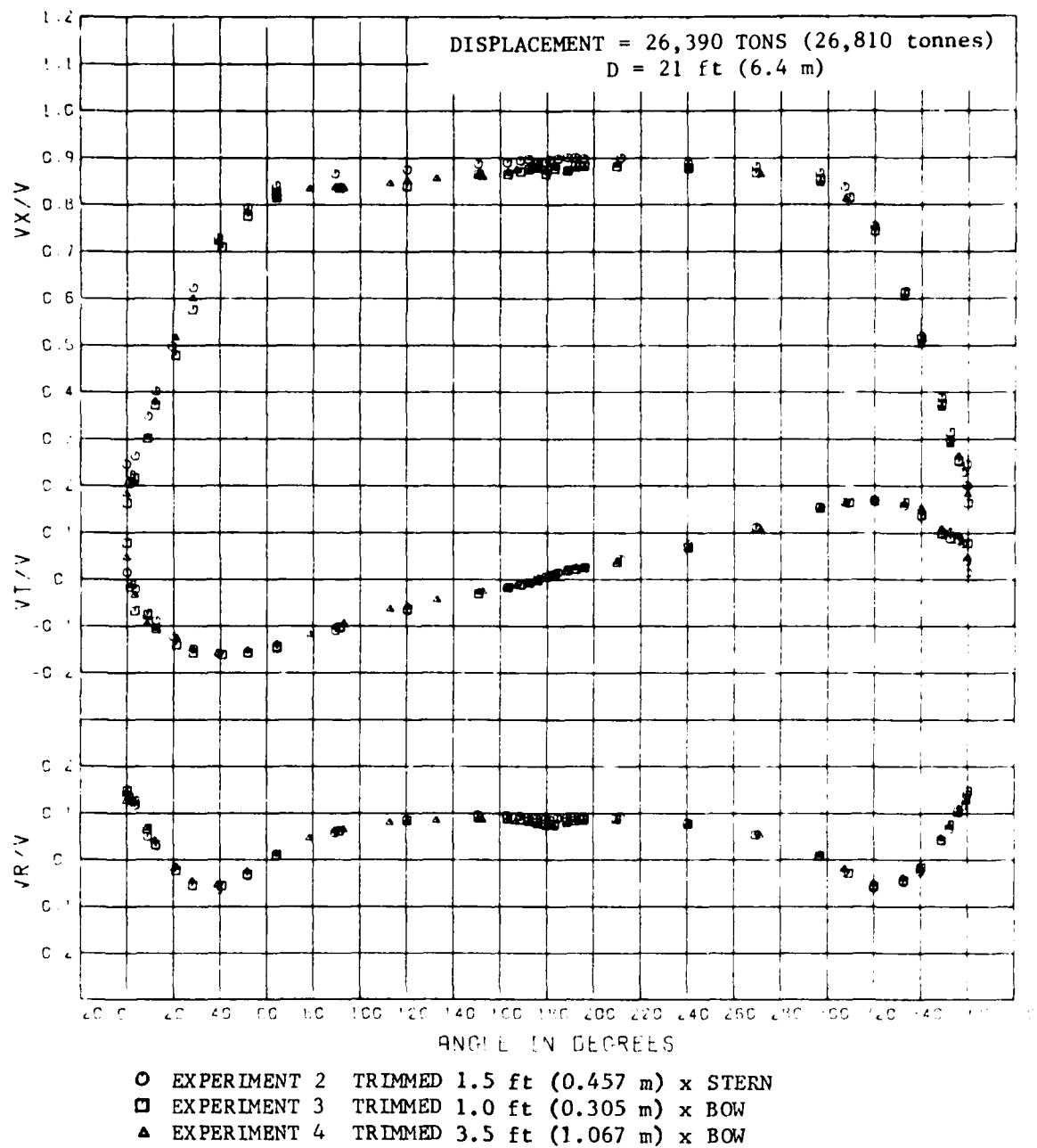


Figure 7 - CIRCUMFERENTIAL DISTRIBUTION OF THE LONGITUDINAL, TANGENTIAL AND RADIAL VELOCITY COMPONENT RATIOS - RADIUS RATIO = 1.178
 COMPOSITE OF EXPERIMENTS 2, 3, AND 4

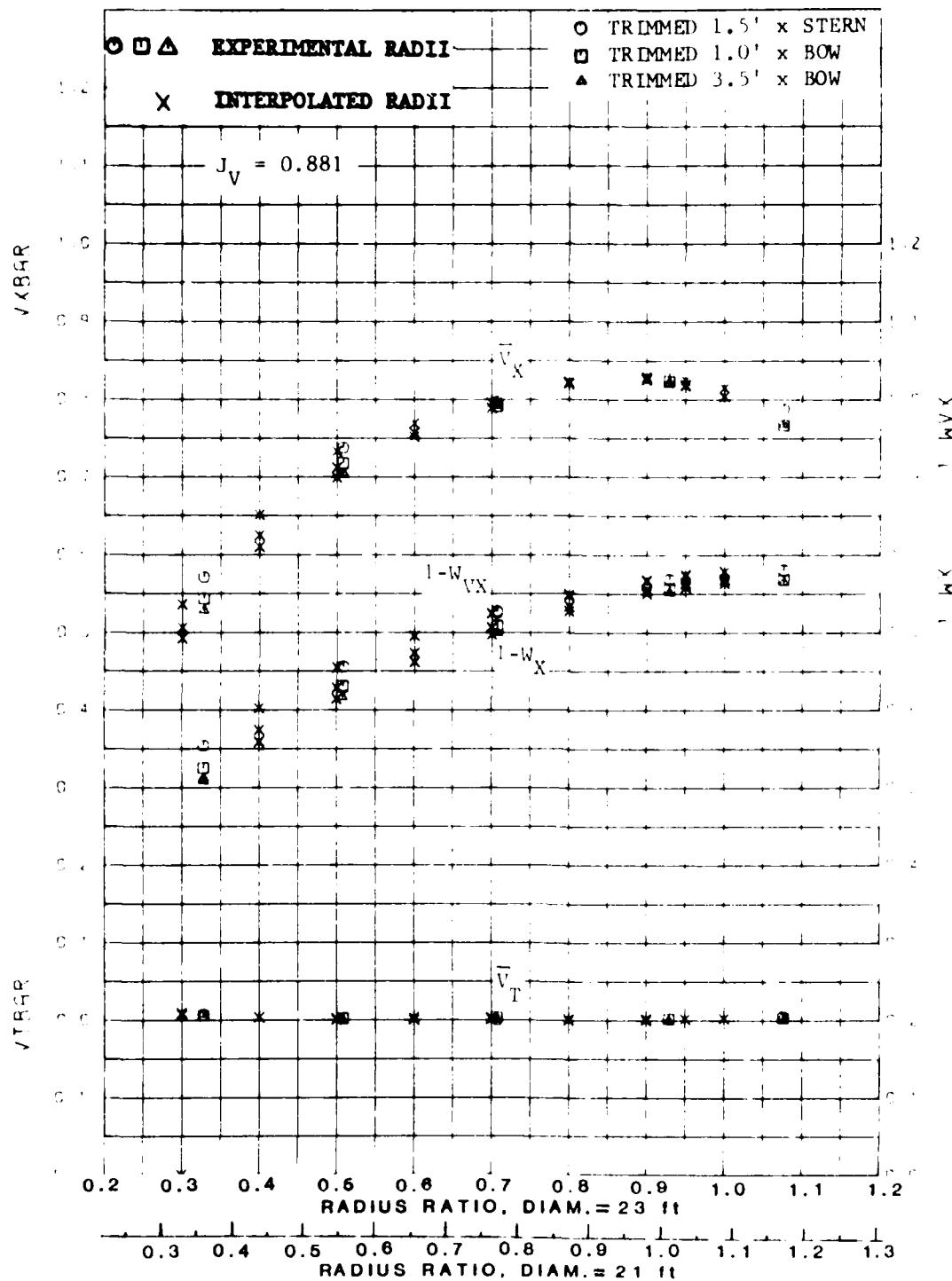


Figure 8 - RADIAL DISTRIBUTION OF THE MEAN VELOCITY COMPONENT RATIOS,
COMPOSITE OF EXPERIMENTS 2, 3, AND 4
DISPLACEMENT 26,390 TONS (26,810 tonnes)

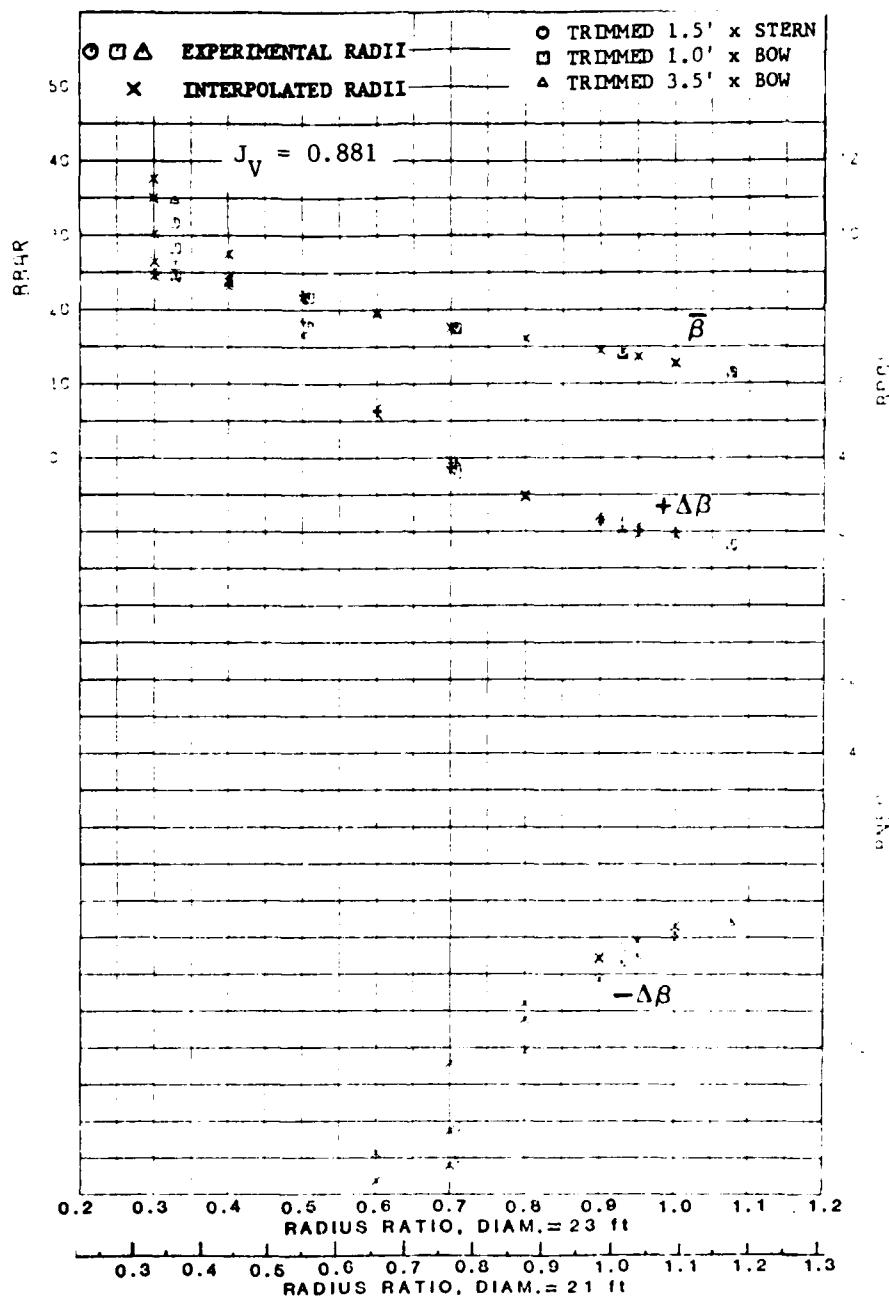


Figure 9 - RADIAL DISTRIBUTION OF THE MEAN ADVANCE ANGLE AND THE MAXIMUM VARIATIONS OF THE ADVANCE ANGLE FOR MODEL 5326
 COMPOSITE OF EXPERIMENTS 2, 3, AND 4
 DISPLACEMENT 26,390 TONS (26,810 tonnes)

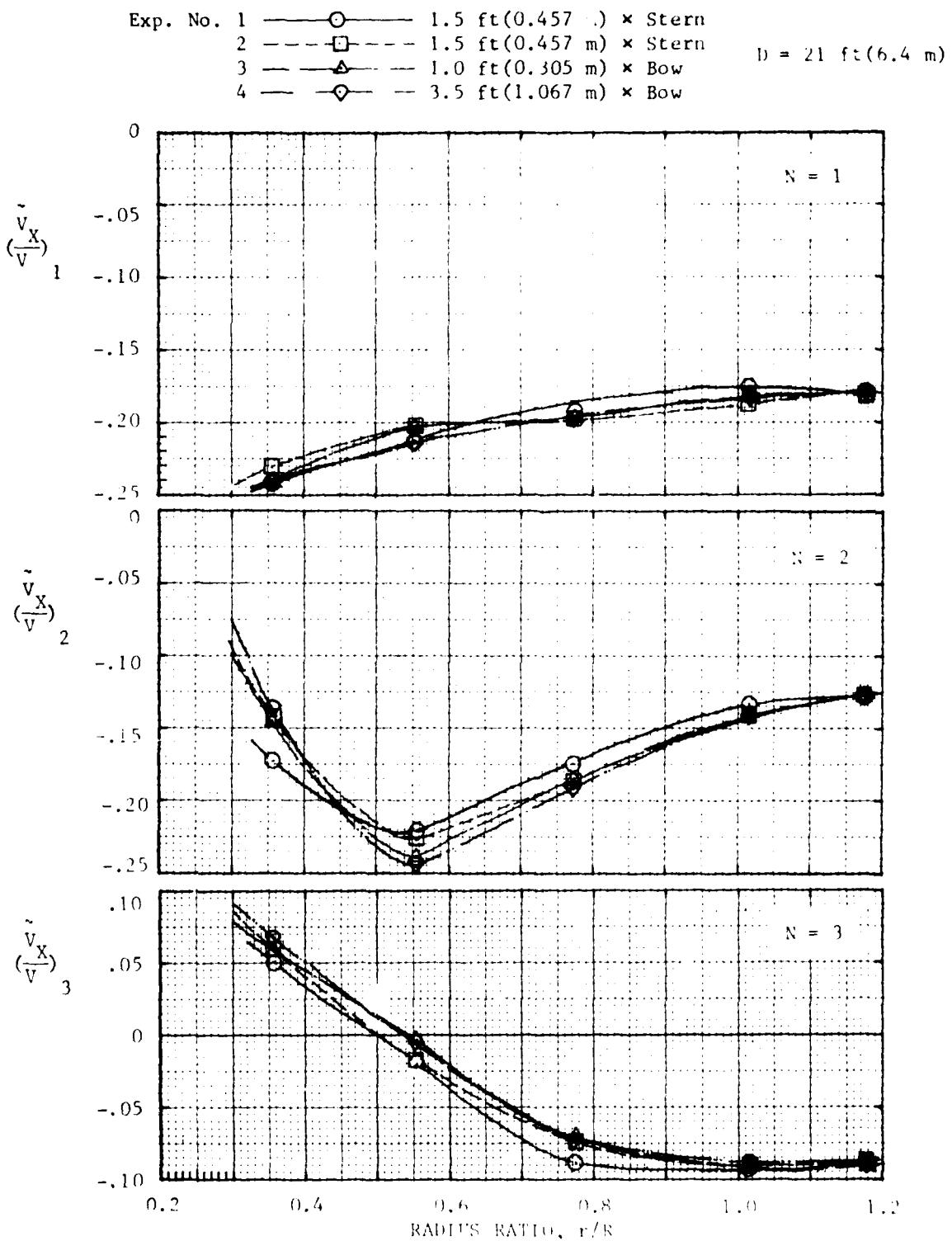


Figure 10 - RADIAL DISTRIBUTIONS OF THE HARMONIC AMPLITUDES \tilde{v}_x/v_N OF THE LONGITUDINAL VELOCITY COMPONENT, FOR $N = 1$ THROUGH 3

Exp. No. 1 —○— 1.5 ft (0.457 m) × Stern
 2 —□— 1.5 ft (0.457 m) × Stern
 3 —△— 1.0 ft (0.305 m) × Bow
 4 —◇— 3.5 ft (1.067 m) × Bow

$D = 21$ ft (6.4 m)

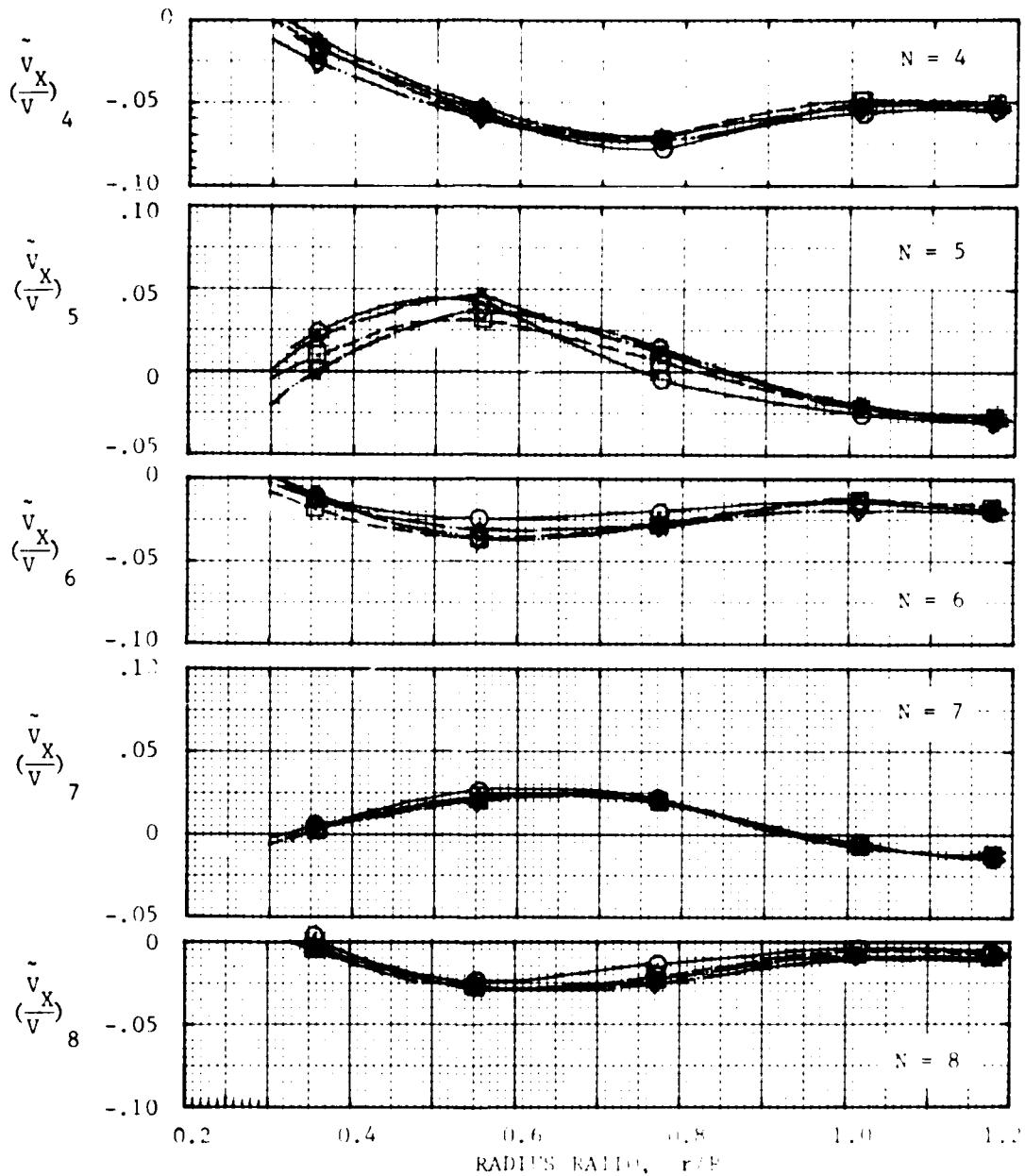


Figure 11 - RADIAL DISTRIBUTIONS OF THE HARMONIC AMPLITUDES $(v_x/v)_N$ OF THE LONGITUDINAL VELOCITY COMPONENT, FOR $N = 4$ THROUGH 8

Exp. No. 1 —○— 1.5 ft(0.457 m) × Stern
 2 —□— 1.5 ft(0.457 m) × Stern
 3 —△— 1.0 ft(0.305 m) × Bow
 4 —◇— 3.5 ft(1.067 m) × Bow

$D = 21$ ft(6.4 m)

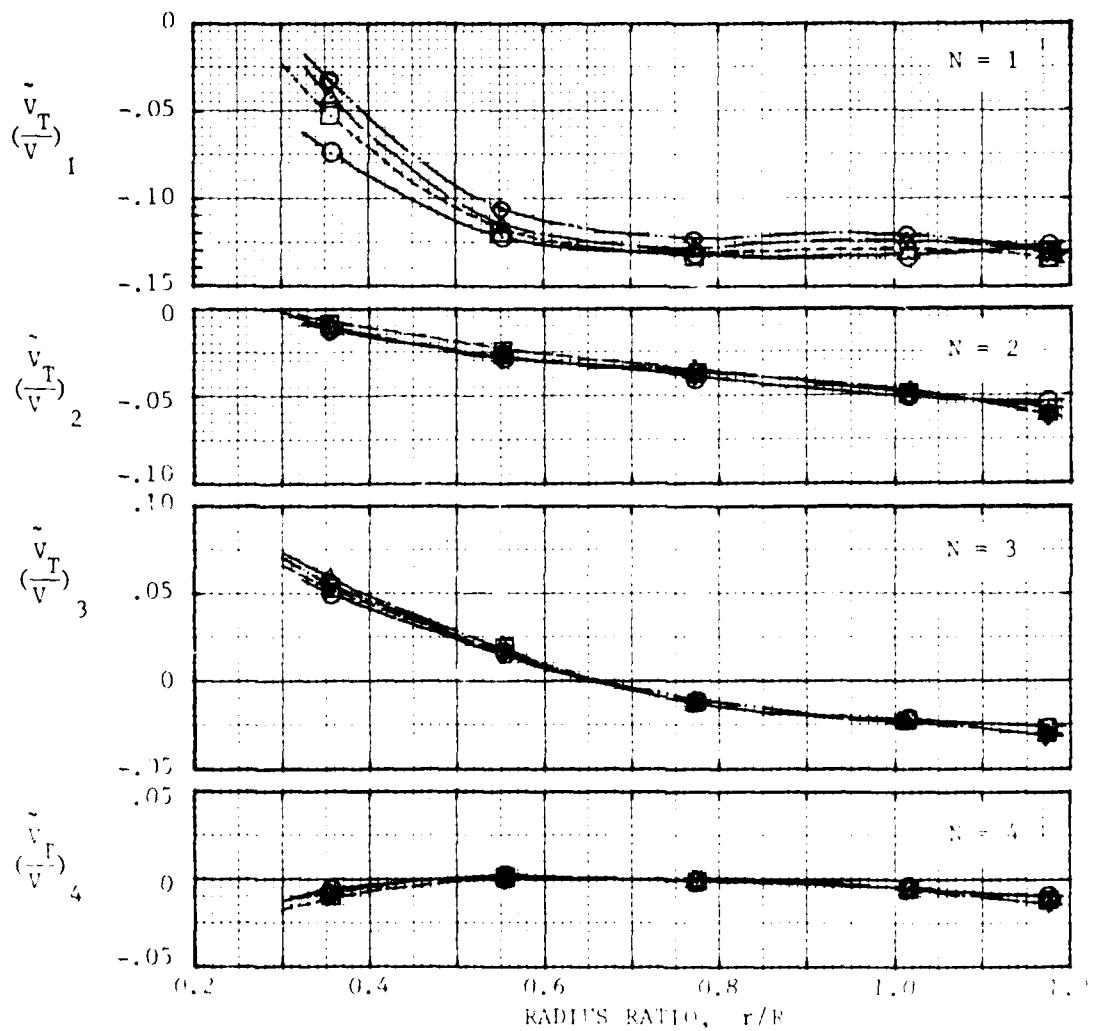


Figure 12 - RADIAL DISTRIBUTIONS OF THE RAYLEIGH AMPLITUDE, C_1/V_T , OF THE TANGENTIAL VELOCITY COMPONENT, FOR $N = 1$ THROUGH 4

Exp. No. 1 —○— 1.5 ft(0.457 m) × Stern
 2 —□— 1.5 ft(0.457 m) × Stern $D = 21$ ft(6.4 m)
 3 —△— 1.0 ft(0.305 m) × Bow
 4 —◇— 3.5 ft(1.067 m) × Bow

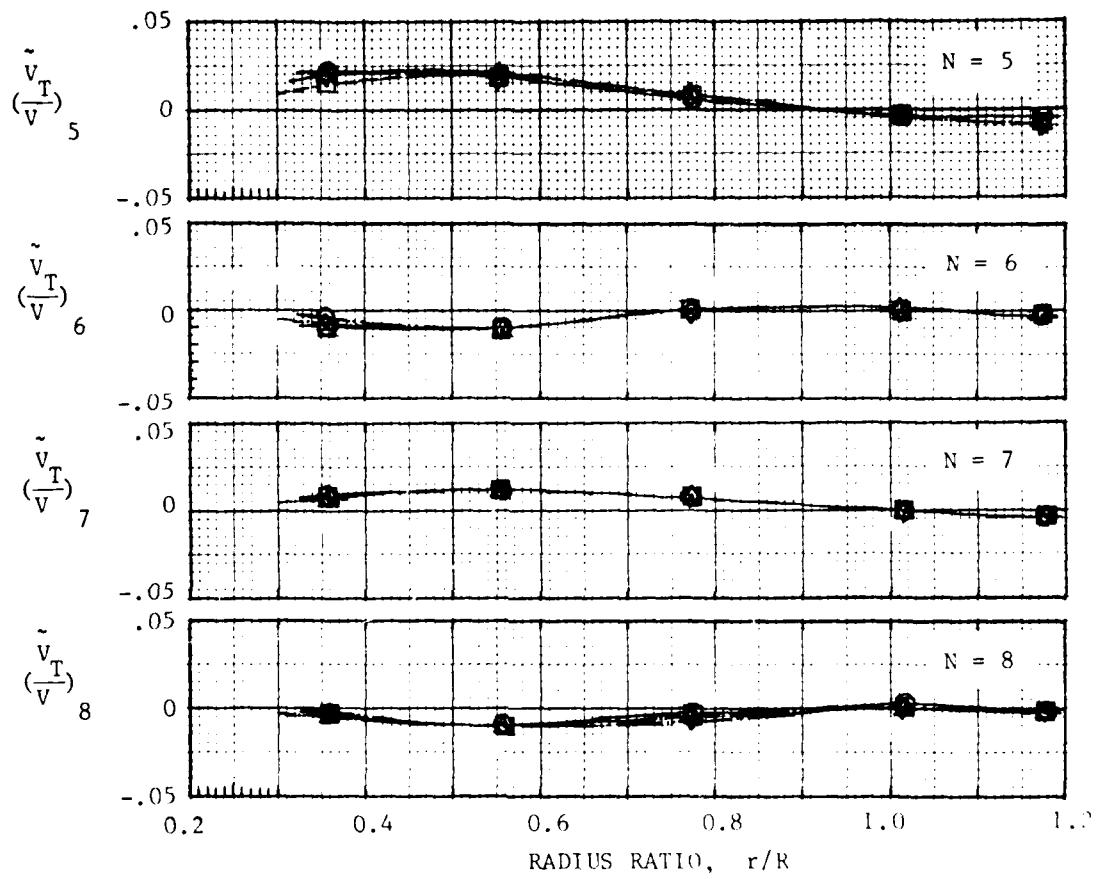


Figure 13 - RADIAL DISTRIBUTIONS OF THE HARMONIC AMPLITUDES $(V_T/V)_N$ OF THE TANGENTIAL VELOCITY COMPONENT, FOR $N = 5$ THROUGH 8

26,390 Tons (26,810 tonnes) DISPLACEMENT
1.0 ft (0.305 m) TRIM BY THE BOW

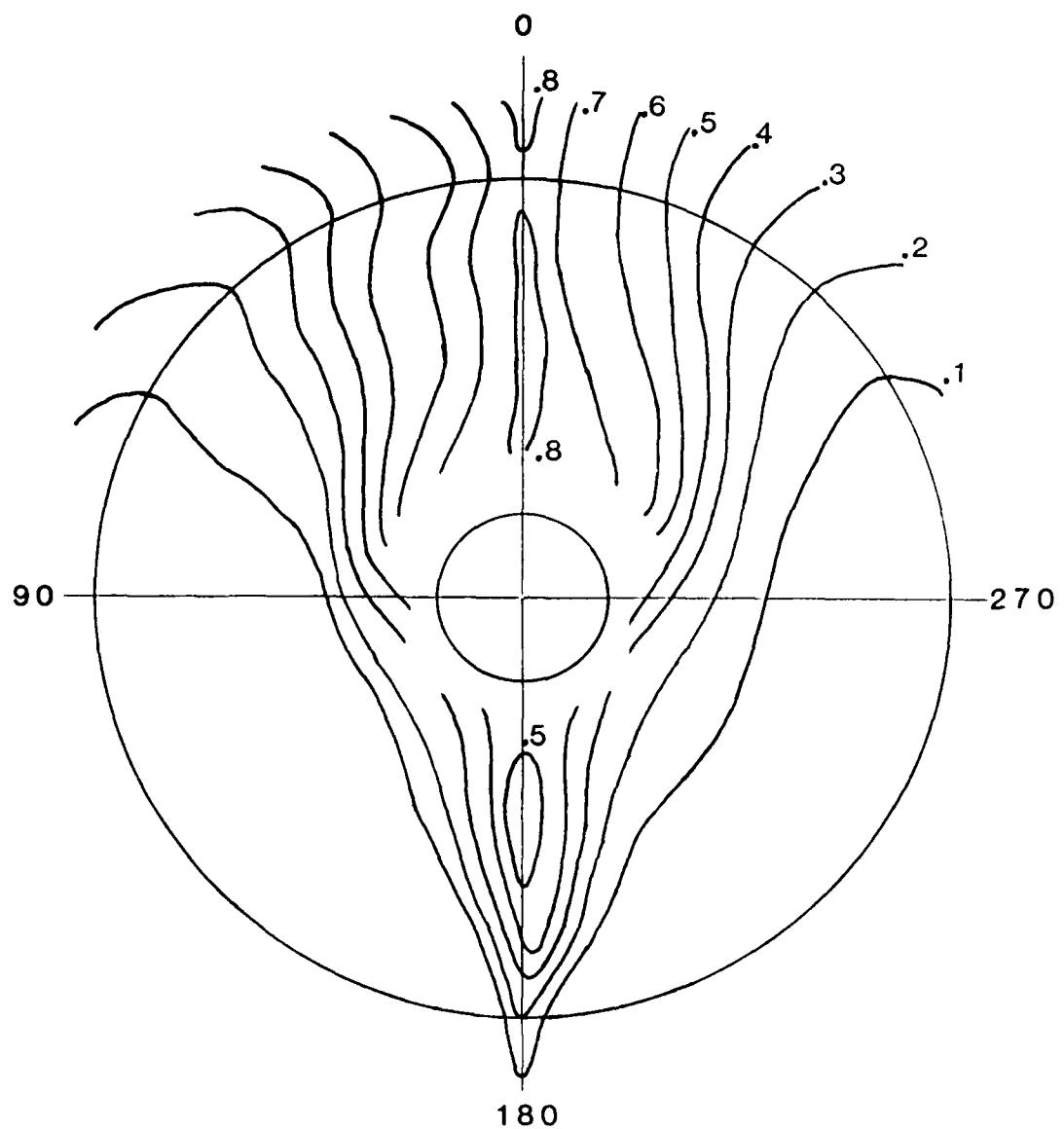


Figure 14 - CONTOUR PLOT OF THE LONGITUDINAL COMPONENT ISO-WAKE
 $w = (1 - V_x / V) / X$ FOR THE AO 177 (MODEL 5326)

- \square — 1.5 ft (0.457 m) TRIM x STERN, DISPL = 27,380 Tons (27,820 tonnes)
- \blacksquare --- 1.5 ft (0.457 m) TRIM x STERN, DISPL = 26,390 Tons (26,810 tonnes)
- \circ — 1.0 ft (0.305 m) TRIM x BOW, DISPL = 26,390 Tons (26,810 tonnes)
- \triangle — 3.5 ft (1.067 m) TRIM x BOW, DISPL = 26,390 Tons (26,810 tonnes)

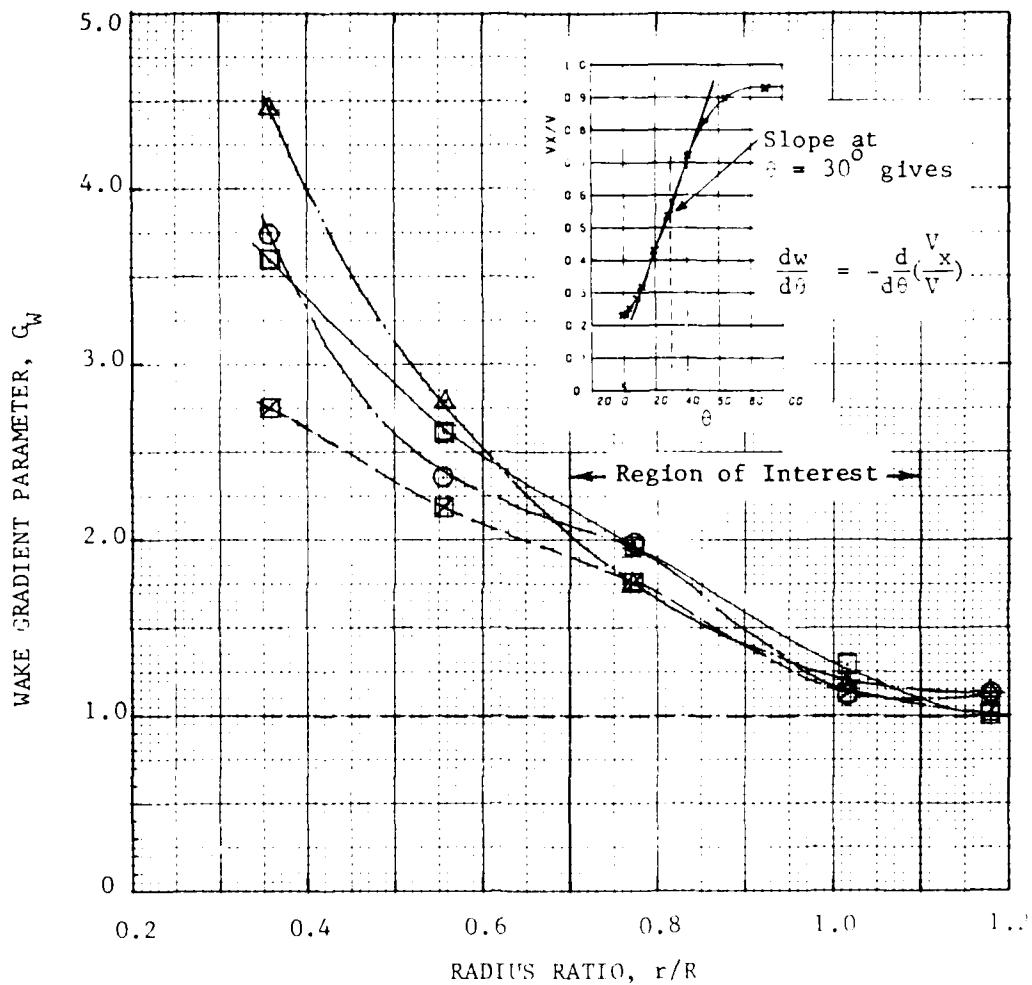


Figure 15 - RADIAL DISTRIBUTION OF THE WAKE GRADIENT PARAMETER
FOR THREE TRIM CONDITIONS

(Taken along the ray $\theta = 30$ degrees)

Table 1 COMPARISON OF THE MEAN VELOCITY COMPONENT RATIOS AND OTHER DERIVED QUANTITIES FOR EXPERIMENT 21 (MAY 1974) AND EXPERIMENT 1 (AUG. 1980)

D = 23 ft (7.01 m), $J_V = 0.881$

EXPERIMENT	RADIUS	V _{XBAR}	V _{TBAR}	V _{RBAR}	-WX	BBAR	BPOS	BNEG
21	.328	.615	.000	.054	.05	27.72	10.34	-16.14
1		.577	.007	.084	.558	26.12	10.61	-19.17
21	.508	.741	.000	.046	.680	22.25	6.38	-14.33
1		.750	-.001	.039	.678	22.60	7.33	-14.88
21	.707	.814	.000	.043	.741	17.38	3.46	-11.99
1		.797	.000	.049	.738	17.55	3.77	-14.68
21	.929	.817	.000	.030	.779	13.85	2.10	-8.80
1		.839	-.002	.027	.777	14.21	2.38	-9.30
21	1.076	.839	.000	.041	.792	12.33	1.71	-3.36
1		.817	.001	.044	.791	12.01	1.69	-3.55

EXPERIMENT 21 (MAY 1974)

An instrumentation problem with tube 5 ($r/R = 1.076$) gave erroneous data between 250° and 350° . Because of the symmetry of the hull (as verified by tubes 1 through 4) predicted values for V_X/V , V_T/V , and V_R/V , were used between 250° and 350° for $r/R = 1.076$.

V_{XBAR} IS CIRCUMFERENTIAL MEAN LONGITUDINAL VELOCITY.

V_{TBAR} IS CIRCUMFERENTIAL MEAN TANGENTIAL VELOCITY.

V_{RBAR} IS CIRCUMFERENTIAL MEAN RADIAL VELOCITY.

1-V_X IS VOLUNTARY MEAN WAKE VELOCITY WITH TANGENTIAL CORRECTION.

BBAR IS MEAN ANGLE OF ADVANCE, THE MAXIMUM AND MEAN ADVANCE ANGLES (DELTA, BETA PLUS).

BPOS IS VARIATION BEYOND THE MAXIMUM AND MEAN ADVANCE ANGLES (DELTA, BETA MINUS).

BNEG

IS VARIATION BEYOND THE MINIMUM AND MEAN ADVANCE ANGLES (DELTA, BETA MINUS).

Table 2 - COMPARISON OF THE MEAN VELOCITY COMPONENT RATIOS AND OTHER DERIVED QUANTITIES
FOR EXPERIMENTS 2, 3, AND 4, DISPLACEMENT 26,390 TONS (26,810 tonnes)

$$D = 21 \text{ ft (6.4 m)}, J_V = 0.881$$

EXPERIMENT	TRIM (feet)	RADIUS	VXBAR	VTBAR	VRBAR	1-WX	BBAR	BPOS	BNEG
2	1.5 x stern	0.359	.571	.007	.082	.554	25.90	10.33	-18.98
3	1.0 x bow	.543	.005	.085	.525	24.80	9.65	-16.41	
4	3.5 x bow	.528	.007	.078	.508	24.15	10.93	-18.23	
2	1.5 x stern	0.556	.738	.001	.038	.656	22.16	7.20	-18.52
3	1.0 x bow	.738	.002	.040	.631	21.59	7.53	-18.18	
4	3.5 x bow	.735	.000	.038	.616	21.27	7.63	-17.13	
2	1.5 x stern	0.774	.797	.000	.048	.729	17.55	3.73	-12.30
3	1.0 x bow	.791	.003	.046	.709	17.41	3.57	-14.15	
4	3.5 x bow	.792	.001	.047	.700	17.42	3.84	-15.07	
2	1.5 x stern	1.017	.828	.002	.029	.749	14.03	2.15	-9.31
3	1.0 x bow	.823	.002	.030	.757	13.95	2.25	-9.26	
4	3.5 x bow	.821	.000	.030	.752	13.92	2.02	-9.74	
2	1.5 x stern	1.178	.788	.005	.044	.781	11.59	1.59	-8.21
3	1.0 x bow	.767	.003	.043	.768	11.30	1.64	-8.89	
4	3.5 x bow	.768	.005	.043	.765	11.31	1.67	-8.62	

VBAR IS CIRCUMFERENTIAL MEAN LONGITUDINAL VELOCITY.

VTBAR IS CIRCUMFERENTIAL MEAN TANGENTIAL VELOCITY.

VRBAR IS CIRCUMFERENTIAL MEAN RADIAL VELOCITY.

1-WX IS VOLVETRIC MEAN WAKE VELOCITY WITH TANGENTIAL CORRECTION.

BBAR IS MEAN ANGLE OF ADVANCE.

BPOS IS VARIATION BETWEEN THE MAXIMUM AND MEAN ADVANCE ANGLES (DELTA BETA PLUS).

BNEG IS VARIATION BETWEEN THE MINIMUM AND MEAN ADVANCE ANGLES (DELTA BETA MINUS).

APPENDIX A
RESULTS OF EXPERIMENT 1

Corresponding to
Trim 1.5 ft (0.457 m) by the Stern
Displacement 27,380 Tons (27,830 tonnes)

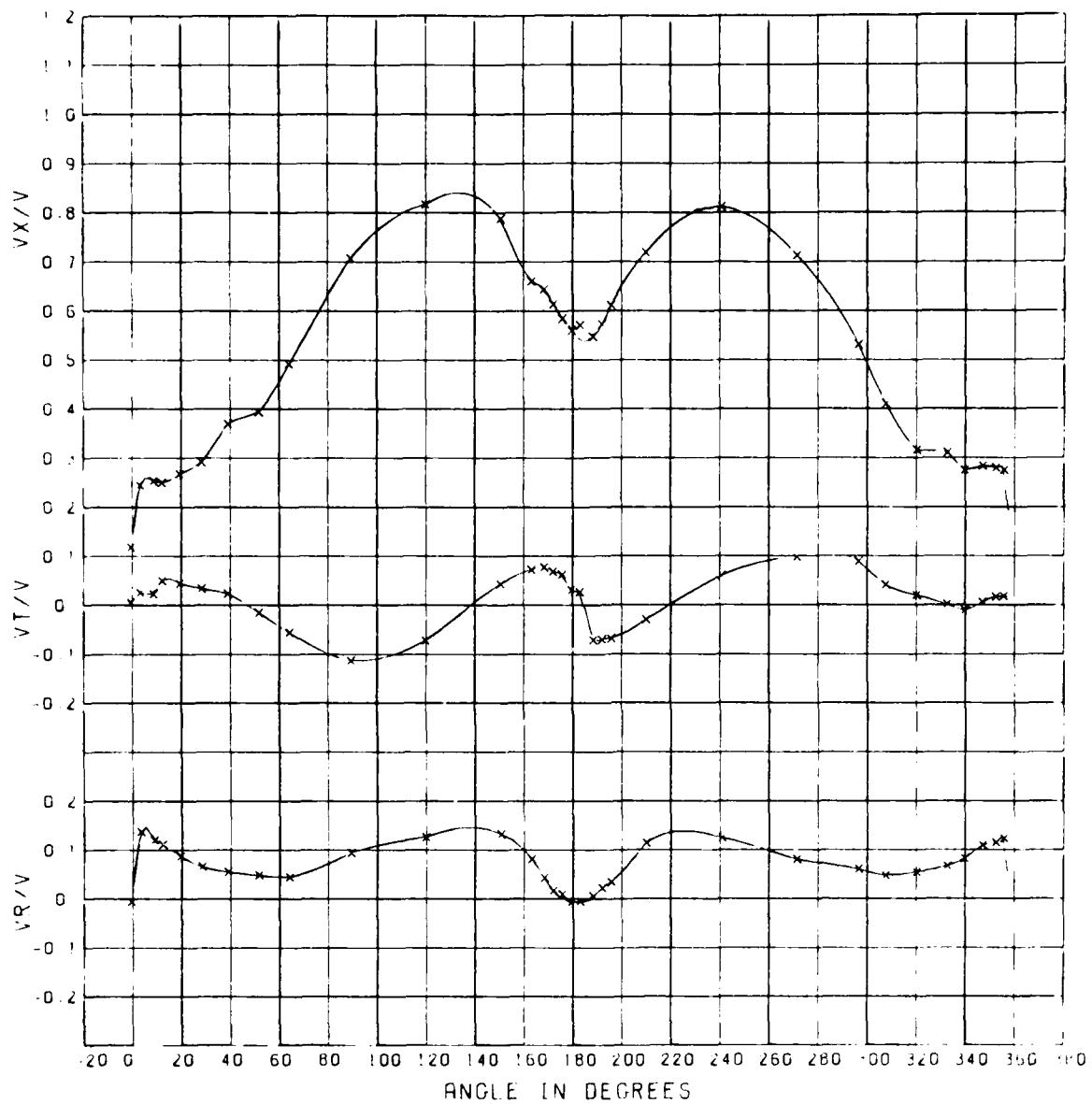


Figure A1 - CIRCUMFERENTIAL DISTRIBUTION OF THE LONGITUDINAL, TANGENTIAL AND RADIAL VELOCITY COMPONENT RATIOS - RADIUS RATIO = 0.35
EXPERIMENT 1

TRIMMED 1.5 FEET BY THE STERN, DISPLACEMENT 27,380 TONS

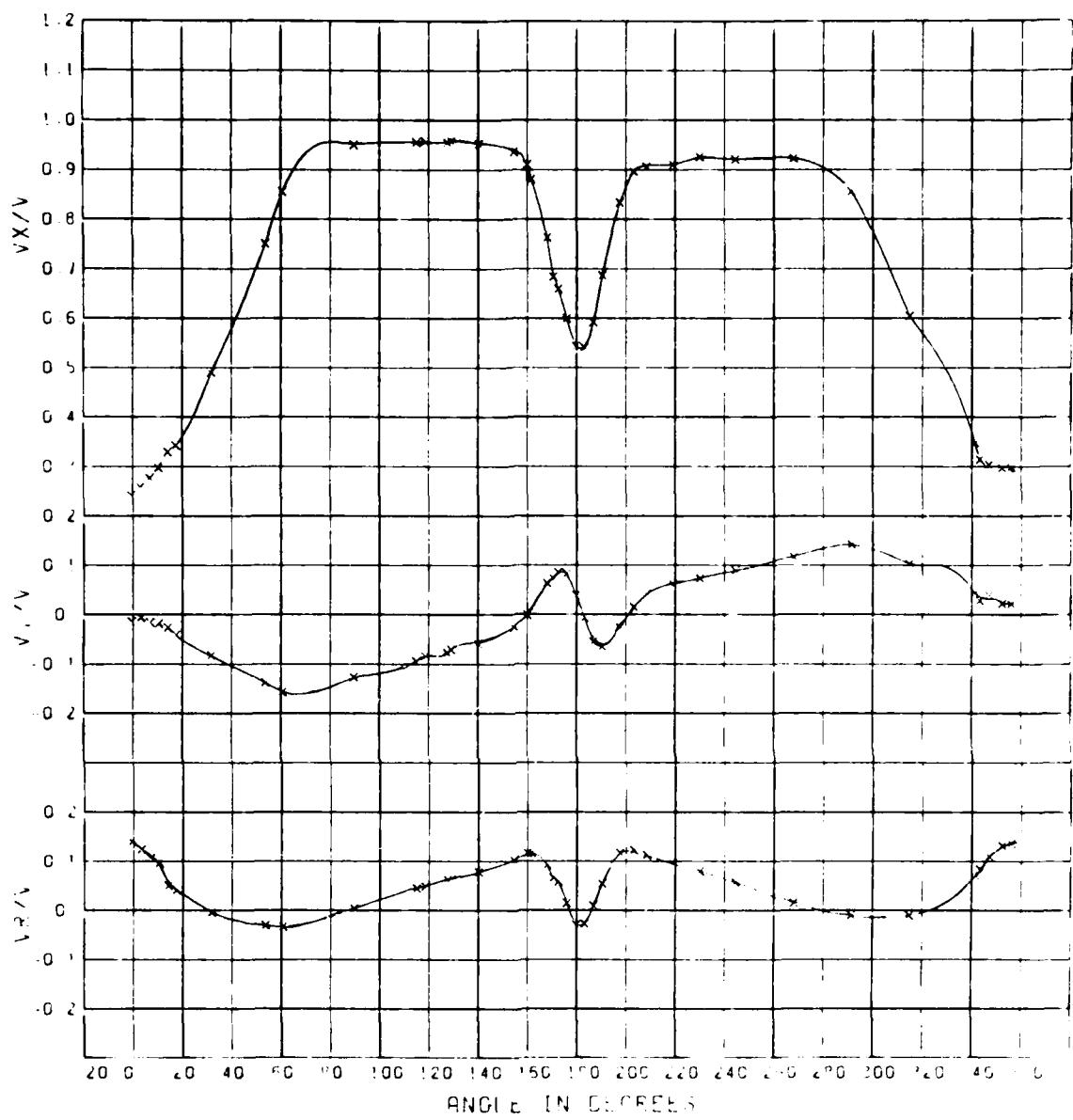


Figure A2 - CIRCUMFERENTIAL DISTRIBUTION OF THE LONGITUDINAL, TANGENTIAL AND RADIAL VELOCITY COMPONENT RATIOS - RADIUS RATIO = 0.556
EXPERIMENT 1

TRIMMED 1.5 FEET BY THE STERN, DISPLACEMENT 27,380 TONS

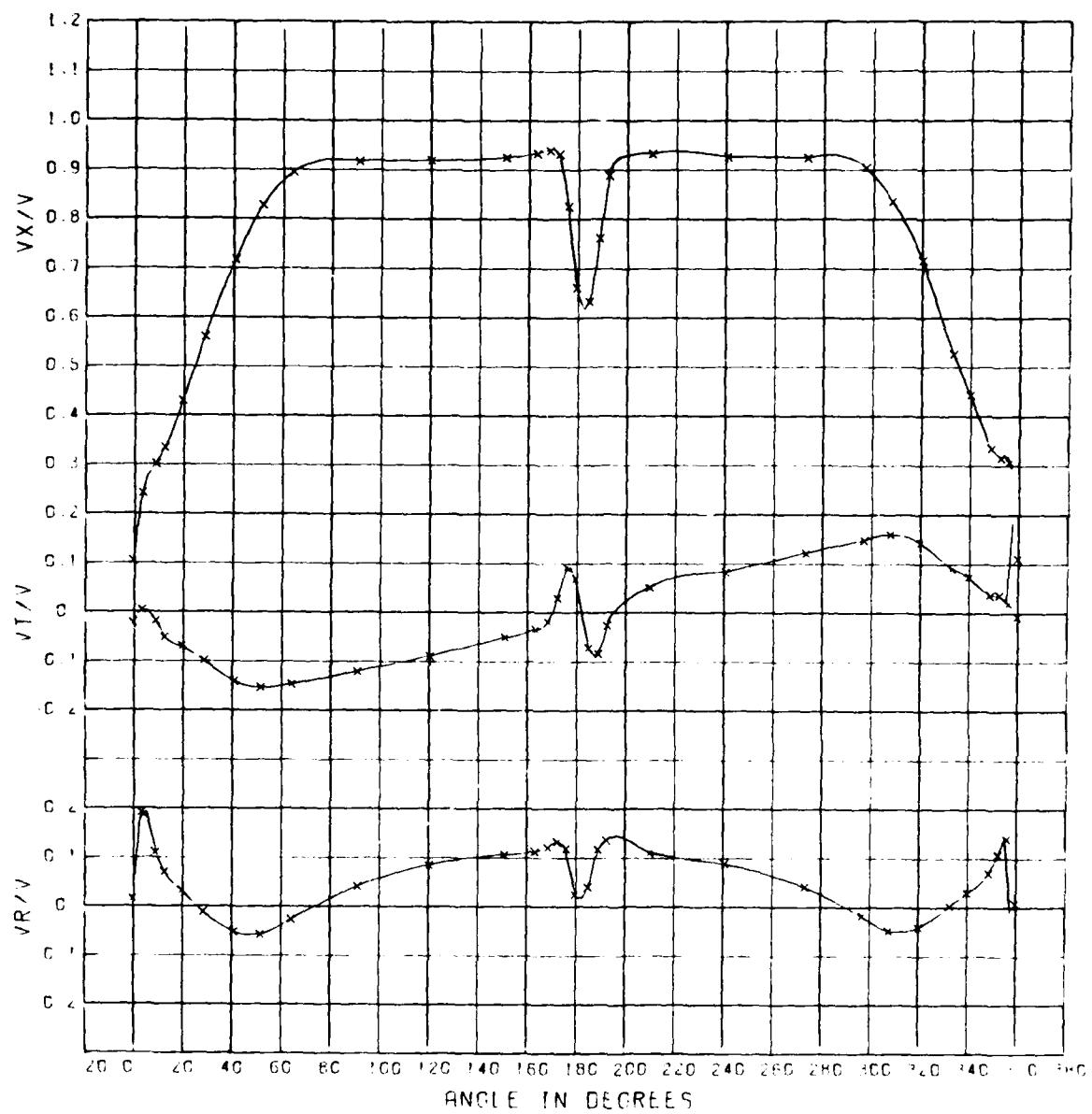


Figure A3 - CIRCUMFERENTIAL DISTRIBUTION OF THE LONGITUDINAL, TANGENTIAL AND RADIAL VELOCITY COMPONENT RATIOS - RADIUS RATIO = 0.774
EXPERIMENT 1

TRIMMED 1.5 FEET BY THE STERN, DISPLACEMENT 27,380 TONS

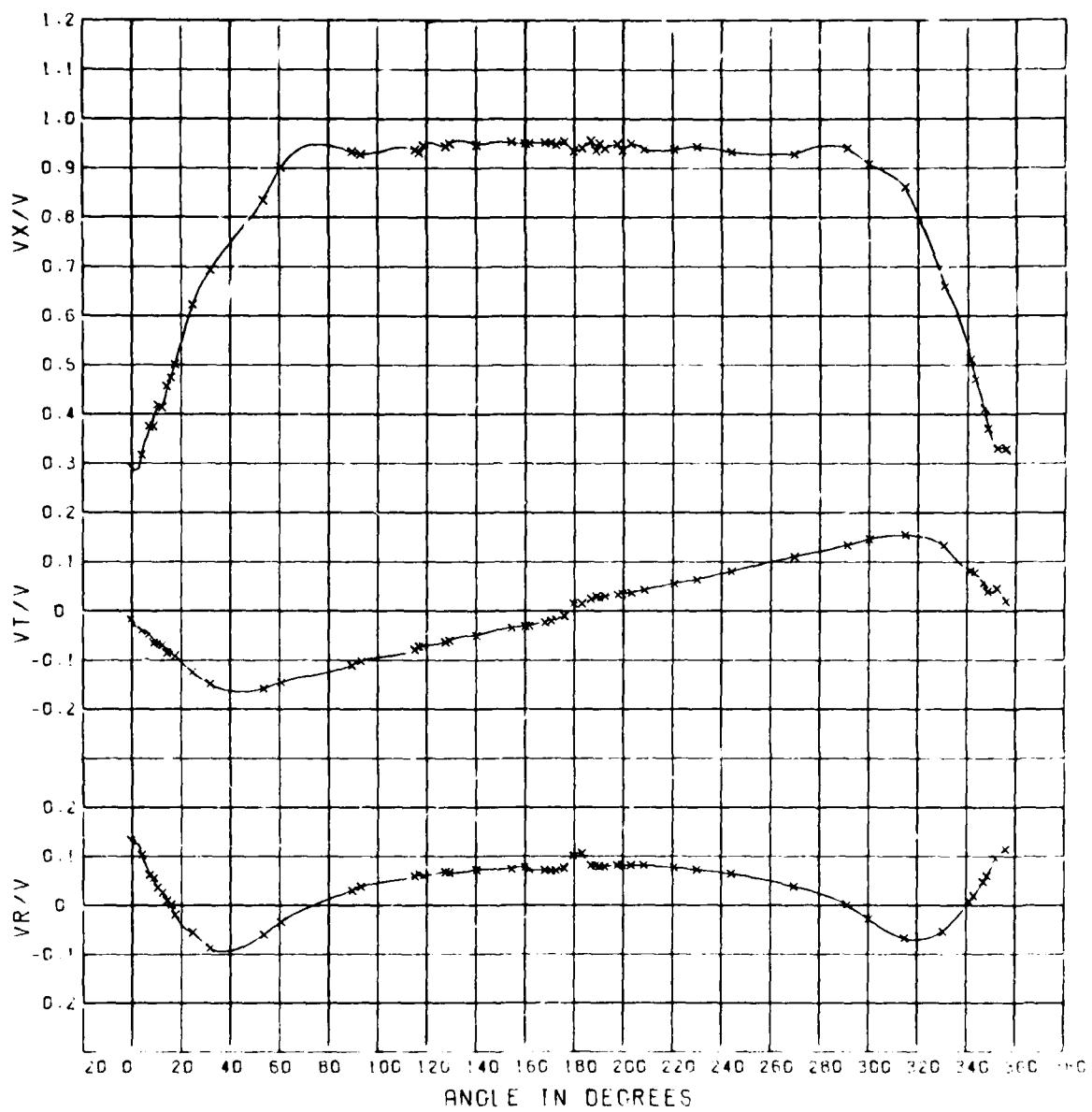


Figure A4 - CIRCUMFERENTIAL DISTRIBUTION OF THE LONGITUDINAL, TANGENTIAL AND RADIAL VELOCITY COMPONENT RATIOS - RADIUS RATIO = 1.017
EXPERIMENT 1

TRIMMED 1.5 FEET BY THE STERN, DISPLACEMENT 27,380 TONS

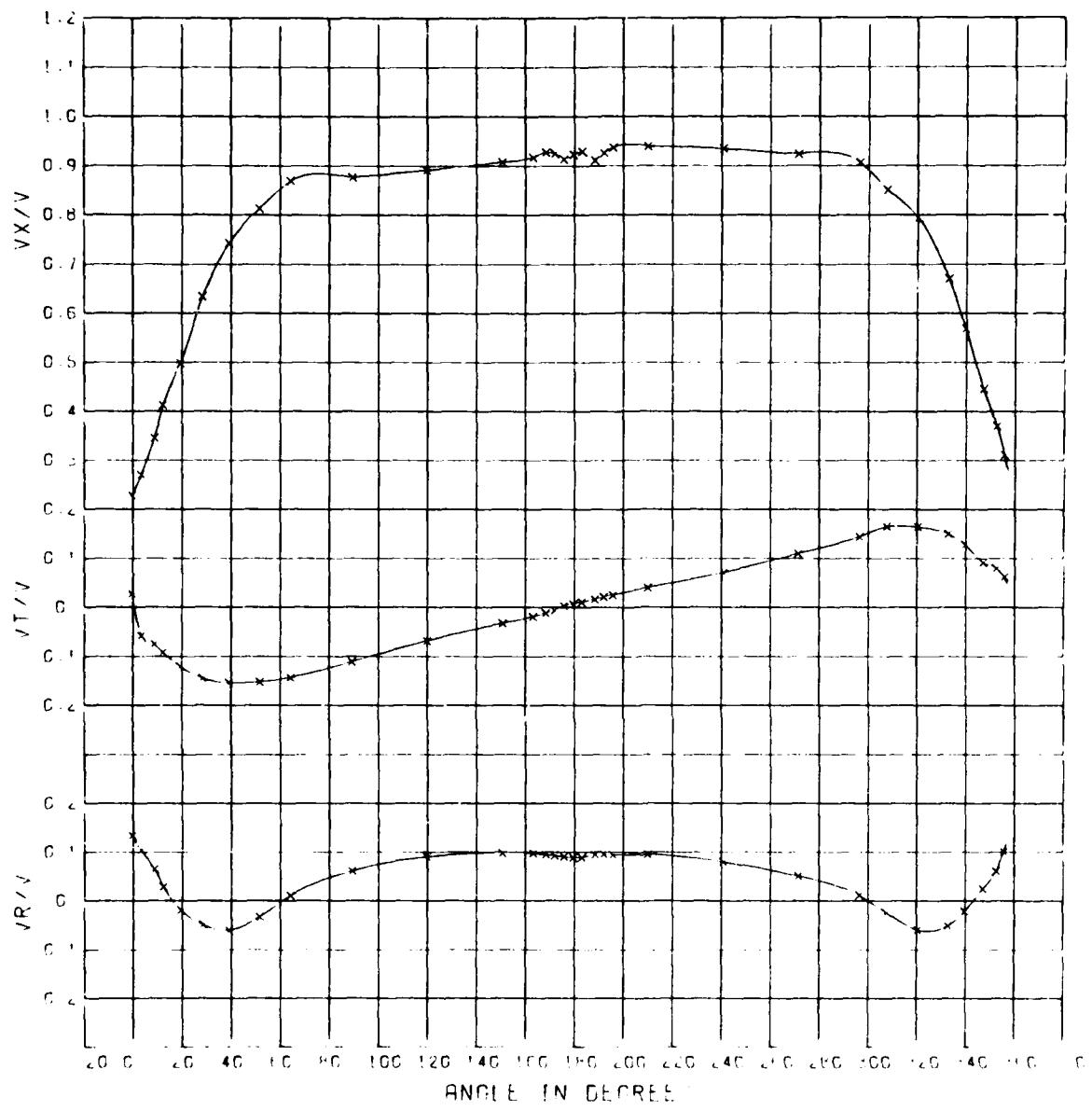


Figure A5 - CIRCUMFERENTIAL DISTRIBUTION OF THE LONGITUDINAL, TANGENTIAL AND RADIAL VELOCITY COMPONENT RATIOS - RADIUS RATIO = 1.178
EXPERIMENT 1

TRIMMED 1.5 FEET BY THE STERN, DISPLACEMENT 27,380 TONS

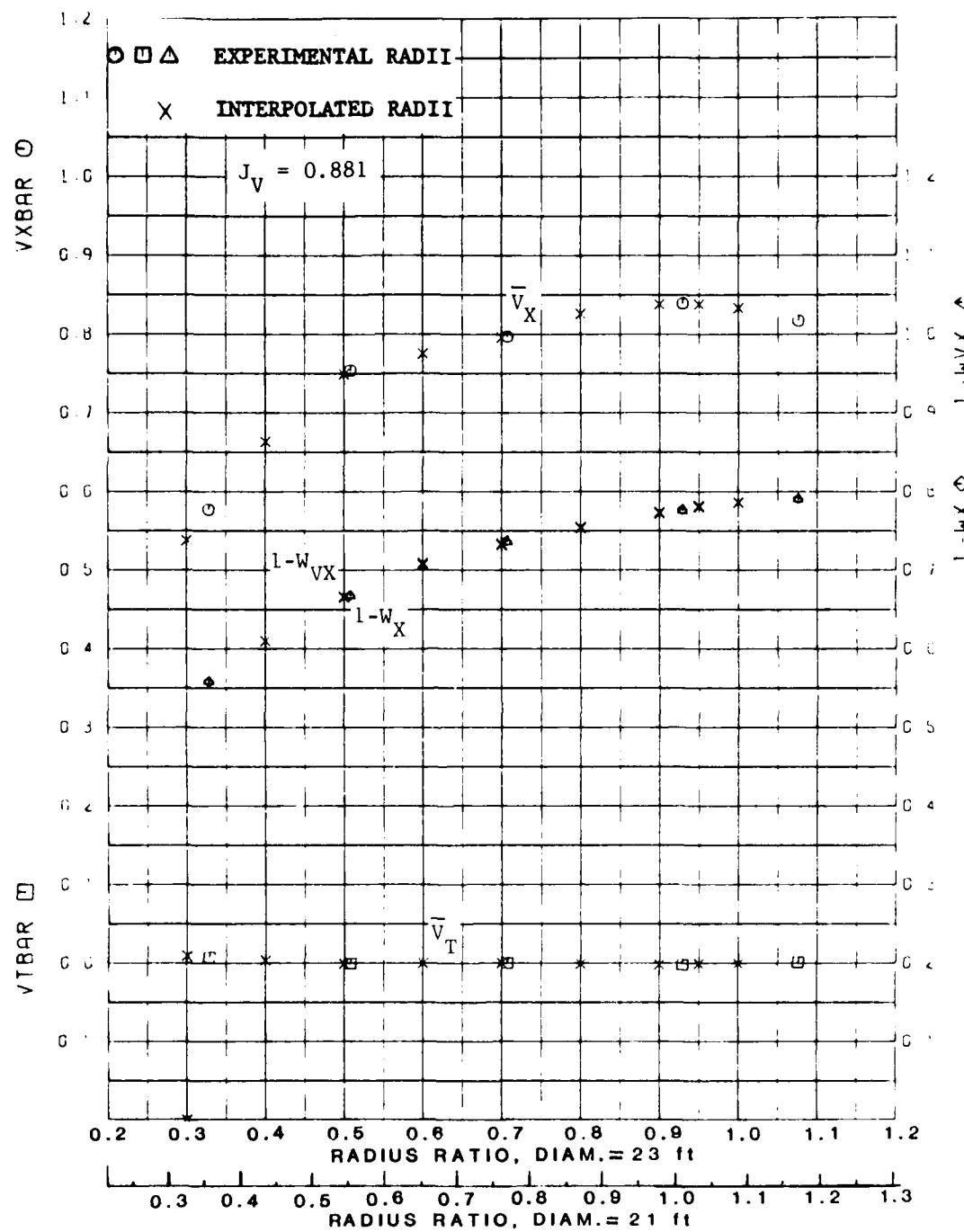


Figure A6 - RADIAL DISTRIBUTION OF THE MEAN VELOCITY COMPONENT RATIOS
EXPERIMENT 1

TRIMMED 1.5 FEET BY THE STERN, DISPLACEMENT 27,380 TONS

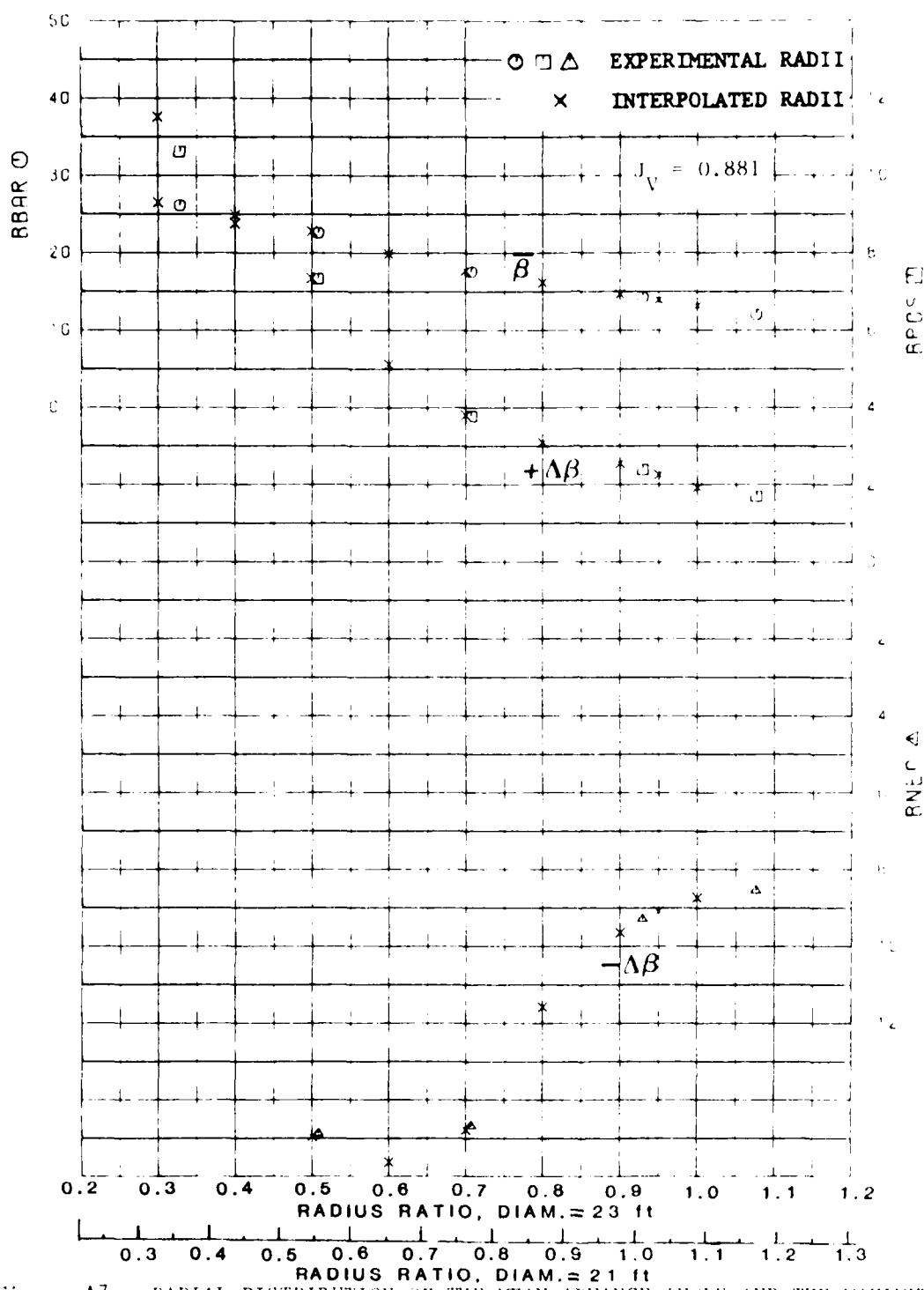


Figure A7 - RADIAL DISTRIBUTION OF THE MEAN ADVANCE ANGLE AND THE MAXIMUM VARIATIONS OF THE ADVANCE ANGLE FOR MODEL 5326
EXPERIMENT I

TRIMMED 1.5 FEET BY THE STERN, DISPLACEMENT 27,380 TONS

Table A1 - LISTING OF THE MEAN VELOCITY COMPONENT RATIOS, THE MEAN ADVANCE ANGLES AND OTHER DERIVED QUANTITIES
EXPERIMENT 1

TRIMMED 1.5 FEET BY THE STERN, DISPLACEMENT 27,380 TONS
D = 21 ft (6.4m), $J_V = 1.01$

RADIUS = .359	.556	.774	1.017	1.178	.730	.300	.407	.520	.609	.760	.800	.900	1.000
$V_XBAR = .577$.754	.797	.639	.617	.359	.499	.625	.717	.763	.806	.833	.839	
$V_TBAR = .007$	-.001	.000	-.002	.001	.020	.011	.005	.001	-.003	-.006	-.006	-.002	
$V_RBAR = .034$.039	.069	.027	.046	.153	.101	.071	.047	.043	.044	.033	.027	
$1-WV = .492$.613	.701	.753	.773	0.010	.435	.509	.582	.616	.676	.706	.731	.751
$1-WT = .493$.614	.704	.755	.775	0.000	.435	.509	.581	.619	.678	.708	.732	.753
$WBAR = .27.1^{\circ}$	23.56	18.32	16.86	12.56	27.87	27.84	26.57	24.71	22.24	19.74	17.95	16.52	15.10
$BPOS = 10.33$	7.46	3.69	2.50	1.76	20.42	12.45	9.89	7.94	6.57	4.65	3.63	3.16	2.60
$\Theta_{BETA} = 115.0^{\circ}$	77.50	205.00	75.01	202.50	172.50	117.50	110.50	107.50	97.50	90.05	77.50	75.00	
$BNEG = -19.9^{\circ}$	-15.48	-15.31	-9.71	-8.93	-36.70	-23.31	-16.13	-16.00	-16.07	-16.12	-16.14	-11.53	-9.84
$\Theta_{BETA} = 7.7^{\circ}$	5.57	5.57	5.00	5.00	67.50	0.00	0.00	0.00	3.00	3.00	2.00	0.00	0.00

V_XBAR IS CIRCUMFERENTIAL MEAN LONGITUDINAL VELOCITY.

V_TBAR IS CIRCUMFERENTIAL MEAN TANGENTIAL VELOCITY.

V_RBAR IS CIRCUMFERENTIAL MEAN RADIAL VELOCITY.

$1-WV$ IS VOLUNTARY MEAN WAVE VELOCITY WITHOUT TANGENTIAL CORRECTION.

$1-WT$ IS VOLUNTARY MEAN WAVE VELOCITY WITH TANGENTIAL CORRECTION.

$BBAR$ IS MEAN ANGLE OF ADVANCE.

$BPOS$ IS VARIATION BETWEEN THE MAXIMUM AND MEAN ADVANCE ANGLES ($\Delta\Theta_{BETA}$ PLUS).

$BNEG$ IS VARIATION BETWEEN THE MINIMUM AND MEAN ADVANCE ANGLES ($\Delta\Theta_{BETA}$ MINUS).

Θ_{BETA} IS ANGLE IN DEGREES AT WHICH CORRESPONDING $BPOS$ OR $BNEG$ OCCURS.

Table A2 - HARMONIC ANALYSIS OF LONGITUDINAL VELOCITY COMPONENT RATIOS
 AT THE EXPERIMENTAL RADII
 EXPERIMENT 1

TRIMMED 1.5 FEET BY THE STERN, DISPLACEMENT 27,380 TONS

WAVE NUMBER	1	2	3	4	5	6	7	8
WAVELENGTH = 359	-0.2327	-0.1430	0.065	-0.0152	0.0171	-0.0152	-0.0113	-0.0044
WAVELENGTH = 556	-0.2167	-0.1223	-0.0187	-0.0565	0.0445	-0.0243	0.0294	-0.0272
WAVELENGTH = 774	-0.2732	-0.1726	-0.0702	-0.1630	0.0555	-0.0256	0.0169	-0.0235
WAVELENGTH = 1092	-0.1234	-0.1786	-0.0315	-0.0455	-0.0246	-0.0140	-0.0155	-0.0105
WAVELENGTH = 1410	-0.1240	-0.1272	-0.0277	-0.0527	-0.0292	-0.0225	-0.0154	-0.0124

Table A3 - HARMONIC ANALYSIS OF LONGITUDINAL VELOCITY COMPONENT RATIOS
 AT THE INTERPOLATED RADII
 EXPERIMENT 1

TRIMMED 1.5 FEET BY THE STERN, DISPLACEMENT 27,380 TONS

WAVELINE	=	1	2	3	4	5	6	7	8
RADIUS = .200		-.2792	.0724	.1416	.0280	-.0484	.0128	-.0335	.0321
RADIUS = .375		-.2402	-.0993	.0889	-.0011	-.0227	-.0062	-.0119	.0073
RADIUS = .400		-.2324	-.1696	.0421	-.0248	.0277	-.0205	.0042	-.0111
RADIUS = .500		-.2294	-.2116	.0014	-.0424	.0427	-.0287	.0148	-.0232
RADIUS = .600		-.2474	-.2122	-.0346	-.0552	.0355	-.0305	.0199	-.0270
RADIUS = .700		-.2071	-.1909	-.0637	-.0622	.0172	-.0282	.0195	-.0256
RADIUS = .875		-.2015	-.1713	-.0813	-.0613	.0316	-.0237	.0198	-.0212
RADIUS = .938		-.1957	-.1577	-.0362	-.0536	-.0112	-.0174	.0052	-.0144
RADIUS = 1.000		-.1903	-.1454	-.0913	-.0498	-.0204	-.0149	-.0146	-.0108

Table A4 - HARMONIC ANALYSIS OF TANGENTIAL VELOCITY COMPONENT RATIOS
 AT THE EXPERIMENTAL RADII
 EXPERIMENT 1

TRIMMED 1.5 FLEET BY THE STERN, DISPLACEMENT 27,380 TONS

HARMONIC	1	2	3	4	5	6	7	8
RADIUS = 359 AMPLITUDE =	-0.3636	-0.3556	0.6560	-0.7108	0.0144	-0.0397	0.0059	-0.0036
RADIUS = 556 AMPLITUDE =	-0.1767	-0.3324	0.0534	-0.0338	0.127	-0.0129	0.103	-0.0291
RADIUS = 774 AMPLITUDE =	-0.1255	-0.3366	-0.2423	-0.0056	0.035	-0.0223	0.079	-0.0079
RADIUS = 1,017 AMPLITUDE =	-0.1704	-0.0484	-0.0235	-0.0048	-0.0024	0.0536	0.013	0.018
RADIUS = 1,472 AMPLITUDE =	-0.1323	-0.0569	-0.2276	-0.1118	-0.0974	-0.0036	-0.0033	-0.0029

Table A5 - HARMONIC ANALYSIS OF TANGENTIAL VELOCITY COMPONENT RATIOS
AT THE INTERPOLATED RADII
EXPERIMENT 1

TRIMMED 1.5 FEET BY THE STERN, DISPLACEMENT 27,380 TONS

POINT	1	2	3	4	5	6	7	8
POINT 2	.209							
POINT 3		.0742	.0324	.1147	-.0191	.0140	.0012	-.0028
POINT 4								.0056
POINT 5								
POINT 6								
POINT 7								
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Table A6 - INPUT DATA FOR WAKE SURVEY ANALYSES
 EXPERIMENT 1
 TRIMMED 1.5 FEET BY THE STERN, DISPLACEMENT 27,380 TONS

RADIUS RATIO = 0.359				RADIUS RATIO = 0.556				RADIUS RATIO = 0.774			
ANGLE	Vx/V	Vy/V	Vz/V	ANGLE	Vx/V	Vy/V	Vz/V	ANGLE	Vx/V	Vy/V	Vz/V
-5.5	.118	.005	-.005	-6	.241	.043	1.6	-6	.023	.103	.015
3.3	.244	.025	.137	1.2	.258	.127	1.1	.242	.004	.191	.191
4.7	.264	.022	.120	6.9	.276	.145	1.0	.249	.019	.110	.110
12.2	.010	.000	.111	10.4	.277	.112	0.2	.334	.022	.073	.073
19.5	.267	.004	.087	14.1	.329	.050	1.5	.424	.070	.031	.031
24.3	.241	.033	.063	17.4	.342	.042	0.4	.560	.049	.012	.012
33.0	.160	.053	.053	31.7	.490	.053	0.5	.705	.154	.051	.051
51.6	.396	.015	.015	51.5	.750	.131	0.7	.728	.139	.053	.053
68.1	.422	.017	.017	60.6	.856	.157	0.3	.622	.161	.054	.054
85.1	.707	.014	.014	49.4	.943	.064	1.6	.812	.145	.054	.054
119.8	.816	.013	.013	40.6	.946	.111	0.2	.900	.183	.025	.025
150.5	.778	.019	.130	116.0	.756	.042	0.4	.648	.152	.024	.024
160.4	.749	.016	.045	114.5	.958	.045	0.4	.917	.122	.040	.040
163.2	.666	.012	.012	104.1	.955	.062	0.2	.912	.091	.024	.024
169.3	.644	.017	.017	120.6	.955	.079	0.2	120.0	.079	.024	.024
172.0	.611	.015	.015	129.1	.958	.071	0.2	120.2	.092	.046	.046
175.7	.544	.002	.002	140.1	.954	.057	0.1	150.4	.023	.107	.107
179.5	.540	.030	.007	144.2	.952	.061	0.1	150.8	.027	.050	.050
183.0	.770	.026	.005	156.0	.756	.042	0.4	.910	.131	.036	.036
188.4	.570	.017	.017	154.5	.937	.025	0.2	163.2	.015	.113	.113
192.0	.572	.022	.022	150.9	.910	.062	0.4	163.2	.015	.120	.120
195.7	.611	.032	.032	161.4	.981	.071	0.2	175.2	.032	.132	.132
210.0	.719	.010	.010	167.2	.746	.141	1.0	175.4	.025	.042	.042
240.7	.811	.018	.118	168.6	.743	.065	0.6	178.5	.011	.115	.115
271.4	.713	.012	.012	170.5	.644	.073	0.2	186.7	.011	.024	.024
286.6	.641	.014	.014	172.5	.659	.065	0.5	.761	.084	.117	.117
317.4	.410	.004	.004	175.9	.599	.012	0.1	.847	.026	.135	.135
320.2	.316	.000	.000	176.7	.544	.039	0.1	209.9	.033	.051	.051
349.4	.244	.001	.001	186.8	.540	.004	0.2	240.5	.027	.042	.042
352.5	.271	.005	.005	190.3	.647	.056	0.0	273.2	.024	.084	.084
355.6	.275	.018	.018	192.6	.644	.073	0.2	296.9	.011	.121	.121
359.5	.114	.005	.005	194.3	.644	.060	0.2	.611	.011	.073	.073
367.1	.243	.005	.005	197.5	.659	.054	0.5	.761	.084	.117	.117
372.5	.271	.015	.015	201.1	.906	.123	1.2	.847	.026	.135	.135
375.6	.275	.018	.018	204.4	.906	.011	1.2	.526	.091	.002	.002
379.5	.114	.005	.005	219.2	.906	.000	0.0	339.9	.043	.074	.074
380.0	.925	.005	.005	230.0	.925	.013	0.8	349.6	.036	.036	.036
384.4	.244	.002	.002	240.0	.920	.013	0.5	352.3	.037	.107	.107
387.1	.243	.005	.005	247.9	.922	.115	0.1	.307	.019	.139	.139
392.5	.271	.015	.015	251.6	.925	.025	1.2	.356.0	.160	.019	.019
395.6	.275	.018	.018	252.1	.927	.042	1.0	.359.5	.118	.008	.008
399.5	.114	.005	.005	316.7	.904	.132	1.0	.160.6	.160	.024	.024
360.0	.925	.005	.005	357.2	.927	.022	1.3	.359.5	.118	.008	.008
365.6	.275	.018	.018	359.9	.928	.021	1.3	.359.5	.118	.024	.024
370.5	.114	.005	.005	360.4	.924	.013	1.4	.316	.037	.107	.107

APPENDIX B
RESULTS OF EXPERIMENT 2

Corresponding to
Trim 1.5 ft (0.457 m) by the Stern
Displacement 26,380 Tons (26,810 tonnes)

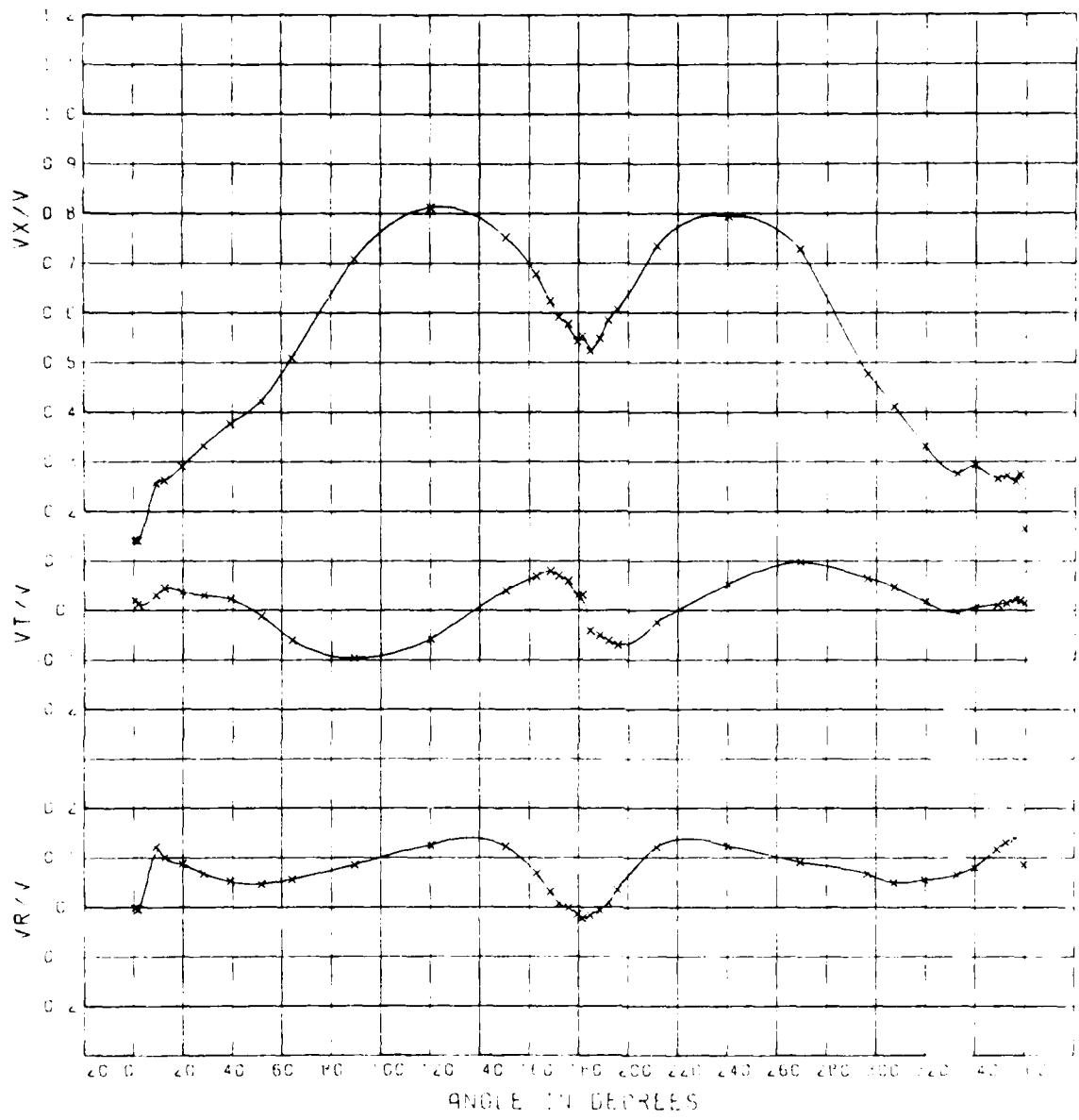


Figure B1 - CIRCUMFERENTIAL DISTRIBUTION OF THE LONGITUDINAL, TANGENTIAL AND RADIAL VELOCITY COMPONENT RATIOS - RADIUS RATIO = 0.359
EXPERIMENT 2

TRIMMED 1.5 FEET BY THE STERN, DISPLACEMENT 26,390 TONS

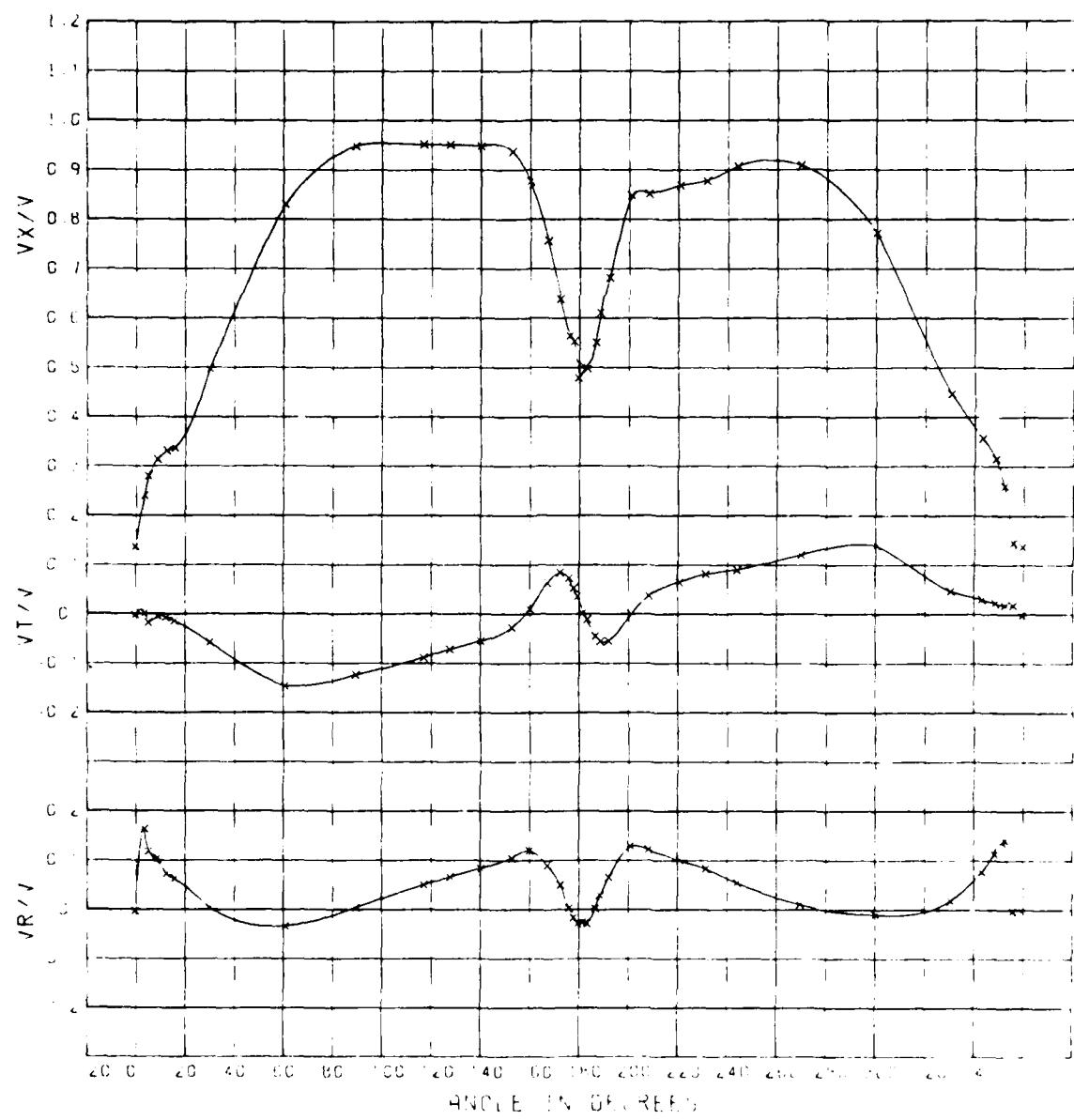


Figure B2 - CIRCUMFERENTIAL DISTRIBUTION OF THE LONGITUDINAL, TANGENTIAL AND RADIAL VELOCITY COMPONENT RATIOS - RADIUS RATIO = 0.556
EXPERIMENT 2

TRIMMED 1.5 FEET BY THE STERN, DISPLACEMENT 26,390 TONS

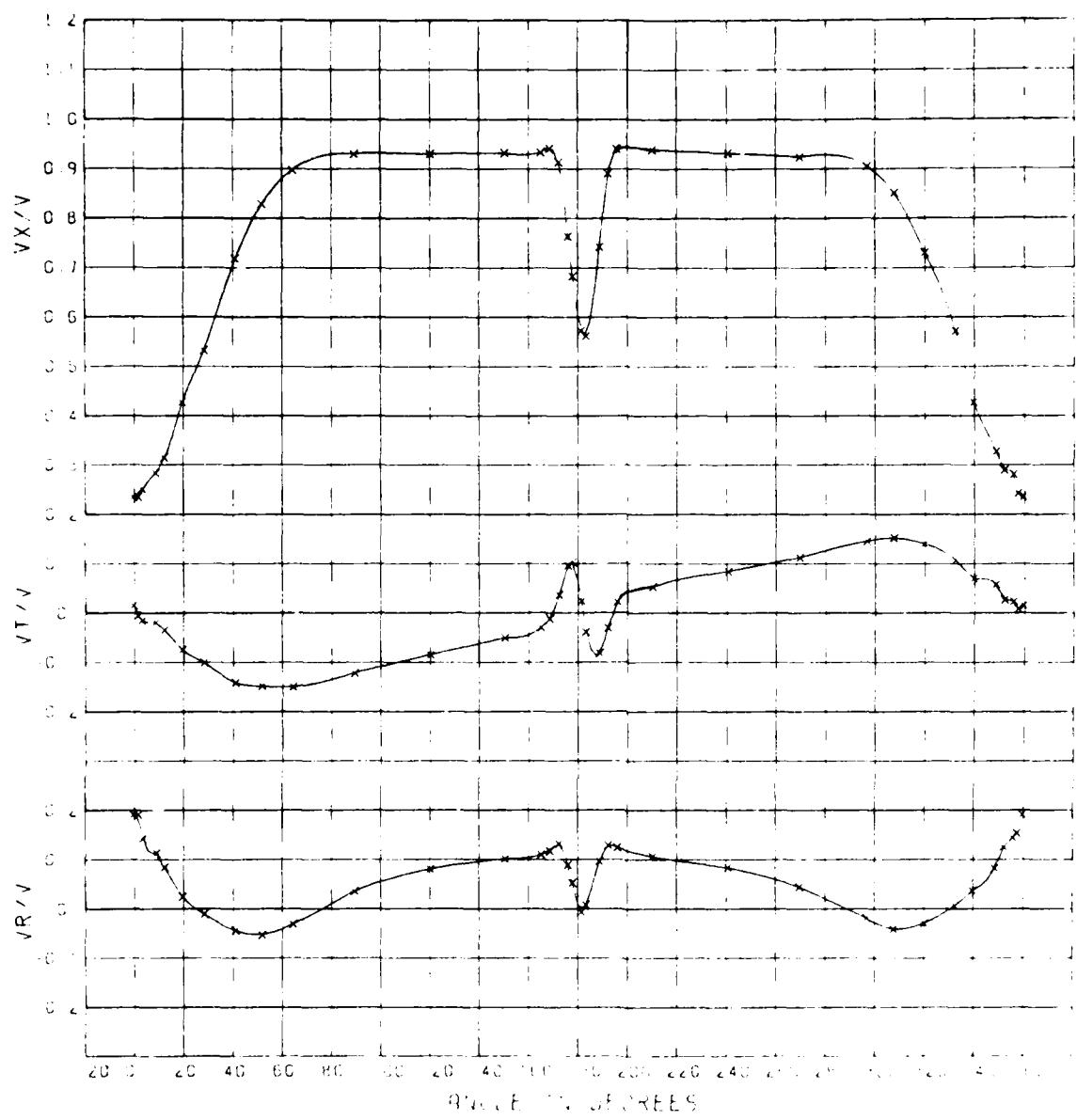


Figure B3 - CIRCUMFERENTIAL DISTRIBUTION OF THE LONGITUDINAL, TANGENTIAL AND RADIAL VELOCITY COMPONENT RATIOS - RADIUS RATIO = 0.774
EXPERIMENT 2

TRIMMED 1.5 FEET BY THE STERN, DISPLACEMENT 26,300 TONS

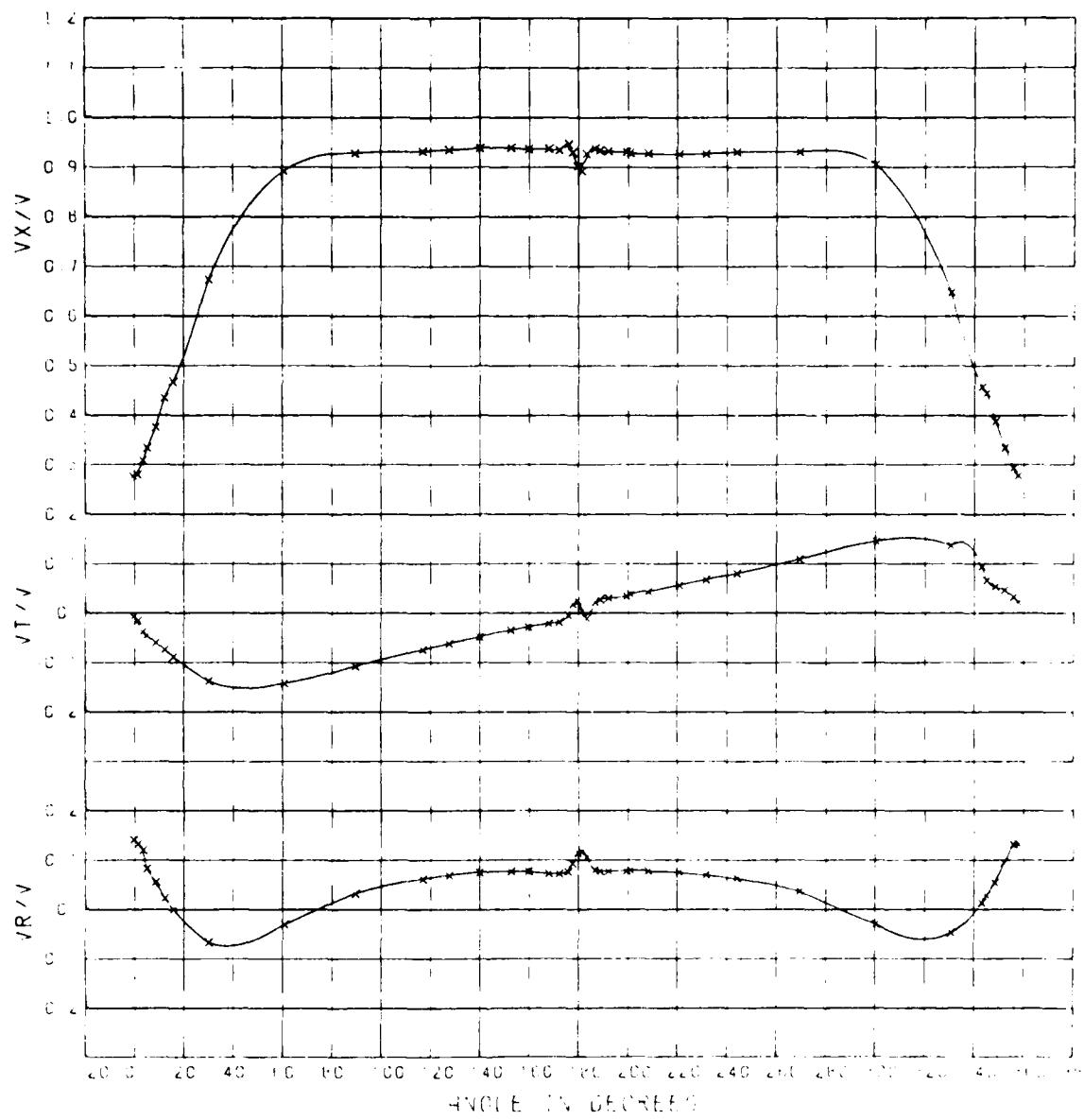


Figure B4 - CIRCUMFERENTIAL DISTRIBUTION OF THE LONGITUDINAL, TANGENTIAL AND RADIAL VELOCITY COMPONENT RATIOS - RADIUS RATIO = 1.017
EXPERIMENT 2

TRIMMED 1.5 FEET BY THE STERN, DISPLACEMENT 26,300 TONS

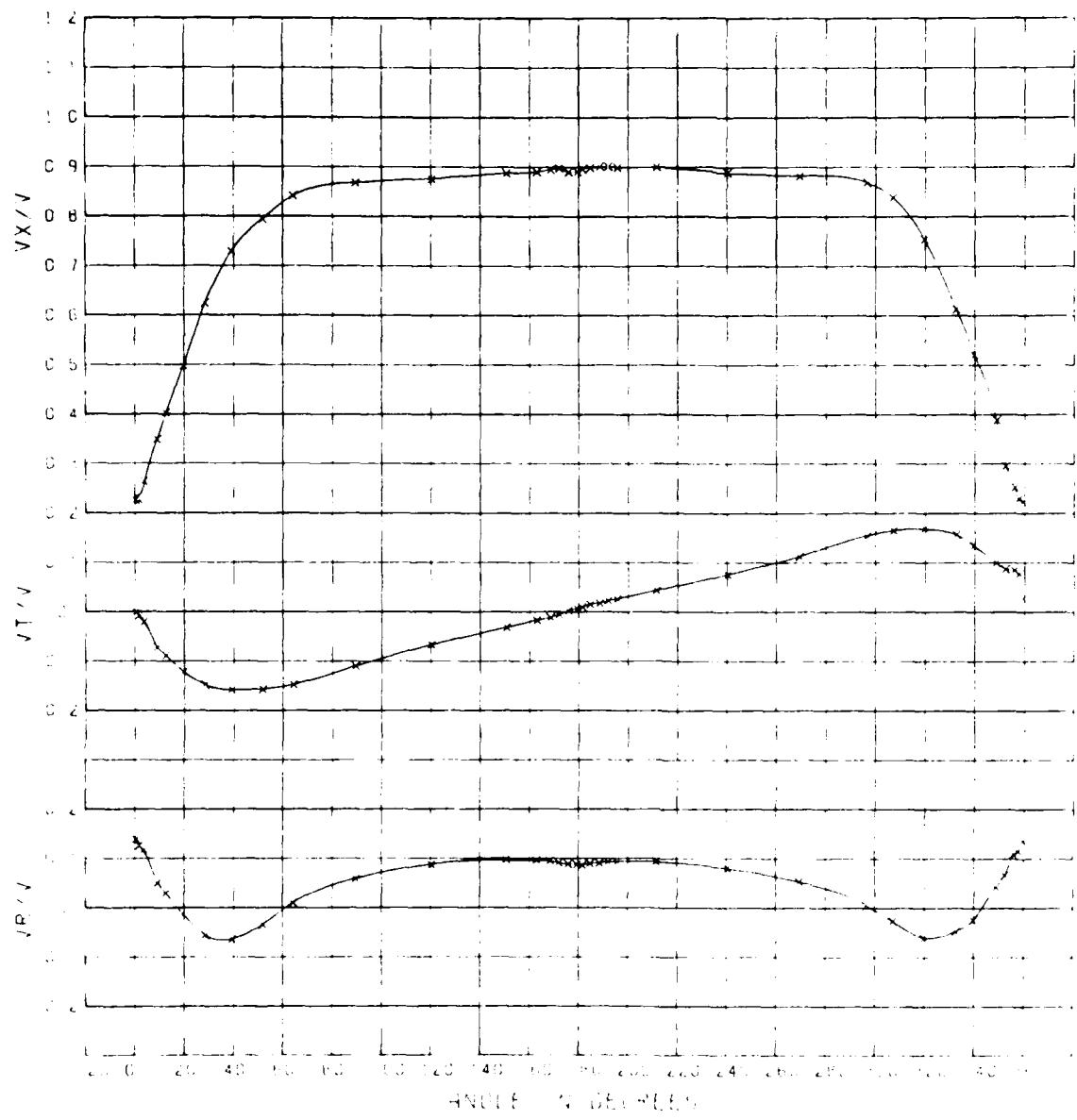


Figure B5 - CIRCUMFERENTIAL DISTRIBUTION OF THE LONGITUDINAL, TANGENTIAL AND RADIAL VELOCITY COMPONENT RATIOS - RADIUS RATIO = 1.178
EXPERIMENT 2

TRIMMED 1.5 FEET BY THE STERN, DISPLACEMENT 26,300 TONS

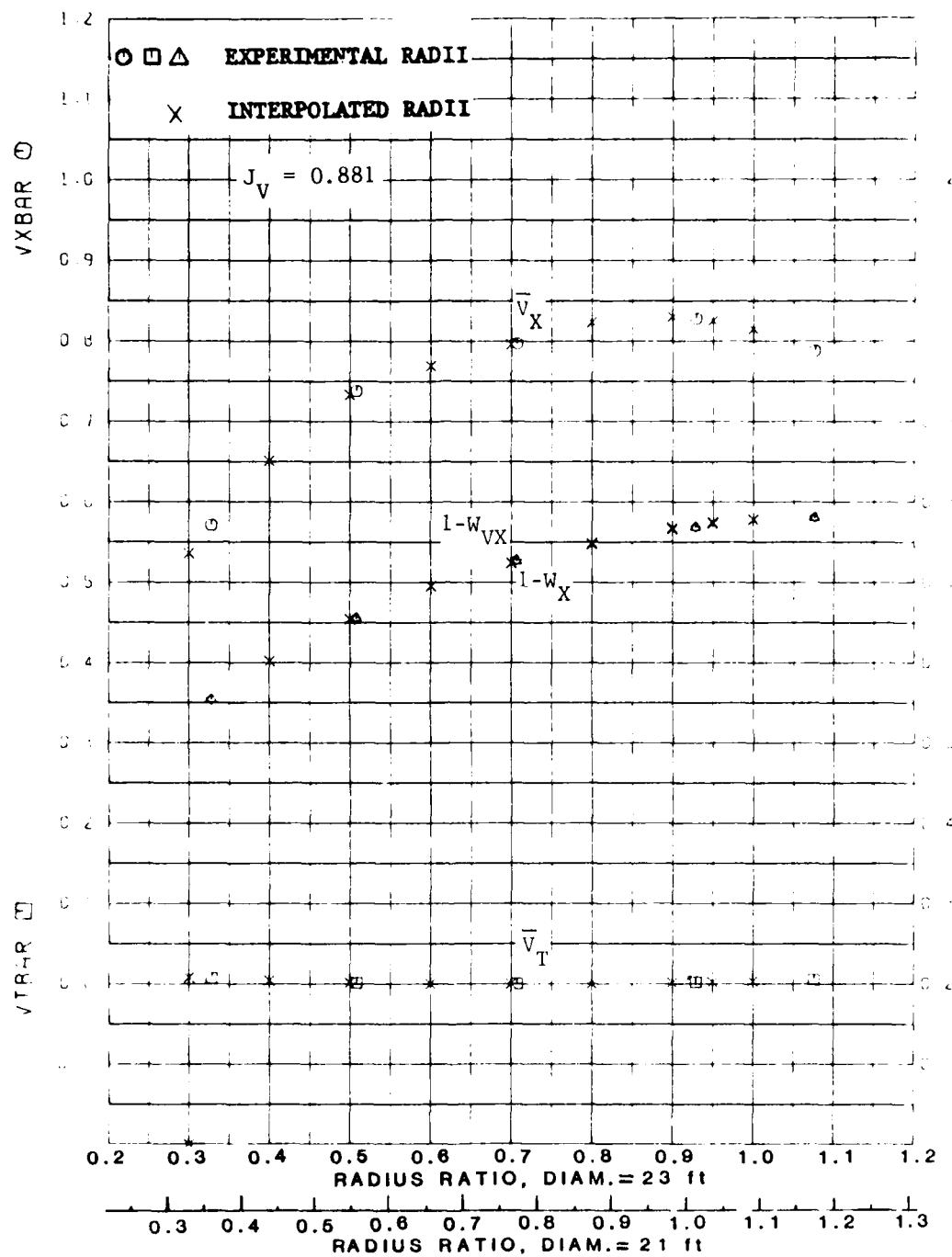


Figure B6 - RADIAL DISTRIBUTION OF THE MEAN VELOCITY COMPONENT RATIOS
EXPERIMENT 2

TRIMMED 1.5 FEET BY THE STERN, DISPLACEMENT 26,390 TONS

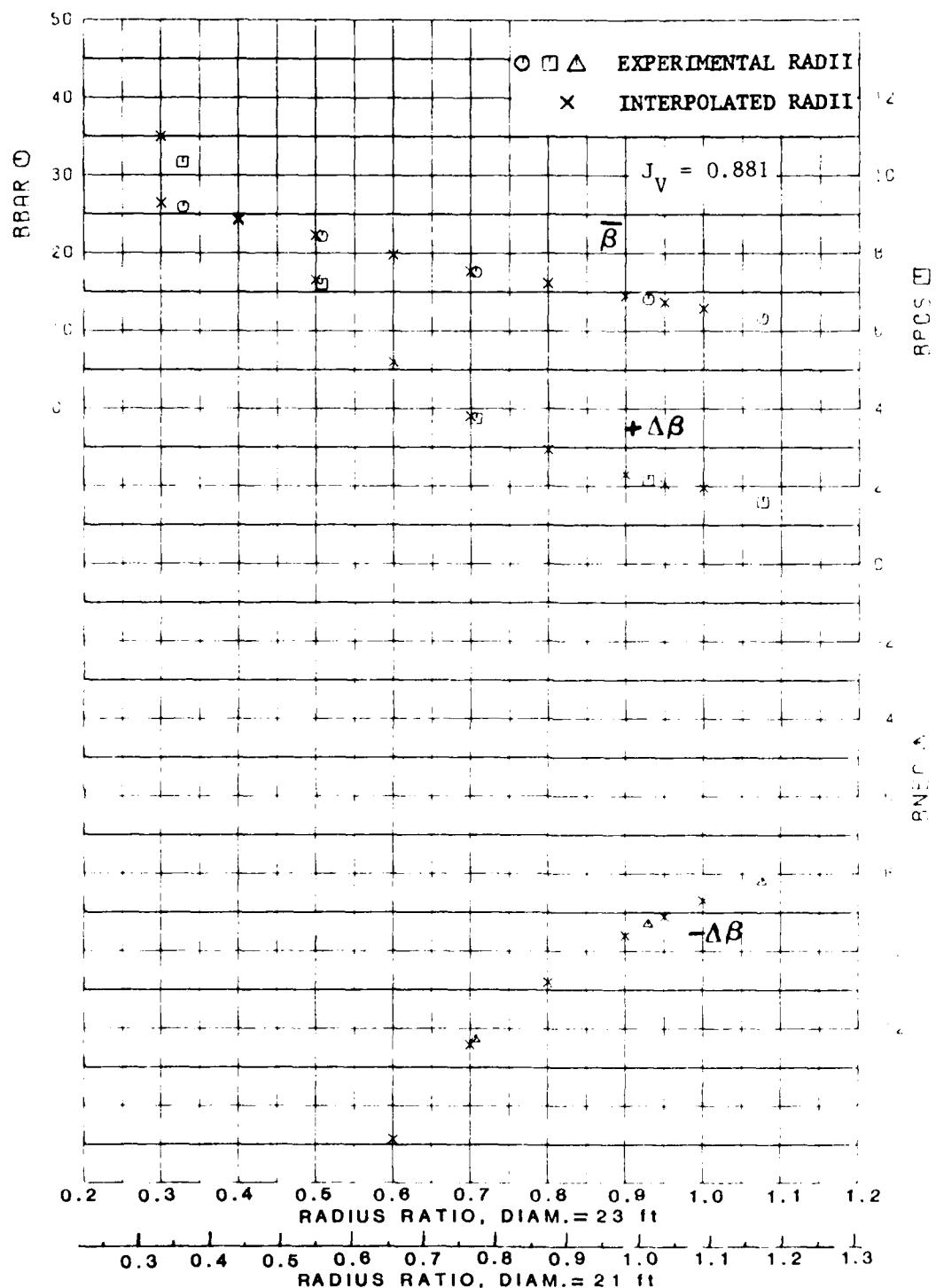


Figure B7 - RADIAL DISTRIBUTION OF THE MEAN ADVANCE ANGLE AND THE MAXIMUM VARIATIONS OF THE ADVANCE ANGLE FOR MODEL 5326
EXPERIMENT 2

TRIMMED 1.5 FEET BY THE STERN, DISPLACEMENT 26,390 TONS

Table B1 - LISTING OF THE MEAN VELOCITY COMPONENT RATIOS, THE MEAN ADVANCE ANGLES AND OTHER DERIVED QUANTITIES
EXPERIMENT 2

TRIMMED 1.5 FEET BY THE STERN, DISPLACEMENT 26,390 TONS
D = 21 ft (6.4 m), $J_V = 1.01$

RADIUS =	359	556	.774	1.017	1.174	.200	.300	.460	.510	.600	.700	.800	.900	1.000
VBAR =	571	738	.797	.826	.768	.358	.500	.615	.702	.753	.781	.806	.827	.843
VBAR =	007	001	.000	.012	.005	.015	.010	.005	.002	.001	.000	.000	.001	
VBAR =	042	.036	.059	.029	.064	.155	.105	.069	.045	.047	.043	.032	.029	
1-WX =	.495	.606	.694	.747	.754	0.060	.444	.510	.575	.624	.669	.710	.726	.745
1-WX =	.496	.607	.695	.748	.756	0.030	.444	.510	.575	.631	.670	.702	.727	.744
BDR =	26.96	23.10	19.32	14.66	12.12	29.35	27.96	26.21	24.25	21.95	15.72	17.94	16.45	16.97
BDR =	10.63	7.49	3.91	2.27	1.66	22.59	12.00	9.61	6.20	6.50	4.74	3.68	2.92	2.35
1-FA =	115.06	95.06	80.06	62.50	176.00	222.50	117.50	112.50	102.50	92.50	85.00	80.00	60.00	42.50
BD =	-15.71	-19.30	-12.03	-9.72	-8.59	-30.45	-21.04	-19.86	-19.32	-17.45	-14.29	-12.39	-10.96	-9.88
1-FA =	0.00	357.50	0.00	0.30	357.50	55.00	2.50	3.00	357.50	357.50	0.00	0.00	0.00	0.00

1-WX IS THE MEAN ADVANCE ANGLE, MEAN LONGITUDINAL VELOCITY.

1-FA IS THE MEAN ADVANCE ANGLE, MEAN TANGENTIAL VELOCITY.

VBAR IS CIRCUMFERENTIAL MEAN PARTIAL VELOCITY.

1-WT IS THE MEAN MEAN WAVE VELOCITY WITHOUT TANGENTIAL CORRECTION.

1-WX IS THE MEAN MEAN WAVE VELOCITY WITH TANGENTIAL CORRECTION.

BBAR IS MEAN ANGLE OF ADVANCE.

BDR IS VARIATION BETWEEN THE MAXIMUM AND MEAN ADVANCE ANGLES (DELTA BETA PLUS).

BD = VARIATION BETWEEN THE MINIMUM AND MEAN ADVANCE ANGLES (DELTA BETA MINUS).

1-FA = ANGLE IN DEGREES AT WHICH CORRESPONDING BOS OR BNG OCCURS.

1-WX = ANGLE IN DEGREES AT WHICH CORRESPONDING BOS OR BNG OCCURS.

Table B2 - HARMONIC ANALYSIS OF LONGITUDINAL VELOCITY COMPONENT RATIOS
 AT THE EXPERIMENTAL RADII
 EXPERIMENT 2

TRIMMED 1.5 FEET BY THE STERN, DISPLACEMENT 26,300 TONS

HARMONIC	1	2	3	4	5	6	7	8
RADIUS = .359 AMPLITUDE =	-0.2326	-0.1425	0.0587	-0.0175	0.0094	-0.0198	0.0223	-0.0363
RADIUS = .556 AMPLITUDE =	-0.2031	-0.2272	-0.0183	-0.0536	0.0331	-0.0378	0.0215	-0.0290
RADIUS = .774 AMPLITUDE =	-0.1991	-0.1866	-0.0741	-0.0712	0.0267	-0.0294	0.0193	-0.0235
RADIUS = 1.017 AMPLITUDE =	-0.1888	-0.1429	-0.1908	-0.0491	-0.0233	-0.0133	-0.0074	-0.0063
RADIUS = 1.179 AMPLITUDE =	-0.1607	-0.1273	-0.0882	-0.0520	-0.0294	-0.0191	-0.0133	-0.0097

Table B3 - HARMONIC ANALYSIS OF LONGITUDINAL VELOCITY COMPONENT RATIOS
 AT THE INTERPOLATED RADII
 EXPERIMENT 2

TRIMMED 1.5 FEET BY THE STERN, DISPLACEMENT 26,390 TONS

HARMONIC	1	2	3	4	5	6	7	8
RADIUS = .200 AMPLITUDE =	-.2724	.0099	.1392	.0361	-.0390	.0126	-.0279	.0336
RADIUS = .300 AMPLITUDE =	-.2453	-.0947	.0866	.0002	-.0061	-.0096	-.0074	.0071
RADIUS = .400 AMPLITUDE =	-.2240	-.1696	.0406	-.0284	.0163	-.0255	.0079	-.0125
RADIUS = .500 AMPLITUDE =	-.2088	-.2148	.0010	-.0498	.0281	-.0352	.0181	-.0250
RADIUS = .600 AMPLITUDE =	-.2027	-.2189	-.0327	-.0636	.0257	-.0366	.0227	-.0286
RADIUS = .700 AMPLITUDE =	-.2016	-.2032	-.0595	-.0703	.0150	-.0329	.0224	-.0264
RADIUS = .800 AMPLITUDE =	-.1981	-.1808	-.0770	-.0673	.0023	-.0262	.0154	-.0204
RADIUS = .900 AMPLITUDE =	-.1941	-.1609	-.0858	-.0557	.0120	-.0173	.0028	-.0112
RADIUS = 1.000 AMPLITUDE =	-.1896	-.1451	-.0904	-.0496	.0220	-.0134	-.0062	-.0066

Table B4 - HARMONIC ANALYSIS OF TANGENTIAL VELOCITY COMPONENT RATIOS
 AT THE EXPERIMENTAL RADII
 EXPERIMENT 2

TRIMMED 1.5 FEET BY THE STERN, DISPLACEMENT 26,350 TONS

HARMONIC	=	1	2	3	4	5	6	7	8
RADIUS =	•359								
AMPLITUDE =	-•0532	-•0069	•0527	-•0119	•0131	-•0104	•0059	-•0060	
RADIUS =	•550								
AMPLITUDE =	-•1190	-•0243	•0166	•0015	•0180	-•0123	•0093	-•0116	
RADIUS =	•774								
AMPLITUDE =	-•1341	-•0370	-•0137	-•0003	•0057	-•0024	•0067	-•0060	
RADIUS =	1.317								
AMPLITUDE =	-•1294	-•0445	-•0247	-•0070	-•0352	-•0055	-•0016	•0009	
RADIUS =	1.178								
AMPLITUDE =	-•1354	-•0588	-•0282	-•0117	-•0074	-•0047	-•0038	-•0022	

Table B5 - HARMONIC ANALYSIS OF TANGENTIAL VELOCITY COMPONENT RATIOS
 AT THE INTERPOLATED RADII
 EXPERIMENT 2

TRIMMED 1.5 FEET BY THE STERN, DISPLACEMENT 26,390 TONS

HARMONIC	=	1	2	3	4	5	6	7	8
RADIUS =	• 200								
AMPLITUDE =		• 0361	• 0113	• 0078	• 0030	• 0019	• 0014	• 0009	• 0059
RADIUS =	• 350								
AMPLITUDE =		-• 0238	-• 0005	• 0651	-• 0186	• 0387	-• 0078	• 0038	-• 0024
RADIUS =	• 400								
AMPLITUDE =		-• 0704	-• 0109	• 0445	-• 0079	• 0154	-• 0116	• 0071	-• 0080
RADIUS =	• 500								
AMPLITUDE =		-• 1053	-• 0139	• 0260	-• 0009	• 0192	-• 0128	• 0089	-• 0111
RADIUS =	• 600								
AMPLITUDE =		-• 1235	-• 0270	• 0089	• 0014	• 0153	-• 0097	• 0092	-• 0135
RADIUS =	• 700								
AMPLITUDE =		-• 1310	-• 0329	-• 0056	• 0008	• 0096	-• 0049	• 0081	-• 0080
RADIUS =	• 800								
AMPLITUDE =		-• 1324	-• 0380	-• 0152	-• 0010	• 0041	-• 0017	• 0055	-• 0046
RADIUS =	• 900								
AMPLITUDE =		-• 1296	-• 0423	-• 0203	-• 0037	-• 0011	-• 0002	• 0017	-• 0007
RADIUS = 1.000									
AMPLITUDE =		-• 1292	-• 0475	-• 0242	-• 0065	-• 0047	-• 0003	-• 0012	• 0009

Table B6 - INPUT DATA FOR WAKE SURVEY ANALYSES
 EXPERIMENT 2
 TRIMMED 1.5 FEET BY THE STERN, DISPLACEMENT 26,390 TONS

ANGLE	RADIUS RATIO = 0.359			RADIUS RATIO = 0.556			RADIUS RATIO = 0.774		
	Vx/Vy	Vz/Vy	ANGLE	Vx/Vy	Vz/Vy	ANGLE	Vx/Vy	Vz/Vy	ANGLE
-0.2	-0.127	-0.011	-0.603	-0.5	-0.02	-0.603	-0.4	-0.235	-0.15
1.0	-0.145	-0.012	-0.027	-0.5	-0.01	-0.163	-0.5	-0.24	-0.19
1.7	-0.145	-0.012	-0.025	-0.5	-0.01	-0.114	-0.5	-0.251	-0.142
9.0	-0.247	-0.011	-0.121	-0.5	-0.01	-0.114	-0.5	-0.251	-0.142
12.5	-0.252	-0.012	-0.100	-0.5	-0.01	-0.103	-0.5	-0.251	-0.142
19.5	-0.290	-0.017	-0.087	-0.5	-0.01	-0.097	-0.5	-0.251	-0.142
29.3	-0.322	-0.011	-0.061	-0.5	-0.01	-0.086	-0.5	-0.251	-0.142
39.1	-0.376	-0.013	-0.052	-0.5	-0.01	-0.077	-0.5	-0.251	-0.142
51.7	-0.422	-0.012	-0.046	-0.5	-0.01	-0.067	-0.5	-0.251	-0.142
64.1	-0.510	-0.010	-0.039	-0.5	-0.01	-0.057	-0.5	-0.251	-0.142
64.3	-0.718	-0.017	-0.031	-0.5	-0.01	-0.047	-0.5	-0.251	-0.142
126.2	-0.412	-0.024	-0.124	-0.5	-0.01	-0.037	-0.5	-0.251	-0.142
156.5	-0.712	-0.016	-0.125	-0.5	-0.01	-0.027	-0.5	-0.251	-0.142
162.9	-0.677	-0.019	-0.064	-0.5	-0.01	-0.017	-0.5	-0.251	-0.142
166.5	-0.425	-0.019	-0.031	-0.5	-0.01	-0.007	-0.5	-0.251	-0.142
172.0	-0.412	-0.011	-0.017	-0.5	-0.01	-0.003	-0.5	-0.251	-0.142
178.8	-0.274	-0.017	-0.002	-0.5	-0.01	-0.003	-0.5	-0.251	-0.142
179.4	-0.542	-0.012	-0.015	-0.5	-0.01	-0.003	-0.5	-0.251	-0.142
181.4	-0.543	-0.012	-0.023	-0.5	-0.01	-0.003	-0.5	-0.251	-0.142
184.5	-0.524	-0.011	-0.011	-0.5	-0.01	-0.003	-0.5	-0.251	-0.142
185.5	-0.544	-0.011	-0.007	-0.5	-0.01	-0.003	-0.5	-0.251	-0.142
192.0	-0.517	-0.011	-0.007	-0.5	-0.01	-0.003	-0.5	-0.251	-0.142
195.7	-0.416	-0.010	-0.007	-0.5	-0.01	-0.003	-0.5	-0.251	-0.142
211.5	-0.745	-0.012	-0.120	-0.5	-0.01	-0.003	-0.5	-0.251	-0.142
250.2	-0.745	-0.012	-0.121	-0.5	-0.01	-0.003	-0.5	-0.251	-0.142
259.3	-0.724	-0.011	-0.091	-0.5	-0.01	-0.003	-0.5	-0.251	-0.142
278.5	-0.675	-0.012	-0.067	-0.5	-0.01	-0.003	-0.5	-0.251	-0.142
307.1	-0.417	-0.012	-0.044	-0.5	-0.01	-0.003	-0.5	-0.251	-0.142
313.7	-0.314	-0.017	-0.051	-0.5	-0.01	-0.003	-0.5	-0.251	-0.142
317.5	-0.276	-0.017	-0.052	-0.5	-0.01	-0.003	-0.5	-0.251	-0.142
334.8	-0.015	-0.017	-0.052	-0.5	-0.01	-0.003	-0.5	-0.251	-0.142
344.7	-0.265	-0.011	-0.117	-0.5	-0.01	-0.003	-0.5	-0.251	-0.142
352.4	-0.272	-0.016	-0.131	-0.5	-0.01	-0.003	-0.5	-0.251	-0.142
355.9	-0.261	-0.017	-0.137	-0.5	-0.01	-0.003	-0.5	-0.251	-0.142
357.4	-0.274	-0.013	-0.133	-0.5	-0.01	-0.003	-0.5	-0.251	-0.142
359.0	-0.273	-0.014	-0.171	-0.5	-0.01	-0.003	-0.5	-0.251	-0.142
359.8	-0.271	-0.011	-0.009	-0.5	-0.01	-0.003	-0.5	-0.251	-0.142

APPENDIX C
RESULTS OF EXPERIMENT 3

Corresponding to
Trim 1.0 ft (0.305 m) by the Bow
Displacement 26,390 Tons (26,810 tonnes)

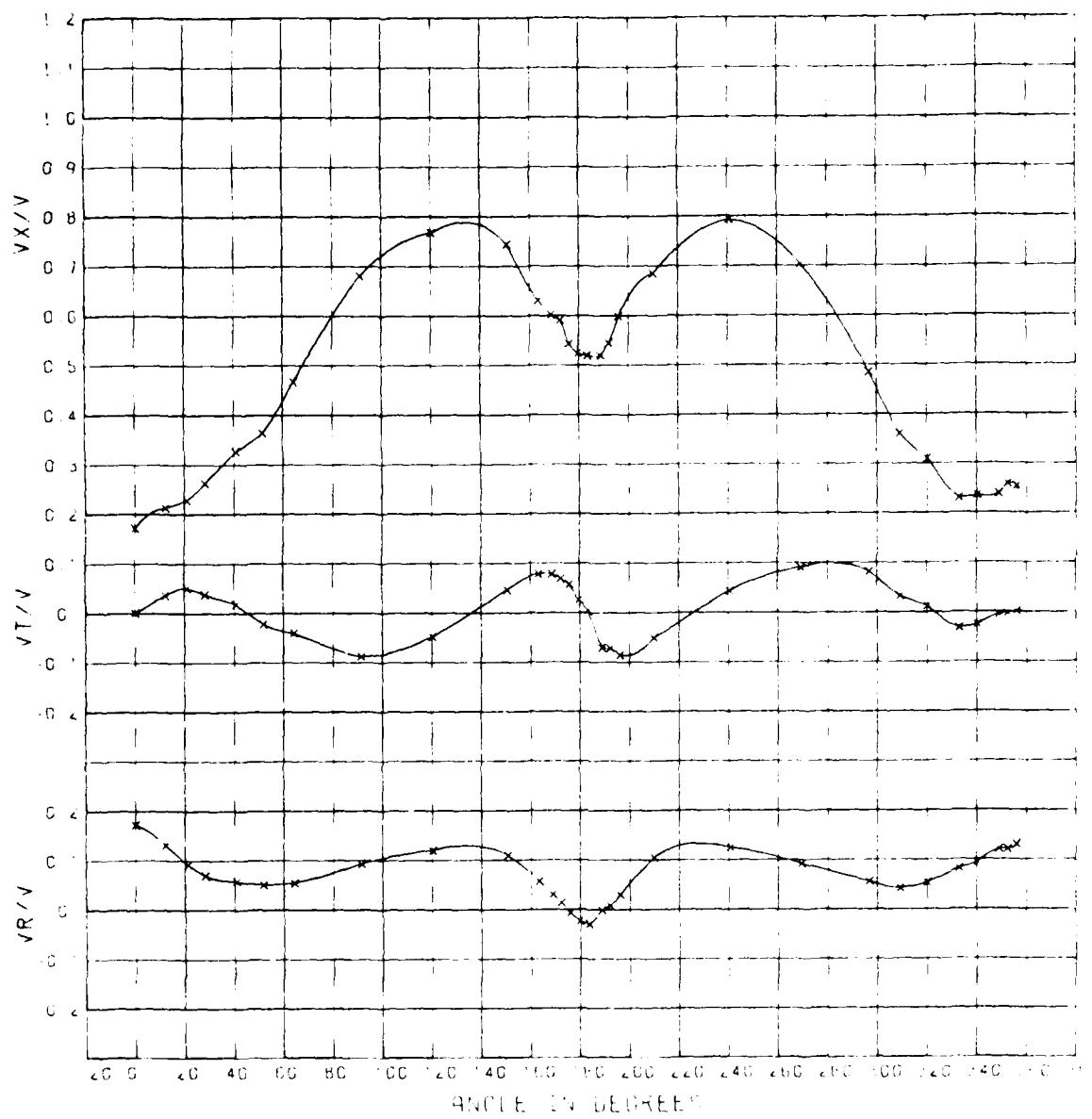


Figure C1 - CIRCUMFERENTIAL DISTRIBUTION OF THE LONGITUDINAL, TANGENTIAL AND RADIAL VELOCITY COMPONENT RATIOS - RADIUS RATIO = 0.359
EXPERIMENT 3

TRIMMED 1.0 FOOT BY THE BOW, DISPLACEMENT 26,390 TONS

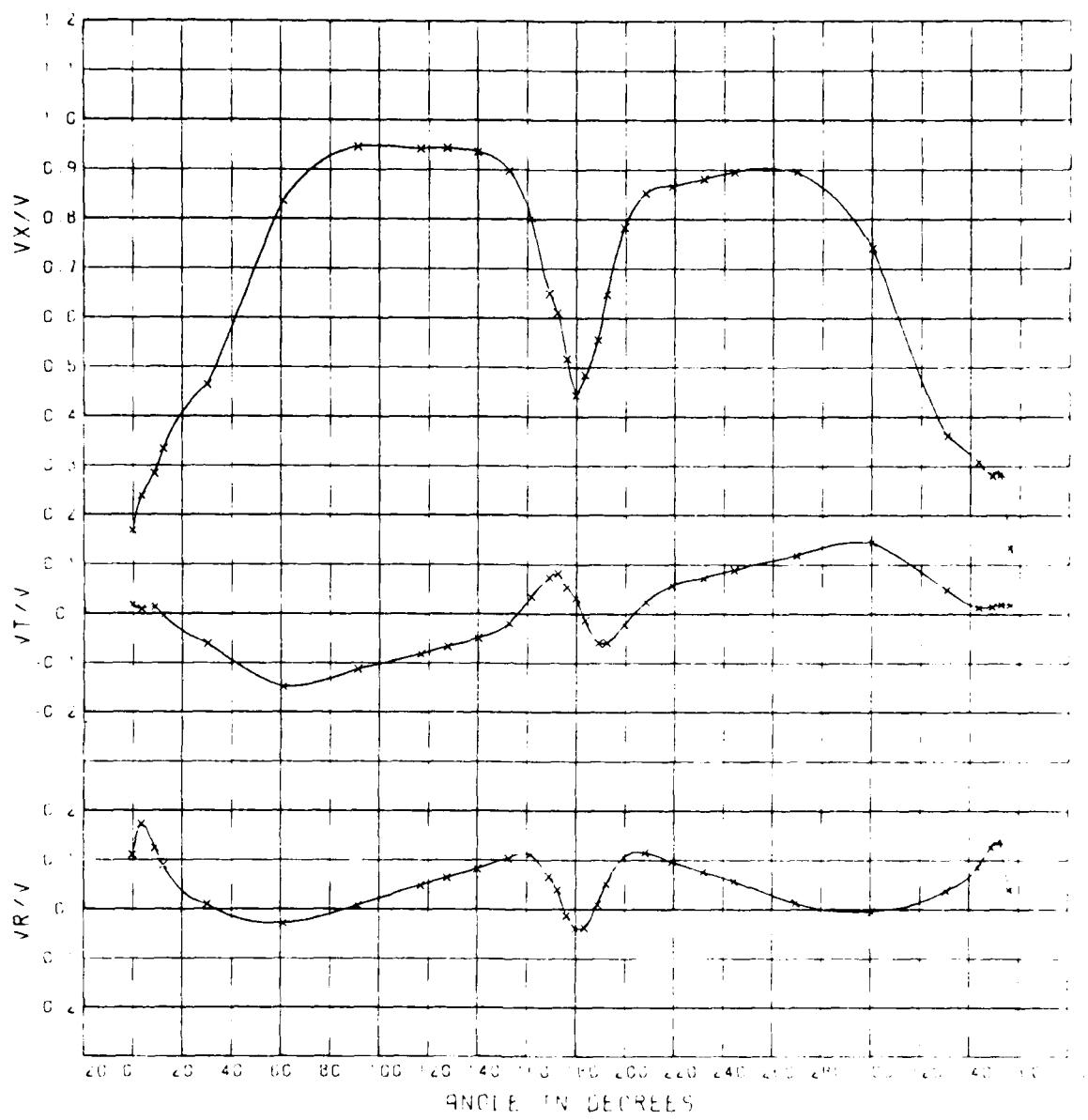


Figure C2 - CIRCUMFERENTIAL DISTRIBUTION OF THE LONGITUDINAL, TANGENTIAL AND RADIAL VELOCITY COMPONENT RATIOS - RADIUS RATIO = 0.556
EXPERIMENT 3

TRIMMED 1.0 FOOT BY THE BOW, DISPLACEMENT 26,390 TONS

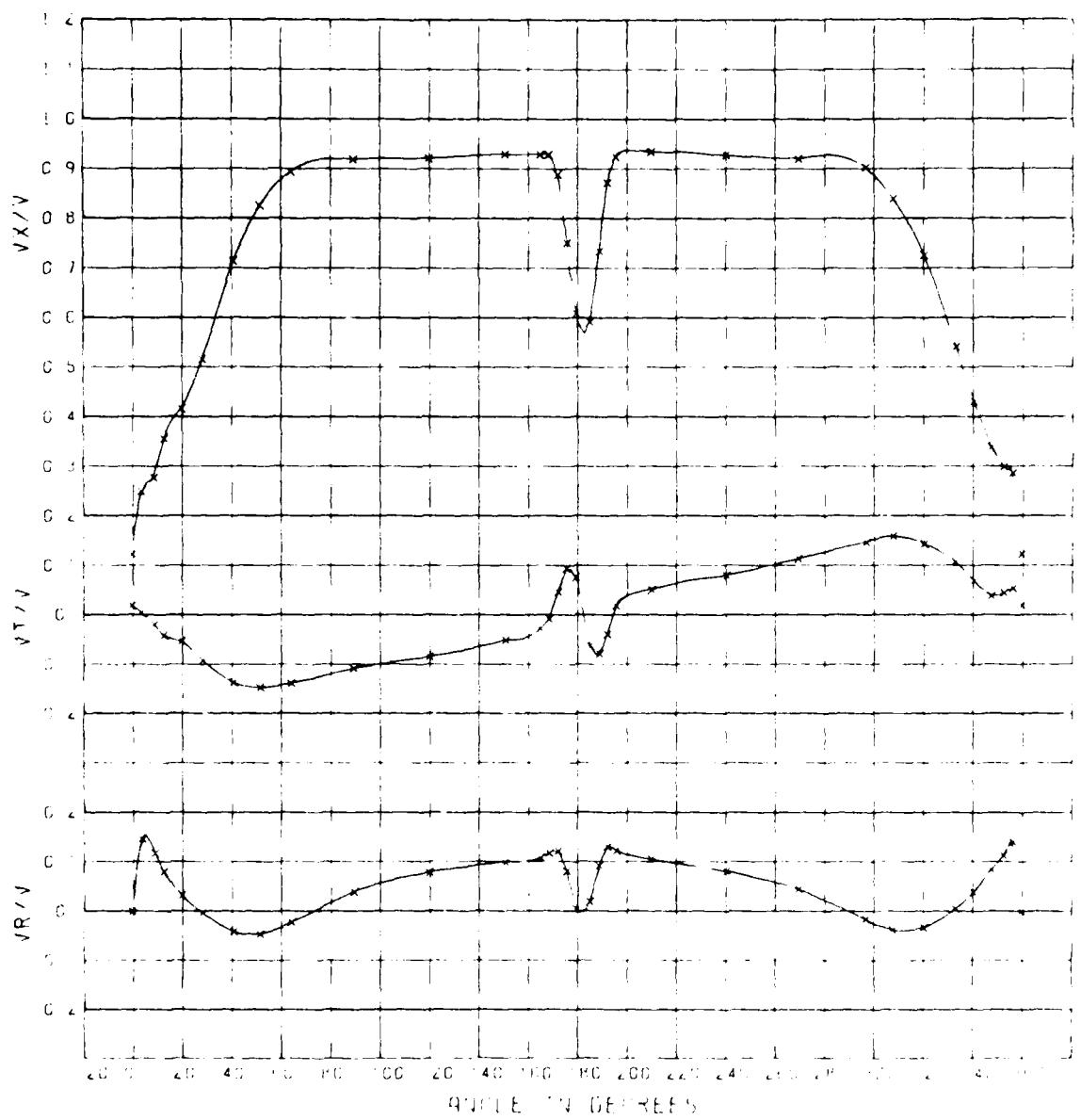


Figure C3 - CIRCUMFERENTIAL DISTRIBUTION OF THE LONGITUDINAL, TANGENTIAL AND RADIAL VELOCITY COMPONENT RATIOS - RADIUS RATIO = 0.774
EXPERIMENT 3

TRIMMED 1.0 FOOT BY THE BOW, DISPLACEMENT 26,390 TONS

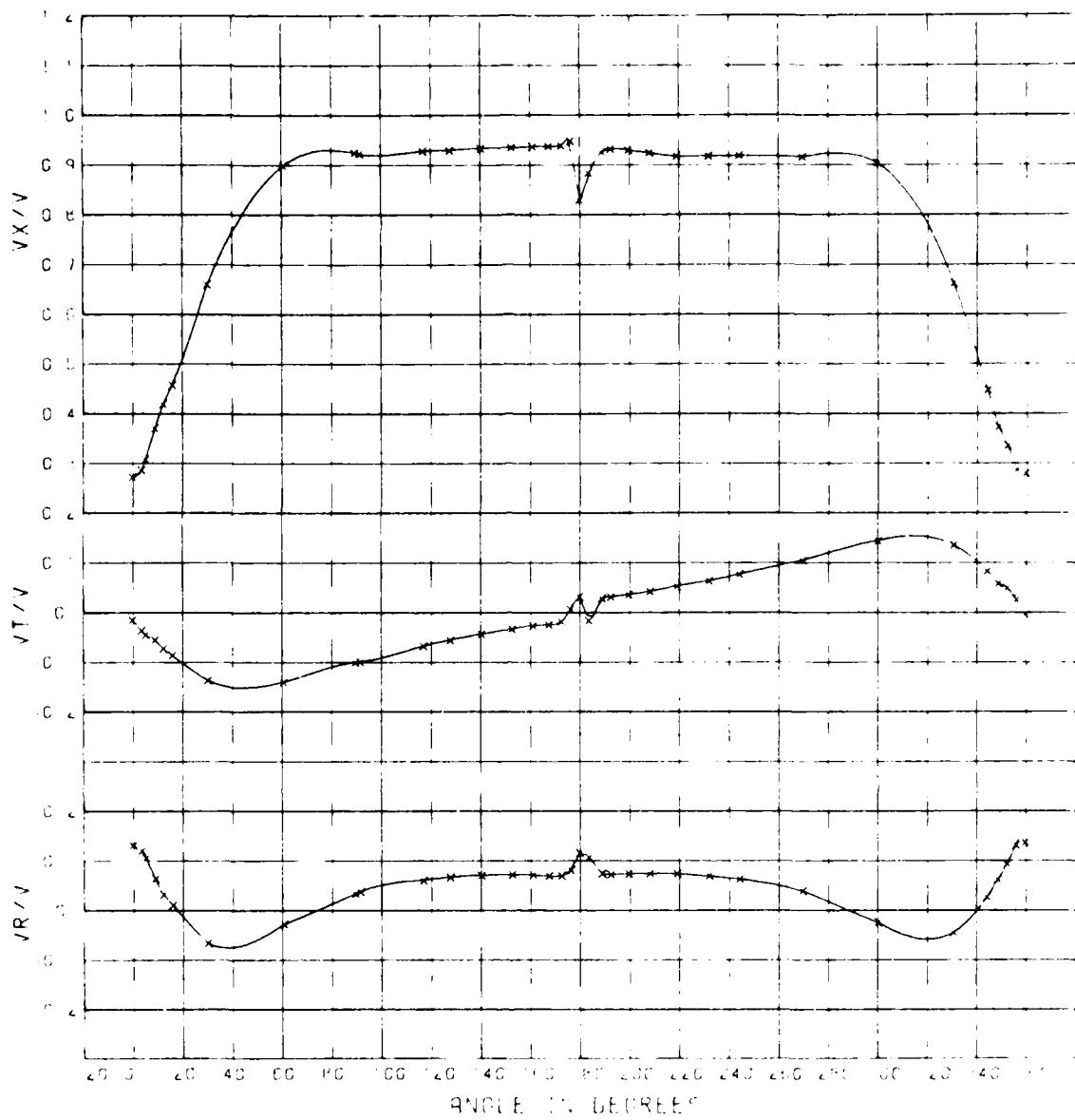


Figure C4 - CIRCUMFERENTIAL DISTRIBUTION OF THE LONGITUDINAL, TANGENTIAL AND RADIAL VELOCITY COMPONENT RATIOS - RADIUS RATIO = 1.017
EXPERIMENT 3

TRIMMED 1.0 FOOT BY THE BOW, DISPLACEMENT 26,390 TONS

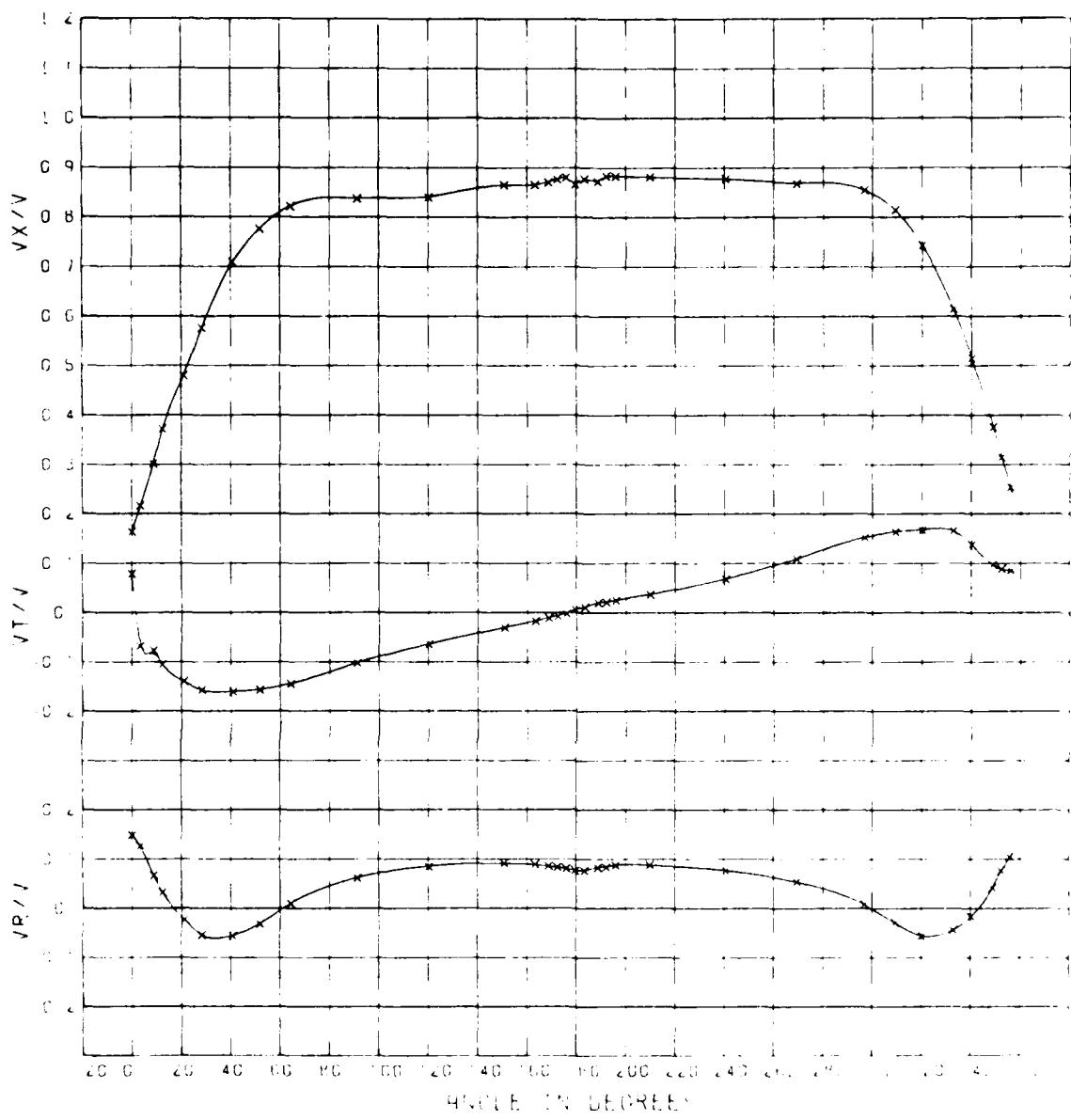


Figure C5 - CIRCUMFERENTIAL DISTRIBUTION OF THE LONGITUDINAL, TANGENTIAL AND RADIAL VELOCITY COMPONENT RATIOS - RADIUS RATIO = 1.178
EXPERIMENT 3

TRIMMED 1.0 FOOT BY THE BOW, DISPLACEMENT 26,390 TONS

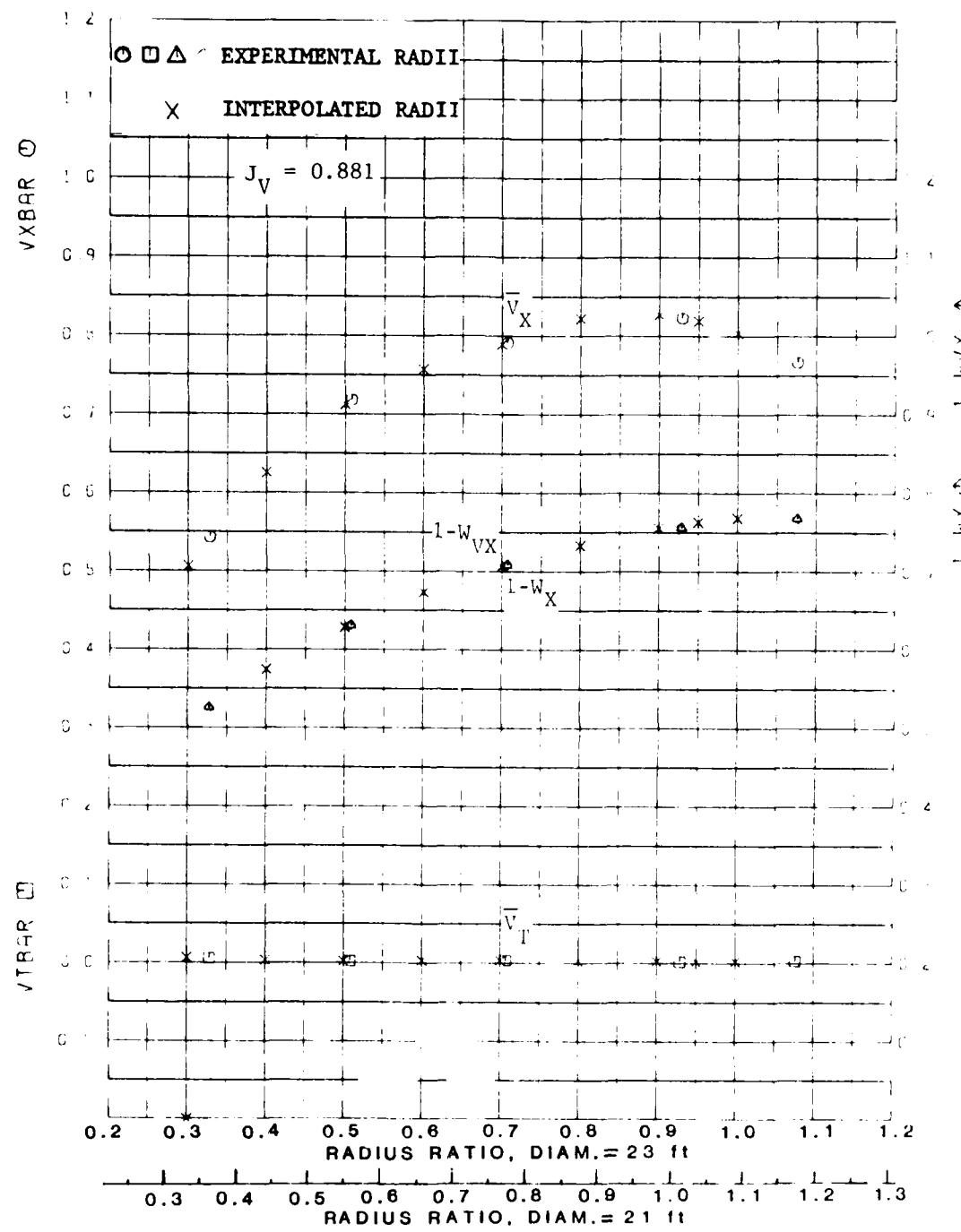


Figure C6 - RADIAL DISTRIBUTION OF THE MEAN VELOCITY COMPONENT RATIOS
EXPERIMENT 3

TRIMMED 1.0 FOOT BY THE BOW, DISPLACEMENT 26,390 TONS

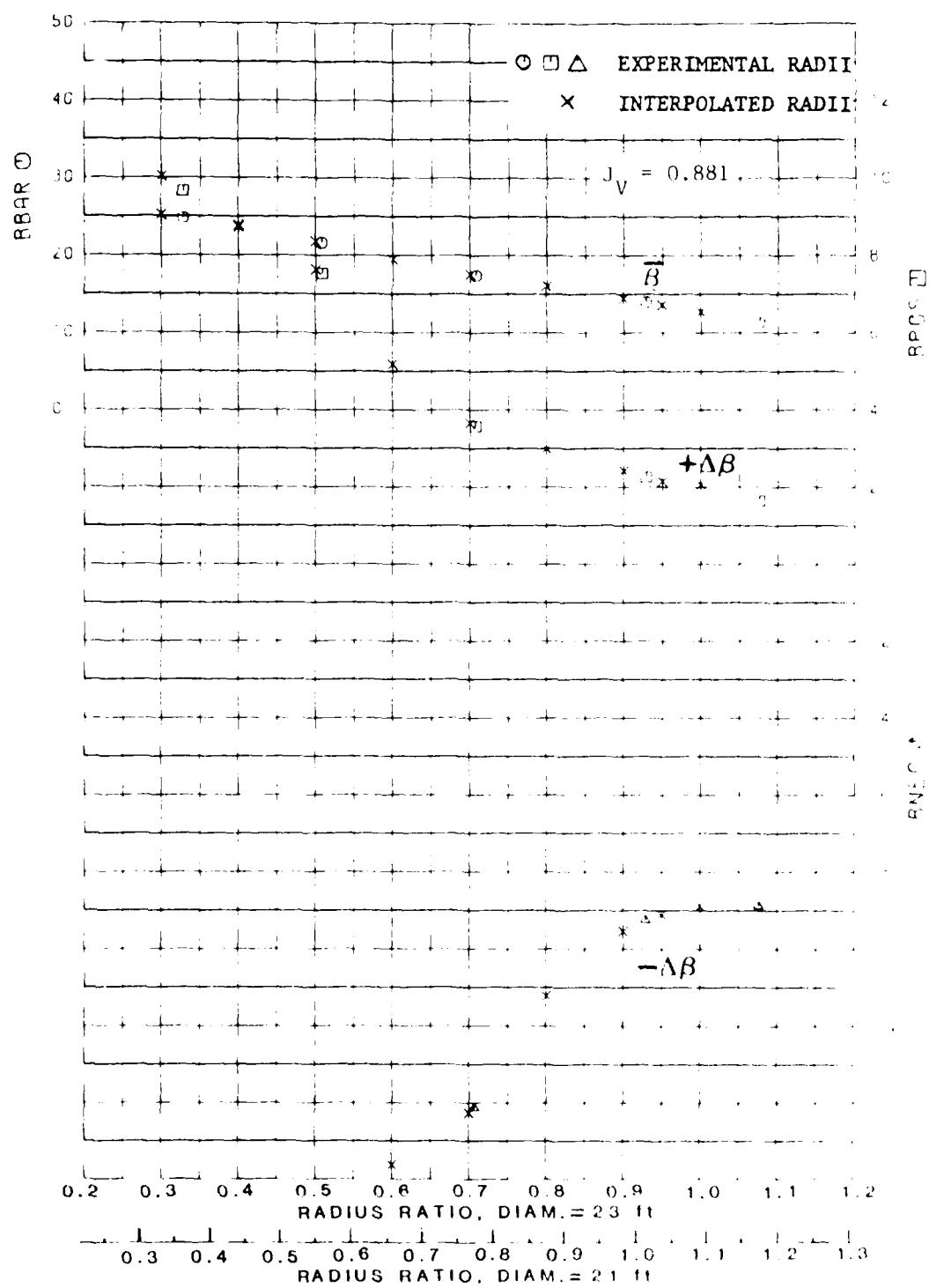


Figure C7 - RADIAL DISTRIBUTION OF THE MEAN ADVANCE ANGLE AND THE MAGNITUDE VARIATIONS OF THE ADVANCE ANGLE FOR MODEL 301 IN EXPERIMENT 3

TRIMMED 1.0 FOOT BY THE BOW, DISPLACEMENT 26,390 TONS

Table C1 - LISTING OF THE MEAN VELOCITY COMPONENT RATIOS, THE MEAN ADVANCE ANGLES AND OTHER DERIVED QUANTITIES
EXPERIMENT 3

TRIMMED 1.0 FEET BY THE BOW, DISPLACEMENT 26,390 TONS
 $D = 21$ ft (6.4 m), $J_V = 1.01$

RADIIUS \approx	0.353	0.556	0.774	1.017	1.176	1.200	1.100	0.903	0.503	0.600	0.700	0.800	0.900	1.000
VMBAR \approx	0.543	0.718	0.791	0.823	0.767	0.327	0.470	0.588	0.678	0.716	0.771	0.801	0.825	0.876
YMBAR \approx	0.005	0.002	0.003	0.032	0.003	0.511	0.307	0.064	0.372	0.002	0.003	0.003	0.002	0.012
Y2BAR \approx	0.015	0.040	0.040	0.043	0.030	0.155	0.107	0.272	0.048	0.043	0.046	0.043	0.033	0.037
J-MAX \approx	0.466	0.579	0.674	0.734	0.751	0.600	0.413	0.481	0.547	0.603	0.647	0.682	0.711	0.750
J-MIN \approx	0.470	0.581	0.676	0.735	0.752	0.600	0.418	0.482	0.548	0.605	0.649	0.683	0.712	0.755
DELTA \approx	25.82	22.51	15.17	14.58	11.62	27.30	26.58	25.21	23.53	21.50	19.48	17.83	16.42	14.83
BPOS \approx	5.9	7.8	3.75	2.37	1.71	23.04	12.02	9.49	6.43	6.65	6.68	3.59	3.00	2.51
THETA \approx	115.00	95.00	77.50	77.50	175.00	225.00	233.00	112.50	100.00	92.50	82.50	77.50	77.50	77.50
BEG \approx	-17.04	-16.94	-14.76	-9.67	-9.10	-16.10	-16.84	-17.19	-19.12	-17.18	-15.83	-13.83	-11.21	-9.61
THETA \approx	0.30	0.50	0.30	0.00	0.00	57.50	20.00	0.00	357.50	357.50	0.00	0.00	0.00	0.00

VMBAR IS MEAN POTENTIAL MEAN LONGITUDINAL VELOCITY.

YMBAR IS MEAN POTENTIAL MEAN TANGENTIAL VELOCITY.

Y2BAR IS MEAN POTENTIAL MEAN RADIAL VELOCITY.

DELTA IS POLARIMETRIC MEAN WAVE VELOCITY WITHOUT TANGENTIAL CORRECTION.

J-MAX IS MEAN ANGLE OF VARIATION BETWEEN THE MAXIMUM AND MEAN ADVANCE ANGLES (DELTA BETA PLUS).

J-MIN IS MEAN ANGLE OF VARIATION BETWEEN THE MINIMUM AND MEAN ADVANCE ANGLES (DELTA BETA MINUS).

BPOS IS VARIATION BETWEEN THE MAXIMUM AND MEAN ADVANCE ANGLES (DELTA BETA PLUS).

THETA IS VARIATION BETWEEN THE MINIMUM AND MEAN ADVANCE ANGLES (DELTA BETA MINUS).

ANGLE IS ANGLE IN DEGREES AT WHICH CORRESPONDING BPOS OR BNEG OCCURS.

Table C2 - HARMONIC ANALYSIS OF LONGITUDINAL VELOCITY COMPONENT RATIOS
 AT THE EXPERIMENTAL RADII
 EXPERIMENT 3

TRIMMED 1.0 FOOT BY THE BOW DISPLACEMENT 26,390 TONS

HARMONIC	=	1	2	3	4	5	6	7	8
RADIUS =	• 35.9								
AMPLITUDE =	• 240.8	• 145.0	• 061.3	• 013.6	• 018.8	• 013.2	• 008.1	• 004.8	
PHASE ANGLE =	269.0	267.4	75.5	240.2	85.7	251.9	142.9	211.6	
RADIUS =	• 55.6								
AMPLITUDE =	• 207.7	• 238.7	• 018.7	• 053.7	• 045.0	• 032.2	• 019.6	• 028.9	
PHASE ANGLE =	281.8	272.7	348.8	271.3	68.2	272.6	88.6	267.5	
RADIUS =	• 77.4								
AMPLITUDE =	• 197.5	• 186.5	• 073.0	• 072.1	• 010.7	• 029.5	• 021.2	• 024.9	
PHASE ANGLE =	269.6	268.7	268.4	266.4	64.9	260.4	69.4	259.7	
RADIUS = 1.017									
AMPLITUDE =	• 183.4	• 142.5	• 093.2	• 054.3	• 022.1	• 016.4	• 007.1	• 010.3	
PHASE ANGLE =	270.7	268.0	270.4	269.6	276.8	279.6	288.2	272.0	
RADIUS = 1.178									
AMPLITUDE =	• 180.9	• 127.5	• 090.7	• 053.0	• 028.6	• 019.1	• 013.6	• 010.6	
PHASE ANGLE =	263.2	268.0	266.2	265.6	267.1	268.8	262.9	269.0	

Table C3 - HARMONIC ANALYSIS OF LONGITUDINAL VELOCITY COMPONENT RATIOS
AT THE INTERPOLATED RADII
EXPERIMENT 3

TRIMMED 1.0 FOOT BY THE BOW DISPLACEMENT 26,366 TONS							
HARMONIC	=	1	2	3	4	5	6
RADIUS =	• 200						
AMPLITUDE =	• 311.1	• 0519	• 1104	• 0454	• 0427	• 0750	• 0711
PHASE ANGLE =	250.0	146.6	91.1	119.3	27.3.8	130.9	233.8
RADIUS =	• 300						
AMPLITUDE =	• 260.3	• 0925	• 0789	• 0128	• 0018	• 0082	• 0089
PHASE ANGLE =	262.0	259.0	82.6	154.8	3.8	197.6	194.3
RADIUS =	• 400						
AMPLITUDE =	• 230.8	• 1754	• 0495	• 0229	• 0287	• 0185	• 0104
PHASE ANGLE =	273.1	269.9	69.1	259.7	87.5	263.6	115.6
RADIUS =	• 500						
AMPLITUDE =	• 214.2	• 2256	• 0242	• 0442	• 0432	• 0288	• 0159
PHASE ANGLE =	280.1	272.3	36.0	270.2	98.4	271.9	94.8
RADIUS =	• 600						
AMPLITUDE =	• 204.7	• 2271	• 0249	• 0600	• 0376	• 0322	• 0217
PHASE ANGLE =	278.1	271.7	300.1	269.4	86.8	258.5	82.6
RADIUS =	• 700						
AMPLITUDE =	• 200.2	• 2028	• 0549	• 0695	• 0215	• 0313	• 0232
PHASE ANGLE =	271.6	269.8	272.5	257.1	81.3	262.4	74.2
RADIUS =	• 800						
AMPLITUDE =	• 195.4	• 1806	• 0765	• 0693	• 0056	• 0269	• 0175
PHASE ANGLE =	269.5	268.6	269.2	267.1	54.4	262.8	66.7
RADIUS =	• 900						
AMPLITUDE =	• 188.9	• 1605	• 0871	• 0635	• 0105	• 0198	• 0060
PHASE ANGLE =	271.6	268.3	270.8	269.1	291.7	272.7	36.8
RADIUS = 1,000							
AMPLITUDE =	• 184.0	• 1448	• 0928	• 0550	• 0207	• 0166	• 0061
PHASE ANGLE =	271.0	269.1	271.6	269.7	279.0	279.3	295.3

Table C4 - HARMONIC ANALYSIS OF TANGENTIAL VELOCITY COMPONENT RATIOS
 AT THE EXPERIMENTAL RADI 1
 EXPERIMENT 3

TRIMMED 1.0 FOOT BY THE BOW DISPLACEMENT 26,390 TONS

HARMONIC	=	1	2	3	4	5	6	7	8
RADIUS =	• 359								
AMPLITUDE =	• 0442	• 0133	• 0591	• 0091	• 0184	• 0089	• 0078	• 0057	
PHASE ANGLE =	170.8	178.4	353.3	164.7	2.9	158.2	328.4	176.4	
RADIUS =	• 556								
AMPLITUDE =	• 1162	• 0237	• 0161	• 0046	• 0215	• 0115	• 0106	• 0108	
PHASE ANGLE =	131.7	174.9	30.4	71.1	2.5	158.9	1.9	161.0	
RADIUS =	• 774								
AMPLITUDE =	• 1305	• 0366	• 0145	• 0025	• 0086	• 0116	• 0079	• 0072	
PHASE ANGLE =	178.2	174.8	183.3	75.5	1.2	112.9	359.3	156.4	
RADIUS = 1.017									
AMPLITUDE =	• 1248	• 0438	• 0248	• 0061	• 0040	• 0014	• 0016	• 0002	
PHASE ANGLE =	181.3	181.3	182.4	187.4	177.1	348.6	222.9	219.0	
RADIUS = 1.176									
AMPLITUDE =	• 1324	• 0627	• 0312	• 0142	• 0104	• 0064	• 0054	• 0033	
PHASE ANGLE =	173.7	179.4	179.9	174.0	175.5	169.2	166.9	150.4	

Table C5 - HARMONIC ANALYSIS OF TANGENTIAL VELOCITY COMPONENT RATIOS
 AT THE INTERPOLATED RADII
 EXPERIMENT 3

TRIMMED 1.0 FOOT BY THE BOW DISPLACEMENT 26,390 TONS							
HARMONIC	1	2	3	4	5	6	7
RADIUS = .260							
AMPLITUDE =	.0630	.0146	.1013	.248	.0058	.0119	.0110
PHASE ANGLE =	26.2	349.0	348.5	184.1	4.1	10.7	265.4
RADIUS = .300							
AMPLITUDE =	.0173	.0022	.3738	.0140	.0148	.0058	.0032
PHASE ANGLE =	128.6	199.5	351.3	175.4	3.0	155.3	304.3
RADIUS = .400							
AMPLITUDE =	.0635	.0153	.0496	.3066	.0202	.0103	.0084
PHASE ANGLE =	176.6	176.6	354.8	151.8	2.8	158.9	340.7
RADIUS = .500							
AMPLITUDE =	.1014	.0254	.0286	.0342	.0220	.0119	.0074
PHASE ANGLE =	181.2	175.2	359.5	95.0	2.6	159.3	357.5
RADIUS = .600							
AMPLITUDE =	.1203	.0308	.0298	.044	.0187	.0188	.0105
PHASE ANGLE =	180.5	174.2	3.3	67.1	2.3	156.4	1.6
RADIUS = .700							
AMPLITUDE =	.1272	.0338	.0059	.0035	.0128	.039	.0103
PHASE ANGLE =	178.7	173.9	183.6	65.3	1.6	144.5	.7
RADIUS = .800							
AMPLITUDE =	.1284	.0373	.0156	.0017	.0076	.0011	.0068
PHASE ANGLE =	178.9	176.4	183.5	82.1	1.1	70.3	356.1
RADIUS = .900							
AMPLITUDE =	.1248	.0416	.0199	.0020	.0016	.0025	.0030
PHASE ANGLE =	180.6	180.4	183.5	186.4	2.1	359.2	334.5
RADIUS = 1.000							
AMPLITUDE =	.1245	.0476	.0241	.0054	.0033	.0016	.0014
PHASE ANGLE =	181.3	181.3	182.6	188.6	177.1	350.0	240.9

Table C6 - INPUT DATA FOR WAKE SURVEY ANALYSES
EXPERIMENT 3

TRIMMED 1.0 FOOT BY THE BOW, DISPLACEMENT 26,340 TONS

ANGLE	RADIUS RATIO = 0.359			RADIUS RATIO = 0.556			RADIUS RATIO = 0.774		
	Vx/V	Vy/V	Vz/V	Vx/V	Vy/V	Vz/V	Vx/V	Vy/V	Vz/V
-0.1	0.172	0.01	0.168	-0.3	0.03	0.119	-0.19	0.122	-0.001
12.2	0.213	0.15	0.131	3.3	0.23	0.173	3.4	0.249	-0.145
21.0	0.224	0.14	0.092	0.7	0.243	0.125	3.3	0.249	-0.064
24.2	0.262	0.14	0.064	1.2	0.134	0.065	0.5	0.276	-0.117
40.8	0.326	0.13	0.054	3.0	0.064	0.019	0.5	0.386	-0.074
51.7	0.362	0.13	0.051	0.7	0.061	0.019	0.5	0.416	-0.054
66.1	0.459	0.14	0.052	0.7	0.036	0.027	0.5	0.515	-0.031
68.2	0.447	0.14	0.055	0.7	0.016	0.007	0.5	0.515	-0.008
91.1	0.647	0.15	0.049	114.7	0.943	0.007	40.6	0.139	-0.042
91.4	0.670	0.14	0.040	127.5	0.944	0.067	51.4	0.149	-0.027
127.2	0.767	0.13	0.040	140.0	0.936	0.020	63.9	0.893	-0.023
158.7	0.767	0.13	0.043	152.5	0.600	0.013	0.5	0.118	-0.033
174.6	0.746	0.13	0.043	152.8	0.495	0.028	0.5	0.110	-0.077
180.1	0.631	0.13	0.057	161.6	0.603	0.036	0.5	0.920	-0.045
188.7	0.652	0.13	0.079	169.9	0.649	0.067	0.5	0.150	-0.022
172.3	0.540	0.14	0.030	170.4	0.611	0.032	0.5	0.929	-0.094
175.8	0.543	0.14	0.037	176.1	0.517	0.013	0.5	0.112	-0.101
179.5	0.520	0.14	0.025	179.7	0.443	0.012	0.5	0.75	-0.046
179.7	0.504	0.14	0.025	183.4	0.485	0.037	0.5	0.605	-0.060
184.7	0.512	0.14	0.033	184.8	0.555	0.046	0.5	0.511	-0.011
184.7	0.523	0.14	0.062	192.2	0.648	0.055	0.5	0.64	-0.024
187.1	0.544	0.14	0.073	169.9	0.649	0.073	0.5	0.929	-0.101
187.1	0.540	0.14	0.074	170.4	0.611	0.032	0.5	0.929	-0.067
187.1	0.543	0.14	0.077	176.1	0.517	0.013	0.5	0.112	-0.140
179.5	0.520	0.14	0.025	179.7	0.443	0.012	0.5	0.75	-0.046
179.5	0.504	0.14	0.025	183.4	0.485	0.037	0.5	0.605	-0.060
179.7	0.512	0.14	0.033	184.8	0.555	0.046	0.5	0.511	-0.011
184.7	0.523	0.14	0.062	192.2	0.648	0.055	0.5	0.64	-0.024
187.1	0.544	0.14	0.073	169.9	0.649	0.073	0.5	0.929	-0.101
187.1	0.540	0.14	0.074	170.4	0.611	0.032	0.5	0.929	-0.067
187.1	0.543	0.14	0.077	176.1	0.517	0.013	0.5	0.112	-0.140
179.5	0.520	0.14	0.025	179.7	0.443	0.012	0.5	0.75	-0.046
179.5	0.504	0.14	0.025	183.4	0.485	0.037	0.5	0.605	-0.060
179.7	0.512	0.14	0.033	184.8	0.555	0.046	0.5	0.511	-0.011
184.7	0.523	0.14	0.062	192.2	0.648	0.055	0.5	0.64	-0.024
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187.1	0.543	0.14	0.077	176.1	0.517	0.013	0.5	0.112	-0.140
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179.5	0.504	0.14	0.025	183.4	0.485	0.037	0.5	0.605	-0.060
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184.7	0.523	0.14	0.062	192.2	0.648	0.055	0.5	0.64	-0.024
187.1	0.544	0.14	0.073	169.9	0.649	0.073	0.5	0.929	-0.101
187.1	0.540	0.14	0.074	170.4	0.611	0.032	0.5	0.929	-0.067
187.1	0.543	0.14	0.077	176.1	0.517	0.013	0.5	0.112	-0.140
179.5	0.520	0.14	0.025	179.7	0.443	0.012	0.5	0.75	-0.046
179.5	0.504	0.14	0.025	183.4	0.485	0.037	0.5	0.605	-0.060
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184.7	0.523	0.14	0.062	192.2	0.648	0.055	0.5	0.64	-0.024
187.1	0.544	0.14	0.073	169.9	0.649	0.073	0.5	0.929	-0.101
187.1	0.540	0.14	0.074	170.4	0.611	0.032	0.5	0.929	-0.067
187.1	0.543	0.14	0.077	176.1	0.517	0.013	0.5	0.112	-0.140
179.5	0.520	0.14	0.025	179.7	0.443	0.012	0.5	0.75	-0.046
179.5	0.504	0.14	0.025	183.4	0.485	0.037	0.5	0.605	-0.060
179.7	0.512	0.14	0.033	184.8	0.555	0.046	0.5	0.511	-0.011
184.7	0.523	0.14	0.062	192.2	0.648	0.055	0.5	0.64	-0.024
187.1	0.544	0.14	0.073	169.9	0.649	0.073	0.5	0.929	-0.101
187.1	0.540	0.14	0.074	170.4	0.611	0.032	0.5	0.929	-0.067
187.1	0.543	0.14	0.077	176.1	0.517	0.013	0.5	0.112	-0.140
179.5	0.520	0.14	0.025	179.7	0.443	0.012	0.5	0.75	-0.046
179.5	0.504	0.14	0.025	183.4	0.485	0.037	0.5	0.605	-0.060
179.7	0.512	0.14	0.033	184.8	0.555	0.046	0.5	0.511	-0.011
184.7	0.523	0.14	0.062	192.2	0.648	0.055	0.5	0.64	-0.024
187.1	0.544	0.14	0.073	169.9	0.649	0.073	0.5	0.929	-0.101
187.1	0.540	0.14	0.074	170.4	0.611	0.032	0.5	0.929	-0.067
187.1	0.543	0.14	0.077	176.1	0.517	0.013	0.5	0.112	-0.140
179.5	0.520	0.14	0.025	179.7	0.443	0.012	0.5	0.75	-0.046
179.5	0.504	0.14	0.025	183.4	0.485	0.037	0.5	0.605	-0.060
179.7	0.512	0.14	0.033	184.8	0.555	0.046	0.5	0.511	-0.011
184.7	0.523	0.							

APPENDIX D
RESULTS OF EXPERIMENT 4

Corresponding to
Trim 3.5 ft (1.067 m) by the Bow
Displacement 26,390 Tons (26,810 tonnes)

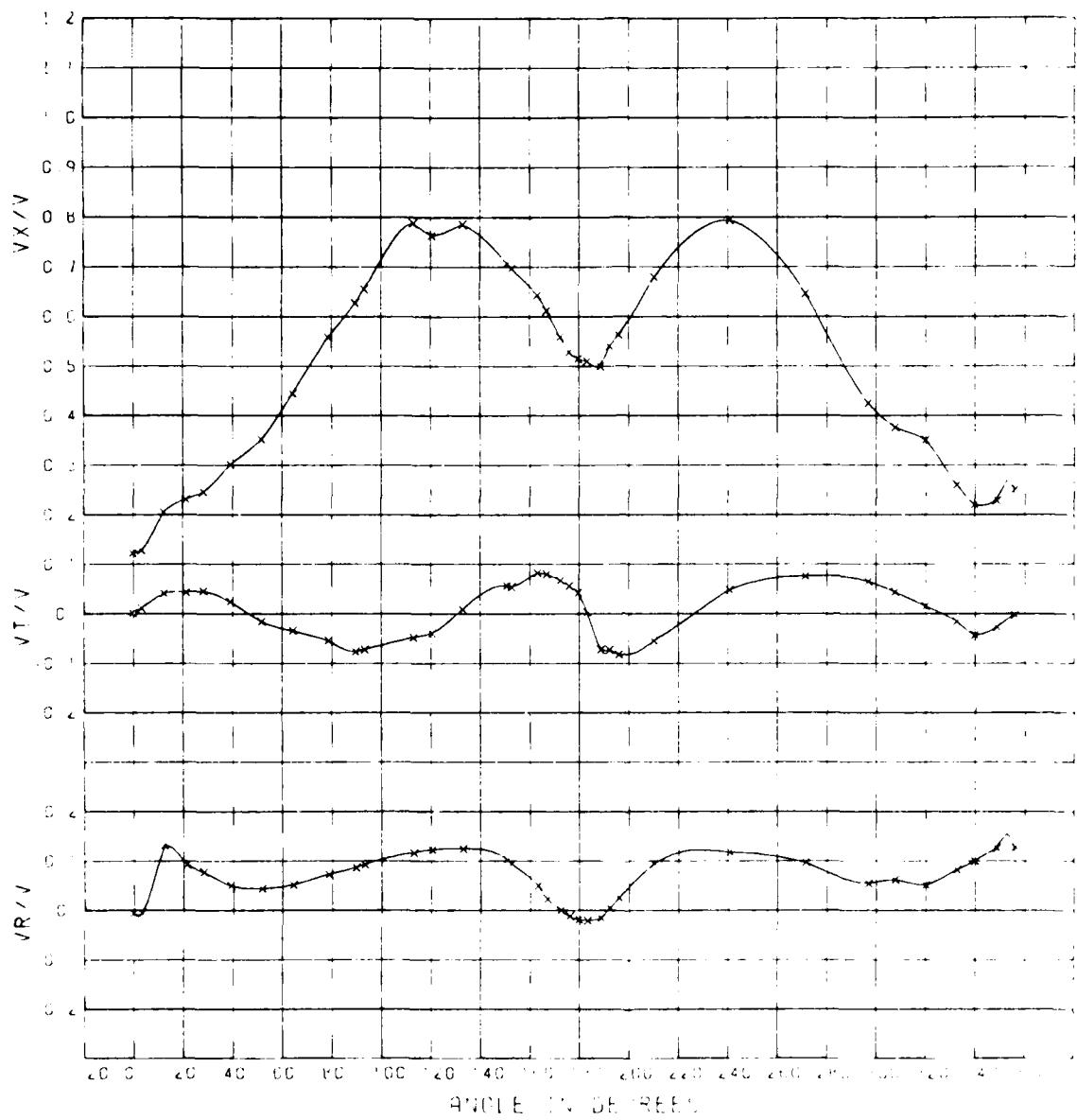


Figure D1 - CIRCUMFERENTIAL DISTRIBUTION OF THE LONGITUDINAL, TANGENTIAL AND RADIAL VELOCITY COMPONENT RATIOS - RADIUS RATIO = 0.359
EXPERIMENT 4

TRIMMED 3.5 FEET BY THE BOW, DISPLACEMENT 26,390 TONS

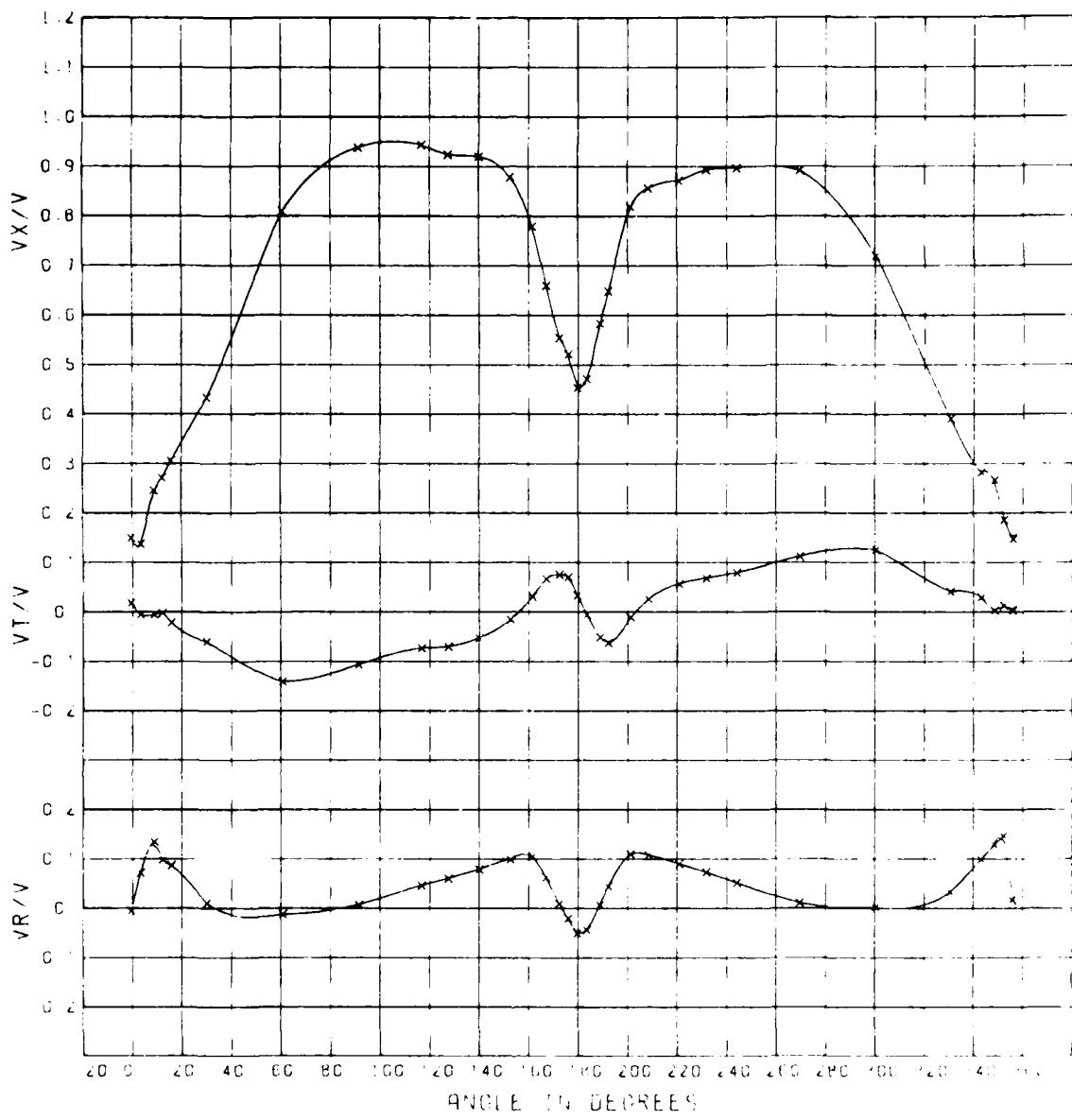


Figure D2 - CIRCUMFERENTIAL DISTRIBUTION OF THE LONGITUDINAL, TANGENTIAL AND RADIAL VELOCITY COMPONENT RATIOS - RADIUS RATIO = 0.556
EXPERIMENT 4

TRIMMED 3.5 FEET BY THE BOW, DISPLACEMENT 26,390 TONS

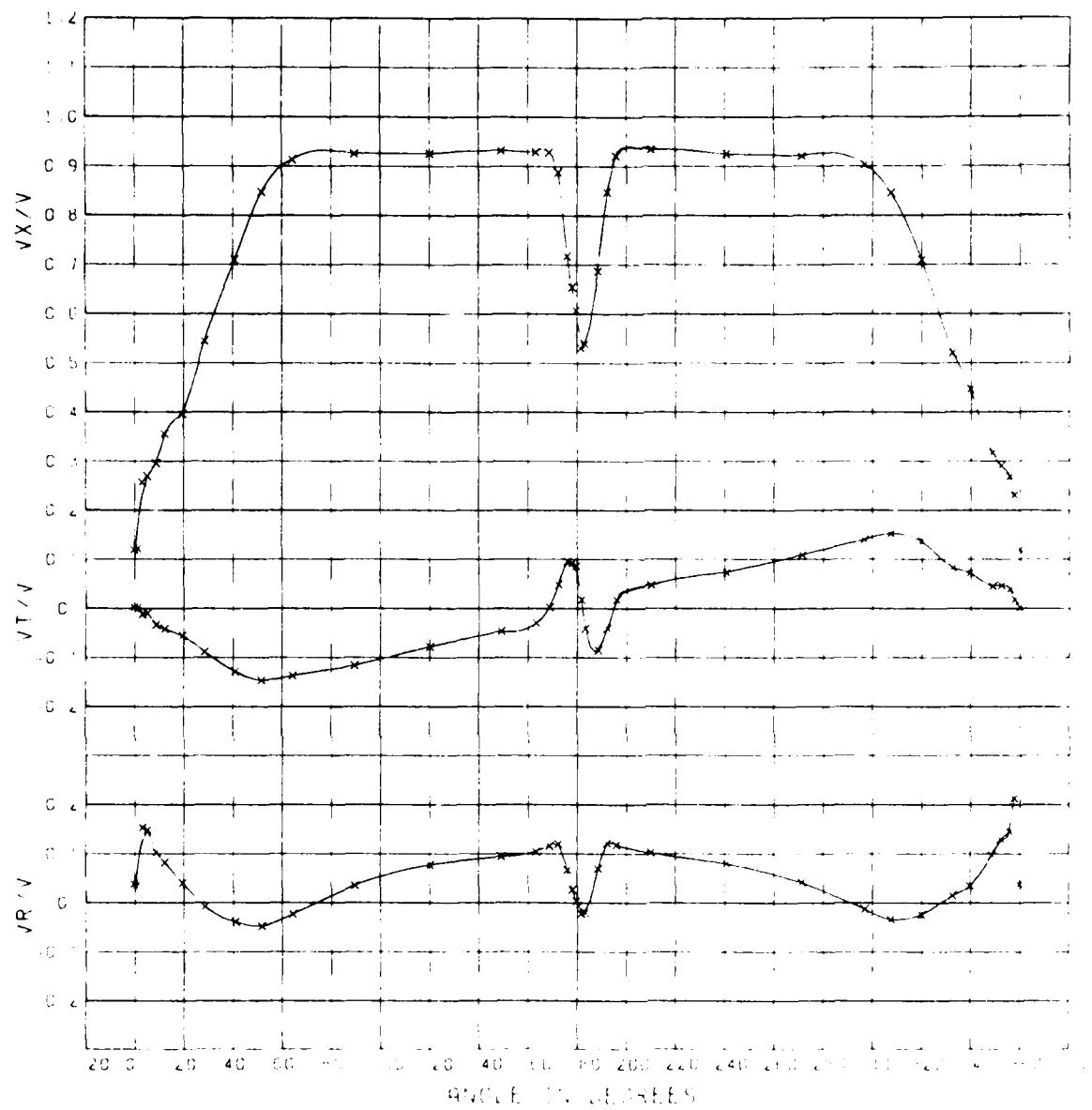


Figure D3 - CIRCUMFERENTIAL DISTRIBUTION OF THE LONGITUDINAL, TANGENTIAL AND RADIAL VELOCITY COMPONENT RATIOS - RADIUS RATIO = 0.774
EXPERIMENT 4

TRIMMED 3.5 FEET BY THE BOW, DISPLACEMENT 26,390 TONS

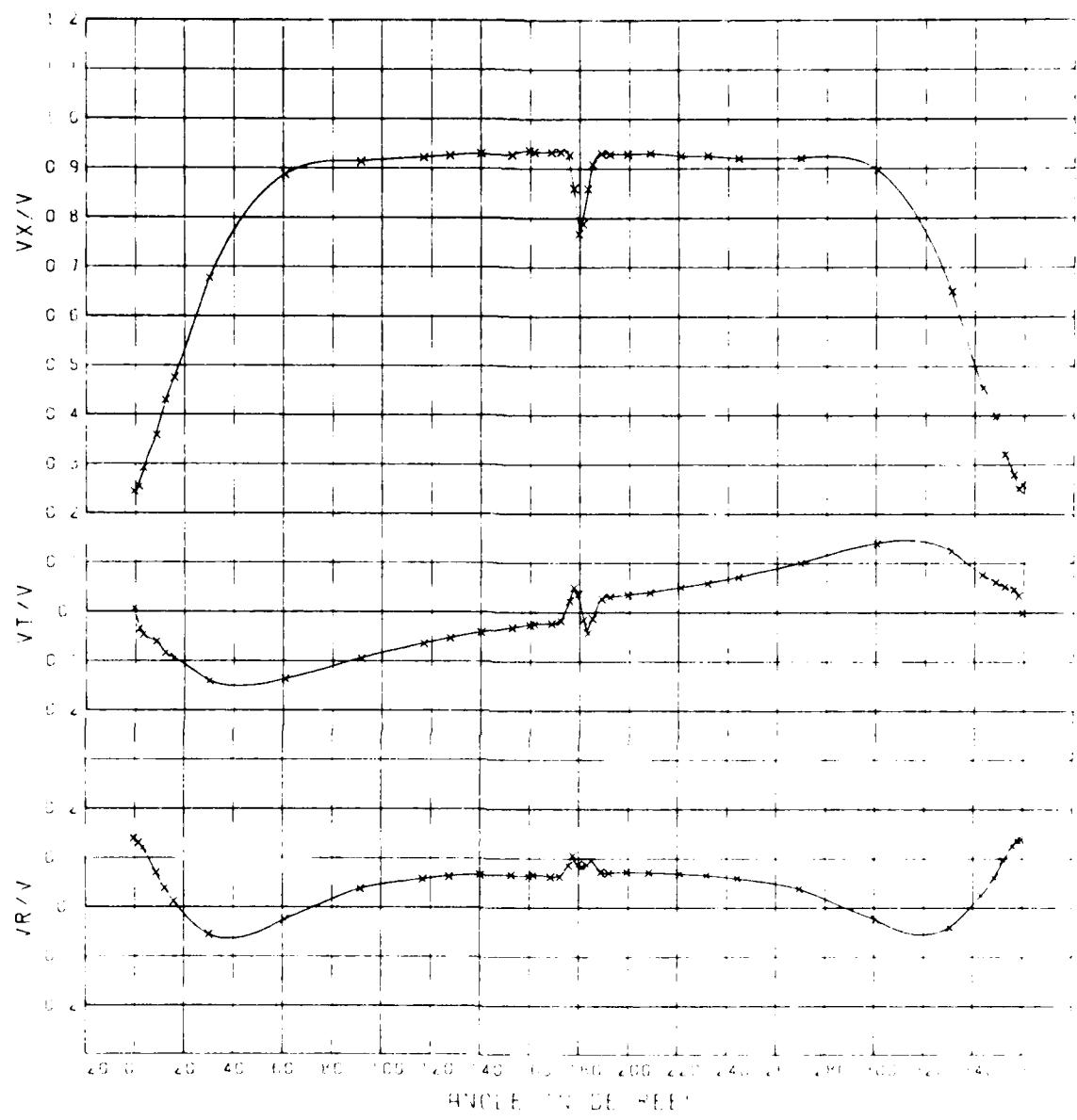


Figure D4 - CIRCUMFERENTIAL DISTRIBUTION OF THE LONGITUDINAL, TANGENTIAL, AND RADIAL VELOCITY COMPONENT RATIOS - RADIUS RATIO = 1.017
EXPERIMENT 4

TRIMMED 3.5 FEET BY THE BOW, DISPLACEMENT 26,300 TONS

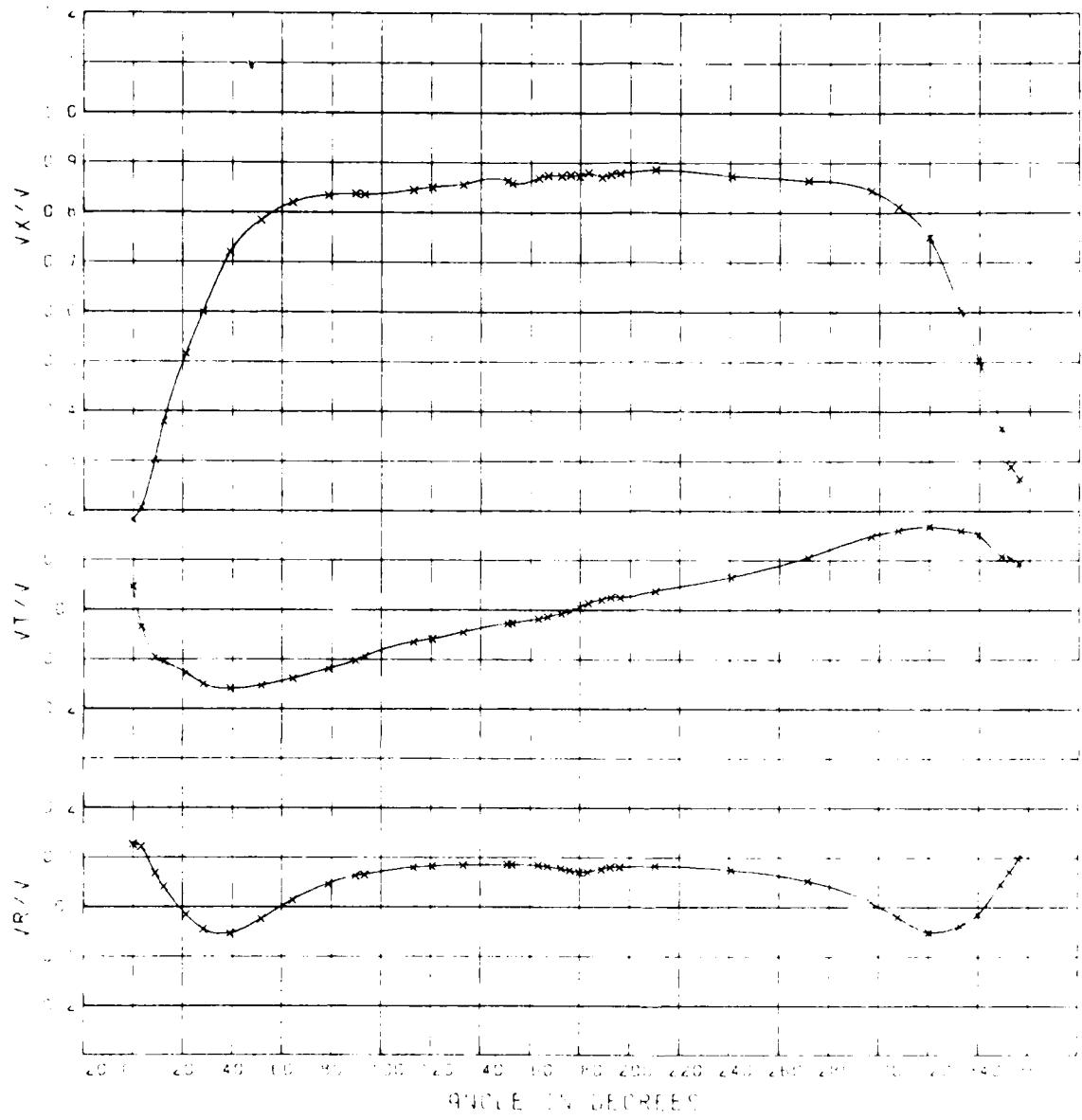


Figure D5 - CIRCUMFERENTIAL DISTRIBUTION OF THE LONGITUDINAL, TANGENTIAL AND RADIAL VELOCITY COMPONENT RATIOS - RADIUS RATIO = 1.178
EXPERIMENT 4

TRIMMED 3.5 FEET BY THE BOW, DISPLACEMENT 26,390 TONS

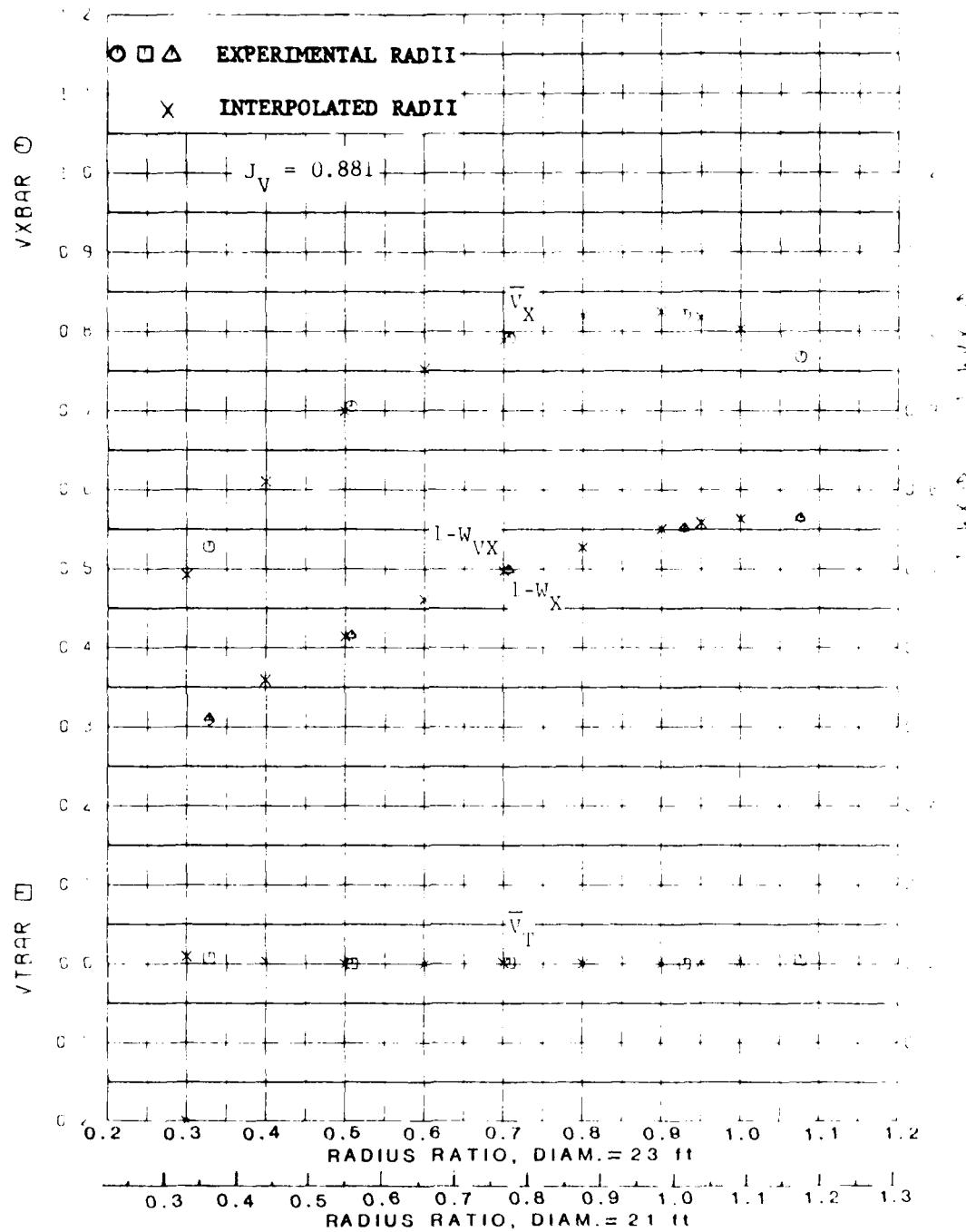


Figure D6 - RADIAL DISTRIBUTION OF THE MEAN VELOCITY COMPONENT RATIOS
EXPERIMENT 4

TRIMMED 3.5 FEET BY THE BOW, DISPLACEMENT 26,300 TONS

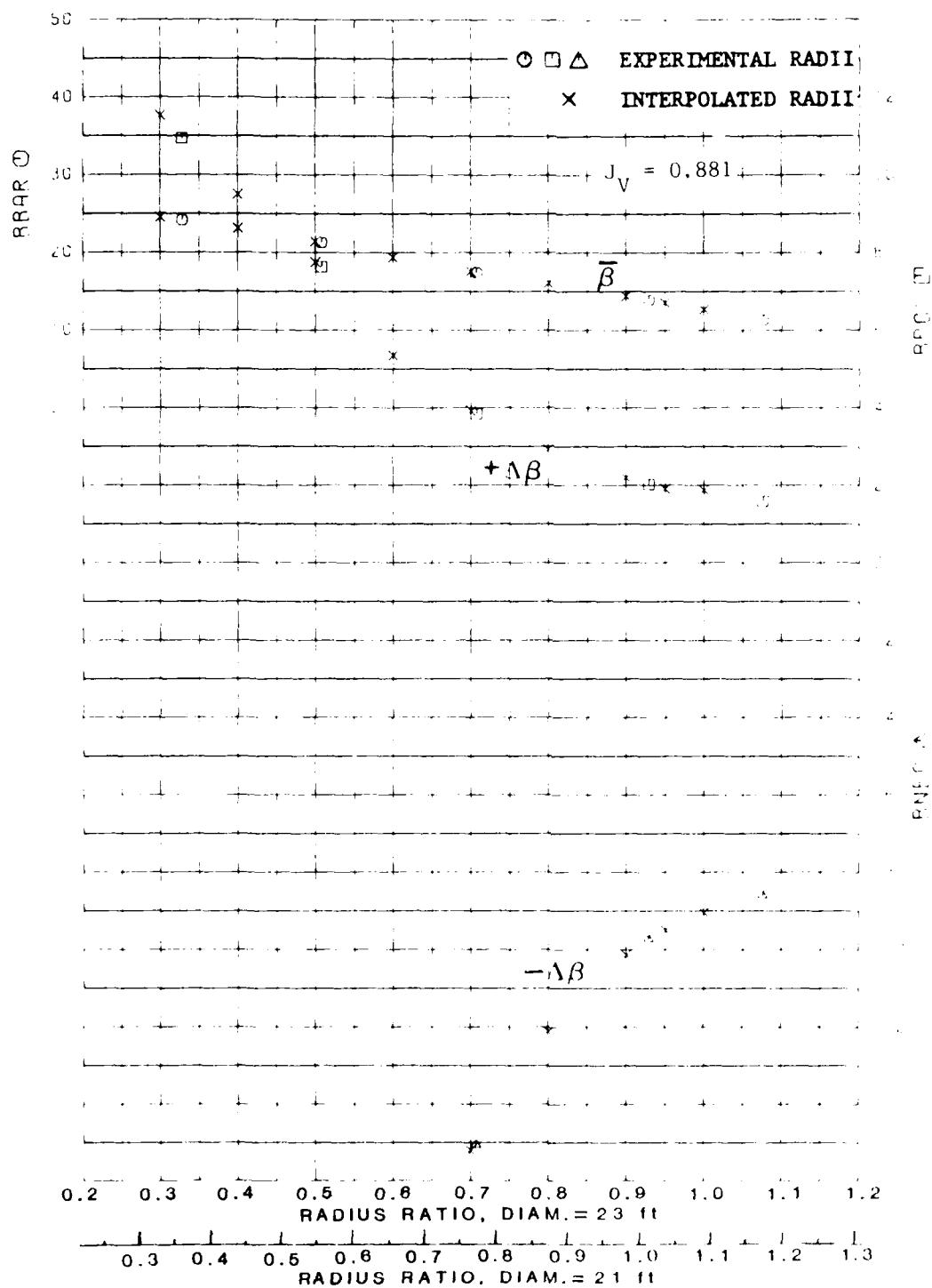


Figure D7 - RADIAL DISTRIBUTION OF THE MEAN ADVANCE ANGLE AND THE MAXIMUM VARIATIONS OF THE ADVANCE ANGLE FOR MODEL 3/6, EXPERIMENT 4

TRIMMED 3.5 FEET BY THE BOW, DISPLACEMENT 14,000 TONS

Table. D1 - LISTING OF THE MEAN VELOCITY COMPONENT RATIOS, THE MEAN ADVANCE ANGLES AND OTHER DERIVED QUANTITIES
EXPERIMENT 4

TRIMMED 3.5 FEET BY THE BOW, DISPLACEMENT 26,390 TONS
 $D = 21$ ft (6.4 m), $J_V = 1.01$

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1. *Antennaria* *lanceolata* (L.) Greene, *Bot. Gaz.* 27: 120. 1894.

THE BOSTONIAN, APRIL 25, 1851.

15 MPa at 62 °C at 24% E.

IS ANGEL IN DURKES AT WHICH CONVERSATION AND POS OF ANGEL OCCURS.

Table D2 - HARMONIC ANALYSIS OF LONGITUDINAL VELOCITY COMPONENT RATIOS
 AT THE EXPERIMENTAL RADII
 EXPERIMENT 4

TRIMMED 3.5 FEET BY THE BOW, DISPLACEMENT 26,390 TONS

PERIOD (SEC)	1	2	3	4	5	6	7	8
2.40115 = 0.359	-0.2407	-0.1372	0.0678	-0.0269	0.0301	-0.0119	0.0051	-0.0031
2.40116 = 0.350	-0.2147	-0.2466	-0.0363	-0.0635	0.0351	-0.0366	0.0197	-0.0265
2.40117 = 0.351	-0.1976	-0.1938	-0.0742	-0.0751	0.0143	-0.0295	0.0221	-0.0279
2.40118 = 1.017	-0.1926	-0.1422	-0.0886	-0.0576	-0.0221	-0.0204	-0.0066	-0.0127
2.40119 = 1.017	-0.1793	-0.1250	-0.0891	-0.0574	-0.0323	-0.0214	-0.0146	-0.0091

AD-A97 930

DAVID W TAYLOR NAVAL SHIP RESEARCH AND DEVELOPMENT CE--ETC, F/6 20/4
MEASUREMENTS OF THE EFFECT OF TRIM ON THE NOMINAL WAKE OF THE N--ETC(U)
MAR 81 M B WILSON, G A HAMPTON

UNCLASSIFIED DTNSRDC/SPD-0544-19

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Table D3 - HARMONIC ANALYSIS OF LONGITUDINAL VELOCITY COMPONENT RATIOS
 AT THE INTERPOLATED RADII
 EXPERIMENT 4

TRIMMED 2.5 FEET BY THE BOW, DISPLACEMENT 26,390 TONS							
HARMONIC	=	1	2	3	4	5	6
RADIUS = .200							
AMPLITUDE =	- .2687	.0603	.1357	.0144	-.0655	.0296	-.0153
RADIUS = .300							
AMPLITUDE =	- .2504	-.0754	.0921	-.0131	-.0254	.0012	-.0016
RADIUS = .400							
AMPLITUDE =	- .2345	-.1723	.0515	-.0355	.0116	-.0195	.0091
RADIUS = .500							
AMPLITUDE =	- .2212	-.2307	.0138	-.0536	.0393	-.0326	.0168
RADIUS = .600							
AMPLITUDE =	- .2106	-.2354	-.0240	-.0658	.0318	-.0353	.0223
RADIUS = .700							
AMPLITUDE =	- .2025	-.2110	-.0569	-.0733	.0226	-.0320	.0242
RADIUS = .800							
AMPLITUDE =	- .1949	-.1866	-.0765	-.0722	.0392	-.0279	.0181
RADIUS = .900							
AMPLITUDE =	- .1881	-.1632	-.0837	-.0635	-.0978	-.0232	.0047
RADIUS = 1.000							
AMPLITUDE =	- .1832	-.1448	-.0881	-.0582	-.0264	-.0206	-.0052
							-.0134

Table D4 - HARMONIC ANALYSIS OF TANGENTIAL VELOCITY COMPONENT RATIOS
 AT THE EXPERIMENTAL RATIOS
 EXPERIMENT 4

TRIMMED 3.5 FEET BY THE BOW, DISPLACEMENT 26,390 TONS

HARMONIC	=	1	2	3	4	5	6	7	8
RADIUS = .359	=	-.0330	-.0120	.0556	-.0100	.0209	-.0033	.0069	-.0027
RADIUS = .556	=	-.1073	-.0268	.0157	-.0003	.0184	-.0139	.0104	-.0098
RADIUS = .774	=	-.1246	-.0364	-.0103	.0010	.0075	-.0017	.0068	-.0094
RADIUS = 1.017	=	-.1205	-.0493	-.0244	-.0356	-.0042	-.0063	-.0017	-.0009
RADIUS = 1.178	=	-.1272	-.0625	-.0323	-.0136	-.0105	-.0067	-.0059	-.0031

Table D5 - HARMONIC ANALYSIS OF TANGENTIAL VELOCITY COMPONENT RATIOS
AT THE INTERPOLATED RADII
EXPERIMENT 4

TRIMMED 3.5 FEET BY THE BOW, DISPLACEMENT 26,390 TONS

HARMONIC	=	1	2	3	4	5	6	7	8
$\text{RADII} =$	29.0	0.0676	0.0042	0.0932	-0.0236	0.0179	0.0203	-0.0044	0.0082
$\text{AMPLITUDE} =$	3.500	0.001	-0.0064	0.0705	-0.0144	0.0203	0.039	0.047	0.0008
$\text{RADII} =$	40.0	-0.0531	-0.0155	0.0460	-0.0073	0.0209	-0.0072	0.0082	-0.0047
$\text{RADII} =$	50.0	-0.0919	-0.0232	0.0254	-0.0023	0.0198	-0.0130	0.0100	-0.0085
$\text{RADII} =$	60.0	-0.1124	-0.0286	0.0094	0.0005	0.0161	-0.0106	0.0100	-0.0103
$\text{RADII} =$	70.0	-0.1209	-0.0329	-0.0029	0.0013	0.0111	-0.0047	0.0084	-0.0103
$\text{RADII} =$	80.0	-0.1233	-0.0373	-0.0120	0.0006	0.0061	-0.0069	0.0058	-0.0078
$\text{RADII} =$	90.0	-0.1203	-0.0420	-0.0179	-0.0016	0.0011	0.0006	0.0020	-0.0032
$\text{RADII} = 1.000$	-0.1202	-0.0461	-0.0235	-0.0049	-0.0035	-0.0000	-0.0012	-0.0010	

Table D6 - INPUT DATA FOR WAKE SURVEY ANALYSES
EXPERIMENT 4

TRIMMED 3.5 FEET BY THE BOW, DISPLACEMENT 26,390 TONS

ANGLE	RADIUS = .327	ANGLE	RADIUS = .504	ANGLE	RADIUS = .707
V/V	V/V	V/V	V/V	V/V	V/V
-5.3	.122	.005	.005	.005	.002
3.2	.128	.011	.005	.005	.002
12.2	.205	.061	.013	.007	.031
21.2	.221	.043	.002	.006	.012
28.1	.245	.045	.002	.002	.012
33.0	.301	.024	.017	.002	.015
51.6	.351	.026	.005	.005	.027
64.1	.456	.034	.003	.003	.025
64.2	.453	.035	.003	.003	.025
79.6	.558	.054	.002	.002	.010
99.5	.627	.077	.001	.001	.013
93.1	.655	.073	.002	.002	.013
112.9	.788	.049	.115	.006	.034
120.3	.763	.040	.122	.005	.028
132.9	.746	.039	.124	.007	.031
150.8	.707	.056	.102	.004	.029
152.4	.697	.054	.092	.005	.028
163.2	.643	.054	.090	.005	.028
165.9	.612	.050	.093	.006	.029
172.4	.559	.050	.079	.006	.029
175.9	.528	.050	.071	.007	.030
179.6	.515	.042	.074	.007	.030
183.2	.510	.042	.071	.007	.030
188.6	.449	.015	.094	.005	.028
192.2	.541	.073	.006	.006	.028
195.9	.562	.061	.025	.006	.028
210.1	.679	.055	.094	.005	.028
240.6	.794	.049	.117	.003	.028
271.4	.646	.075	.075	.003	.028
296.7	.425	.054	.075	.003	.028
301.5	.376	.044	.062	.003	.028
310.9	.351	.015	.051	.003	.028
332.5	.260	.016	.042	.002	.028
333.4	.220	.044	.093	.002	.028
344.7	.228	.067	.125	.002	.028
355.0	.251	.053	.127	.002	.028
356.7	.122	.005	.005	.005	.028

CONTINUED

Table D6 - INPUT DATA FOR WAKE SURVEY ANALYSES
EXPERIMENT 4
TRIMMED 3.5 FEET BY THE BOW, DISPLACEMENT 26,390 TONS

ANGLE	RADIUS = .92V VKV	VR/V V/V	VR/V V/V	ANGLE	RADIUS = 1.07V VKV	VR/V V/V	VR/V V/V
-5	-2.44	.007	.139	-5	-1.35	.131	.046
1.5	-2.54	-.135	.123	1.5	-1.58	.123	.046
3.1	-2.89	-.058	.123	3.1	-1.03	.123	.046
3.2	-2.96	-.036	.118	3.2	-1.07	.123	.046
8.5	-3.55	-.062	.070	8.5	-1.39	.070	.046
12.1	-4.29	-.082	.039	12.1	-1.97	.039	.046
15.7	-4.50	-.088	.015	15.7	-1.00	.009	.046
15.8	-5.01	-.100	-.009	15.8	-1.43	-.023	-.046
30.0	-6.76	-.085	-.029	30.0	-1.38	-.029	-.046
60.6	-9.85	-.095	-.036	60.6	-1.97	-.095	-.046
91.3	-9.17	-.036	.039	91.3	-1.04	-.036	.046
91.4	-9.10	-.034	.039	91.4	-1.04	-.034	.046
116.7	-9.21	-.064	.057	116.7	-1.32	-.064	.057
116.8	-9.23	-.065	.057	116.8	-1.32	-.065	.057
127.4	-9.31	-.056	.062	127.4	-1.23	-.056	.062
127.5	-9.20	-.053	.063	127.5	-1.19	-.053	.063
140.0	-9.33	-.042	.066	140.0	-1.04	-.042	.066
140.0	-9.26	-.042	.067	140.0	-1.04	-.042	.067
152.6	-9.27	-.043	.067	152.6	-1.04	-.043	.067
159.7	-9.36	-.026	.064	159.7	-1.04	-.026	.064
161.5	-9.33	-.024	.065	161.5	-1.04	-.024	.065
169.5	-9.33	-.023	.062	169.5	-1.04	-.023	.062
172.1	-9.34	-.019	.062	172.1	-1.04	-.019	.062
172.3	-9.32	-.017	.063	172.3	-1.04	-.017	.063
175.7	-9.27	-.023	.063	175.7	-1.04	-.023	.063
177.5	-8.60	-.050	.106	177.5	-1.04	-.050	.106
179.5	-7.68	-.037	.087	179.5	-1.04	-.037	.087
181.2	-7.89	-.017	.083	181.2	-1.04	-.017	.083
183.0	-8.51	-.002	.082	183.0	-1.04	-.002	.082
183.4	-8.68	-.042	.090	183.4	-1.04	-.042	.090
185.0	-9.09	-.013	.096	185.0	-1.04	-.013	.096
185.6	-9.32	-.047	.071	185.6	-1.04	-.047	.071
188.7	-9.28	-.028	.073	188.7	-1.04	-.028	.073
192.1	-9.33	-.031	.071	192.1	-1.04	-.031	.071
192.5	-9.25	-.031	.070	192.5	-1.04	-.031	.070
195.4	-9.29	-.035	.072	195.4	-1.04	-.035	.072
208.3	-9.31	-.040	.072	208.3	-1.04	-.040	.072
220.8	-9.12	-.050	.069	220.8	-1.04	-.050	.069
220.7	-9.20	-.070	.070	220.7	-1.04	-.070	.070
231.5	-9.29	-.058	.065	231.5	-1.04	-.058	.065
231.8	-9.21	-.059	.065	231.8	-1.04	-.059	.065
244.1	-9.15	-.072	.060	244.1	-1.04	-.072	.060
244.2	-9.26	-.071	.060	244.2	-1.04	-.071	.060
269.4	-9.26	-.040	.056	269.4	-1.04	-.040	.056
269.5	-9.15	-.100	.041	269.5	-1.04	-.100	.041
300.3	-8.98	-.040	.053	300.3	-1.04	-.040	.053
310.6	-8.51	-.126	.041	310.6	-1.04	-.126	.041
321.2	-8.24	-.09	.040	321.2	-1.04	-.09	.040
343.2	-8.48	-.043	.023	343.2	-1.04	-.043	.023
344.2	-8.64	-.070	.027	344.2	-1.04	-.070	.027
349.6	-8.13	-.053	.056	349.6	-1.04	-.053	.056
352.2	-8.45	-.061	.065	352.2	-1.04	-.061	.065
352.2	-8.16	-.055	.092	352.2	-1.04	-.055	.092
355.9	-8.24	-.100	.023	355.9	-1.04	-.100	.023
357.8	-8.51	-.046	.068	357.8	-1.04	-.046	.068
359.4	-8.75	-.009	.137	359.4	-1.04	-.009	.137
359.5	-8.64	-.134	.007	359.5	-1.04	-.134	.007

DTNSRDC ISSUES THREE TYPES OF REPORTS

1. DTNSRDC REPORTS, A FORMAL SERIES, CONTAIN INFORMATION OF PERMANENT TECHNICAL VALUE. THEY CARRY A CONSECUTIVE NUMERICAL IDENTIFICATION REGARDLESS OF THEIR CLASSIFICATION OR THE ORIGINATING DEPARTMENT.
2. DEPARTMENTAL REPORTS, A SEMIFORMAL SERIES, CONTAIN INFORMATION OF A PRELIMINARY, TEMPORARY, OR PROPRIETARY NATURE OR OF LIMITED INTEREST OR SIGNIFICANCE. THEY CARRY A DEPARTMENTAL ALPHANUMERICAL IDENTIFICATION.
3. TECHNICAL MEMORANDA, AN INFORMAL SERIES, CONTAIN TECHNICAL DOCUMENTATION OF LIMITED USE AND INTEREST. THEY ARE PRIMARILY WORKING PAPERS INTENDED FOR INTERNAL USE. THEY CARRY AN IDENTIFYING NUMBER WHICH INDICATES THEIR TYPE AND THE NUMERICAL CODE OF THE ORIGINATING DEPARTMENT. ANY DISTRIBUTION OUTSIDE DTNSRDC MUST BE APPROVED BY THE HEAD OF THE ORIGINATING DEPARTMENT ON A CASE-BY-CASE BASIS.

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